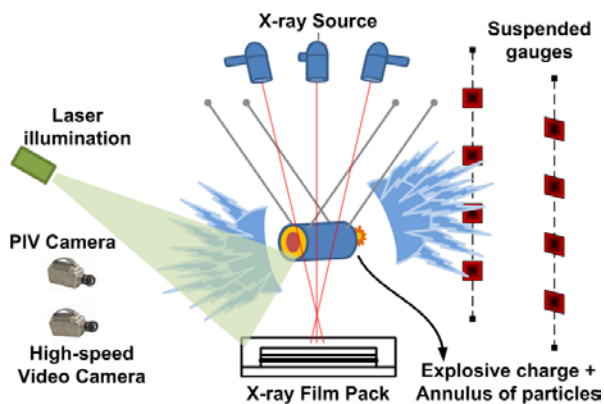


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CENTER FOR COMPRESSIBLE
MULTIPHASE TURBULENCE

TST Review

December 12-13, 2018



Agenda for TST Site Visit December 12 and 13, 2018

Wednesday December 12, 2018

- 7:40 Van arriving University Hilton for pickup
- 7:45 Van leaving University Hilton
- 8:00-8:45 Full Breakfast (Review team caucus)
- 8:45-10:00 T1 - CCMT Overview and Integration (Jackson, Bala)
- 10:00-10:30 T2 – Integrated Simulations (Bertrand Rollin)
- 10:30-10:45 Coffee break
- 10:45-11:15 T3 – Experiments (Heather Zunino)
- 11:15-11:45 T4 – Experiments (Angela Diggs, Kyle Hughes)
- 11:45-1:00 Lunch
- 1:00-1:45 T5 – CMT-nek (Jason Hackl, David Zwick)
- 1:45-2:30 T6 - UB (Chanyoung Park)
- 2:30-2:45 Coffee break
- 2:45-3:30 T7 - CS (Tania Banerjee)
- 3:30-4:15 T8 - Exascale (Herman Lam, Greg Stitt)
- 4:15-4:20 T9 - Internship presentations by Fred Ouellet
- 4:20-5:30 Social with Students/Posters (NEB Main Lobby)
- 5:30 Transportation to University Hilton by CCMT Faculty
- 6:00-7:30 Dinner at the University Hilton, hosted by CCMT Faculty

Thursday December 13, 2018

- 7:45 Van pickup at University Hilton
- 8:00-9:00 Continental Breakfast
- 9:00-10:45 Student Presentations (15 minutes each for presentations and questions)
- T10 – Mesoscale Simulations (Josh Garno)
 - T11 – PIEP Modeling (Chandler Moore)
 - T12 – Point-particle vs. Finite-size particle models (Samaun Nili)
 - T13 – Load Balancing (Keke Zhai)
 - T14 – BE of CMT-nek, Part 1 (Aravind Neelakantan)
 - T15 – BE of CMT-nek, Part 2 (Trokon Johnson)
- 10:45-11:30 Discussions between TST and CCMT PIs
- 11:30 Box Lunch; Transportation to hotel and/or airport as needed



CENTER FOR COMPRESSIBLE MULTIPHASE TURBULENCE

TST Review December 12-13, 2018 Attendee List

Faculty

S. Balachandar “Bala”	University of Florida	bala1s@ufl.edu
Rafi Haftka	University of Florida	haftka@ufl.edu
Nam-Ho Kim	University of Florida	nkim@ufl.edu
Herman Lam	University of Florida	hlam@ufl.edu
Sanjay Ranka	University of Florida	ranka@cise.ufl.edu
Greg Stitt	University of Florida	gstitt@ece.ufl.edu
Tom Jackson	University of Florida	tlj@ufl.edu
Siddharth Thakur “ST”	University of Florida	sst@ufl.edu
Bertrand Rollin	Embry-Riddle	rollinb@erau.edu
Ju Zhang	Florida Institute of Technology	jzhang@fit.edu

Review Team

Abhinav Bhatele (Chair)	LLNL	bhatele1@llnl.gov
David Daniel	LANL	ddd@lanl.gov
Maya Gokhale	LLNL	gokhale2@llnl.gov
John Hewson	Sandia	jchewso@sandia.gov
Mark Schraad	LANL	schraad@lanl.gov
Greg Weirs	Sandia	vgweirs@sandia.gov

Others

Andrew Cook	LLNL	cook33@llnl.gov
Angela Diggs	University of Florida	angela.diggs.1@us.af.mil
David Etim	HQ	David.Etim@nnsa.doe.gov
John Feddema	SNL	jtfedde@sandia.gov
Donald Littrell	Eglin Air Force Base	donald.littrell@us.af.mil
Kambiz Salari	LLNL	salari1@llnl.gov

Research Staff

Tania Banerjee	University of Florida	tmishra@cise.ufl.edu
Kei Fujisawa	University of Florida	fujisawa_ocean_eng@yahoo.co.jp
Jason Hackl	University of Florida	jason.hackl@ufl.edu
Nguyen Tri Nguyen	University of Florida	tringuyenttt@gmail.com
Chanyoung Park	University of Florida	cy.park@ufl.edu





CENTER FOR COMPRESSIBLE MULTIPHASE TURBULENCE

Students

Ryan Blanchard	University of Florida	rlb1116@ufl.edu
Sai Chenna	University of Florida	chenna@hcs.ufl.edu
Brad Durant	University of Florida	neoncrash@ufl.edu
Giselle Fernandez	University of Florida	gisellefernandez@ufl.edu
Joshua Garno	University of Florida	jpgarno@ufl.edu
Trokon Johnson	University of Florida	johnson@hcs.ufl.edu
Kyle Hughes	University of Florida	kylethughes89@ufl.edu
Rahul Koneru	University of Florida	rahul.koneru@ufl.edu
Tadbhagya Kumar	University of Florida	tkumar12@ufl.edu
Adeesha Malavi	University of Florida	adeeshaw@ufl.edu
Yash Mehta	University of Florida	ymehta@ufl.edu
Chandler Moore	University of Florida	wcmoore@ufl.edu
Aravind Neelakantan	University of Florida	aravindneela@ufl.edu
Samaun Nili	University of Florida	samaunnili@ufl.edu
Frederick Ouellet	University of Florida	f.ouellet@ufl.edu
Carlo Pascoe	University of Florida	poppyc@ufl.edu
Keke Zhai	University of Florida	zhaikeke@ufl.edu
Heather Zunino	Arizona State University	hzunino@asu.edu
David Zwick	University of Florida	dpzwick@ufl.edu

Administration Staff

Hollie Starr	University of Florida	hstarr@ufl.edu
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NNSA

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CCMT Overview

Y5 Accomplishments & Y6 Plan

T.L. Jackson
S. Balachandar



The slide features a blue vertical bar on the left with the text 'CCMT'. The main content is centered on a white background. At the bottom, there is an orange bar containing the logos for the University of Florida, the Department of Energy, and NASA.

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AST Meeting Agenda








<u>Wednesday</u>	<u>Thursday</u>
<ul style="list-style-type: none">➤ Overview & Integration (Jackson, Balachandar)➤ Integrated Simulations (Rollin)➤ <i>Coffee Break</i>➤ ASU Experiments (Heather Zunino)➤ Eglin Experiments (Hughes/Diggs)➤ <i>Lunch</i>➤ CMT-nek (Hackl, Zwick)➤ V&V and UQ (Park)➤ <i>Coffee Break</i>➤ CMT-nek: CS Updates (Tania Banerjee)➤ BE Results at Scale (Lam, Stitt)➤ Internship Presentation (Fred Ouellet)➤ Student Poster Session➤ <i>Dinner 6:00-7:30 (University Hilton)</i>	<ul style="list-style-type: none">➤ Student Presentations (6)➤ Open Discussions➤ <i>Box Lunch/Transportation</i>

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UNIVERSITY OF FLORIDA **Leadership**

Physics and Code Development							UQ and V&V	
								
S. (Bala) Balachandar	Siddharth Thakur (ST)	Thomas Jackson	Paul Fischer	Ju Zhang	Bertrand Rollin	Stanley Ling	Raphael Haftka	Nam-Ho Kim

Experiments			CS/Exascale			
						
Ronald Adrian	Charles Jenkins	Donald Littrell	Sanjay Ranka	Herman Lam	Gregory Stitt	Scott Parker

CCMT *UF members in red*


3

UNIVERSITY OF FLORIDA **Research Staff & Senior PhD Students**

			
<i>Tania Banerjee</i>	<i>Angela Diggs (Eglin AFB)</i>	<i>Kei Fujisawa</i>	<i>Jason Hackl</i>
			
<i>Nguyen T. Nguyen</i>	<i>Chanyoung Park</i>	<i>Carlo Pascoe</i>	<i>Sangjune Bae</i>

CCMT 4

UF UNIVERSITY OF FLORIDA **Current Students (19)**



Ryan Blanchard Sai Chenna Brad Durant Giselle Fernandez Joshua Garno Trokon Johnson

Kyle Hughes Rahul Koneru Li Lu (Illinois) Adeesha Malavi Maneesh Merugu Chandler Moore

Aravind Neelakantan Samaun Nili Frederick Ouellet Steven Paek Keke Zhai Heather Zunino (ASU) David Zwick

CCMT 5

UF UNIVERSITY OF FLORIDA **Internship Program – Completed (19)**

- Heather Zunino LANL May-Aug, 2014 Dr. Kathy Prestridge
- Kevin Cheng LLNL May-Aug, 2014 Dr. Maya Gokhale
- Nalini Kumar Sandia March-Aug, 2015 Dr. James Ang
- Christopher Hajas LLNL May-Aug, 2015 Dr. Maya Gokhale
- Christopher Neal LLNL June-Aug, 2015 Dr. Kambiz Salari
- Carlo Pascoe LLNL June-Aug, 2015 Dr. Maya Gokhale
- Giselle Fernandez Sandia Oct-Dec, 2015 Drs. Gregory Weirs & Vincent Mousseau
- Justin Mathew LANL May-Aug, 2015 Dr. Nick Hengartner
- David Zwick Sandia May-Aug, 2016 Drs. John Pott & Kevin Ruggirello

CCMT 6

UF UNIVERSITY OF FLORIDA **Internship Program – Completed (19)**

• Goran Marjanovic	Sandia	Aug-Nov, 2016	Drs. Paul Crozier & Stefan Domino
• Georges Akiki	LANL	May-Aug, 2016	Drs. Marianne Francois & Duan Zhang
• Paul Crittenden	LLNL	Spring, 2017	Drs. Kambiz Salari & Sam Schofield
• Mohamed Gadou	LANL	Summer, 2017	Dr. Galen Shipman
• Trokon Johnson	LANL	Summer, 2017	Drs. Cristina Garcia- Cardona, Brendt Wohlberg, Erik West
• Yash Mehta	LLNL	Summer, 2017	Dr. Kambiz Salari
• Kyle Hughes	LANL	Fall, 2017	Dr. Kathy Prestridge
• Prashanth Sridharan	LANL	Spring, 2018	Dr. Marianne Francois
• Fred Ouellet	LANL	Summer, 2018	Dr. Christoph Junghans
• Brad Durant	LANL	Fall, 2018	Dr. Joseph Schmidt

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UF UNIVERSITY OF FLORIDA **Internship Program – Not Completed**

- Joshua Garno, PhD (MAE, Physics and UQ)
- Chandler Moore, PhD (MAE, Physics) (NSF Graduate Research Fellowship)
- Ryan Blanchard, PhD (ECE, BE)

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Graduated Students (19)

- Kevin Cheng, MS (2014), Dr. Alan George, ECE
- Hugh Miles, BS (2015), Dr. Greg Stitt, ECE
- Chris Hajas, MS (2015), Dr. Herman Lam, ECE
- Angela Diggs, PhD (2015), Dr. S. Balachandar, MAE
 - Currently employed at Eglin AFB and working with center
- Subbu Annamalai, PhD (2015), Dr. S. Balachandar, MAE
 - Senior Systems Engineer, Optym, Gainesville FL
- Parth Shah, MS (2016), Drs. H. Lam and G. Stitt
- Georges Akiki, PhD (2016), Dr. S. Balachandar, MAE
 - Postdoctoral Associate, LANL, Dr. Marianne Francois
- Nalini Kumar, PhD (2017), Dr. H. Lam, ECE
 - Intel, Santa Clara CA
- Ajay Ramaswamy, MS (2017), Drs. H. Lam and G. Stitt
- Justin Matthew, MS (2017), Drs. Haftka and Kim, MAE
 - Proctor & Gamble, Cincinnati OH

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Graduated Students (19)

- Yiming Zhang, PhD (2018), Drs. Haftka and Kim, MAE
 - GE Global Research, Niskayuna, NY
- Mohamed Gadou, PhD (2018), Dr. Ranka, CISE
 - Bloomberg, New York, Software Engineer
- Brandon Osborne, MS (2018), Dr. Balachandar, MAE
- Yash Mehta, PhD (2018), Dr. Balachandar, MAE
 - Postdoctoral Associate, LLNL
- Kyle Hughes, PhD (2018), Drs. Haftka and Kim, MAE
 - Postdoctoral Associate, LANL, Dr. Prestridge
- Giselle Fernandez, PhD (2018), Drs. Haftka and Kim, MAE
 - Postdoctoral Associate, LANL
- Goran Marjanovic, PhD (2018), Dr. Balachandar, MAE
 - Raytheon
- Paul Crittenden, PhD (2018), Dr. Balachandar, MAE
 - Adjunct Professor, Department of Mathematics, UNF
- Prashanth Sridharan, PhD (2018), Drs. Balachandar and Jackson, MAE

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UNIVERSITY OF FLORIDA **PhD Students Expected Graduation Date**

- 2019
 - Carlo Pascoe
 - Heather Zunino
 - Rahul Koneru
 - Samaun Nili
 - Fred Ouellet
 - Keke Zhai
 - David Zwick
- 2020+
 - Brad Durant
 - Josh Garno
 - Sai Chenna
 - Trokon Johnson
 - Chandler Moore
 - Aravind Neelakantan
 - Ryan Blanchard

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UNIVERSITY OF FLORIDA **Students & Postdocs Placed at DOE/DOD Laboratories**

- Angela Diggs – Eglin AFB (2015)
- Georges Akiki – Postdoctoral Associate, LANL, Dr. Marianne Francois (2016)
- Cameron Stewart – Naval Surface Warfare Center, Indian Head, MD (2018)
- Kyle Hughes – Postdoctoral Associate, LANL, Dr. Prestridge (2018)
- Goran Marjanovic – Raytheon (2018)
- David Zwick – Multiphysics Code Development Computer Scientist, Computational Multiphysics Dept., Sandia National Laboratories (2019)
- Jason Hackl – Naval Surface Warfare Center, Indian Head, MD (2019)
- Giselle Fernandez – Sandia or LANL (2019)
- Yash Mehta – LANL (?) (2019)
- Waiting in line: 3 domestic, 4 international

- Academia: 3 (Georges Akiki; Bertrand Rollin; Paul Crittenden)
- Industry: 3 postdocs (Mrugesh Shringarpure, Subbu Annamalai; Chanyoung Park)


11 students (3 PhD; 7 BS and MS)

CCMT 12

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Deep Dive Workshops

- Exascale & CS Issues
 - Feb 3-4, 2015
 - University of Florida
- Multiphase Physics – I
 - Oct 13-14, 2016
 - Tampa FL
- CMT-nek/nek5000 User & Developer
 - April 17-18, 2018
 - Tampa FL
- Multiphase Physics – II
 - Stanford; Florida co-lead
 - October 22-23, 2018
- IUTAM Symposium on Dynamics and Stability of Fluid Interfaces
 - April 2-5, 2018
 - Gainesville, FL
- Workshop on Multiphase Flows
 - November 15-16, 2018
- Trilab Multiphase Workshop
 - 3 days; 2019; initial planning stages
 - S. Balachandar and Duan Zhang (LANL)



CMT-nek/nek5000 User & Developer Meeting
April 17-18, 2018


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Internal Workshops

- "Dakota - Tutorial" – organized by Chanyoung Park, February 19, 2015
- "Good Software Engineering Practices and Beyond" – organized by Bertrand Rollin, Feb 19, 2015
- "A Boot Camp on CMT-nek" – organized by B. Rollin and J. Hackl, November 29, 2017
- CMT-nek Multiphase Flows – early 2019



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CMT-nek Work Camp, 2017

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Additional Information

- Graduate Program Announcements
 - David Zwick – NSF Graduate Research Fellowship (Aug 2016)
 - Georges Akiki - MAE Best Dissertation Award (TSFD; May 2017)
 - Chandler Moore – NSF Graduate Research Fellowship (Aug 2017)
- Other metrics
 - Publications: 154
 - Presentations: 104
- Video Presentation at SC18
- Center Webpage
 - <http://www.eng.ufl.edu/ccmt/>

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Educational Programs

- Institute for Computational Science (ICE)
- Course in Verification, Validation and Uncertainty Quantification taught every third semester (N. Kim, R. Haftka)
- Yearly a specialized course for HPC for computational scientists (as part of the Computational Engineering Certificate) (S. Ranka)
- Fall, 2016, 2018 – graduate course on multiphase flows (S. Balachandar)
- Discusses exascale challenges and the NGEE work in the reconfigurable computing course (EEL5721/4720) and digital design (EEL4712) (H. Lam, G. Stitt)
- Uses the CCMT center as a motivational example in Introduction to Electrical and Computer Engineering (EEL3000) (H. Lam, G. Stitt)
- EEL6763 (Parallel Computer Architecture) (Ian Troxel)

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Management: Tasks and Teams

The Center is organized by physics-based tasks and cross-cutting teams, rather than by faculty and their research groups

Hour time slots	Exascale	CMT-nek	CS	Micro	Macro/Meso	UQ	Exp
Exascale	X	X	X			X	
CMT-nek	X	X	X	X	X		
CS	X	X	X				
Micro		X		X	X	X	X
Macro/Meso		X		X	X	X	X
UQ	X			X	X	X	X

- Weekly interactions (black); Regular interactions (red)
- Teams include students, staff, and faculty
- All staff and large number of graduate students located on 2nd floor of PS&T Building
- Construction in PS&T Building to add 6 new office spaces for students
- All meetings held in PS&T Building

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4th Annual CCMT Picnic (March, 2018)



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Outline

- Background
- Y5 accomplishments
- Y6 plans - Integration and timeline

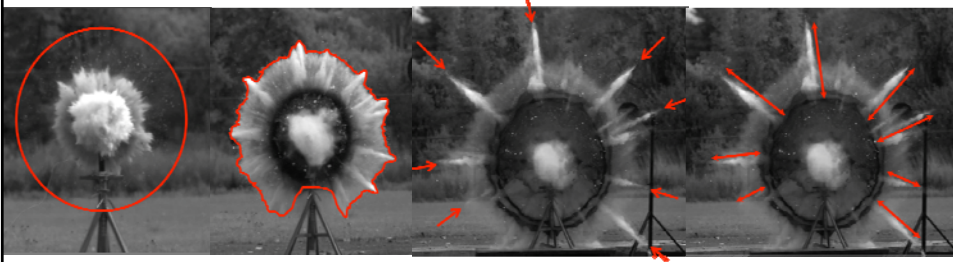
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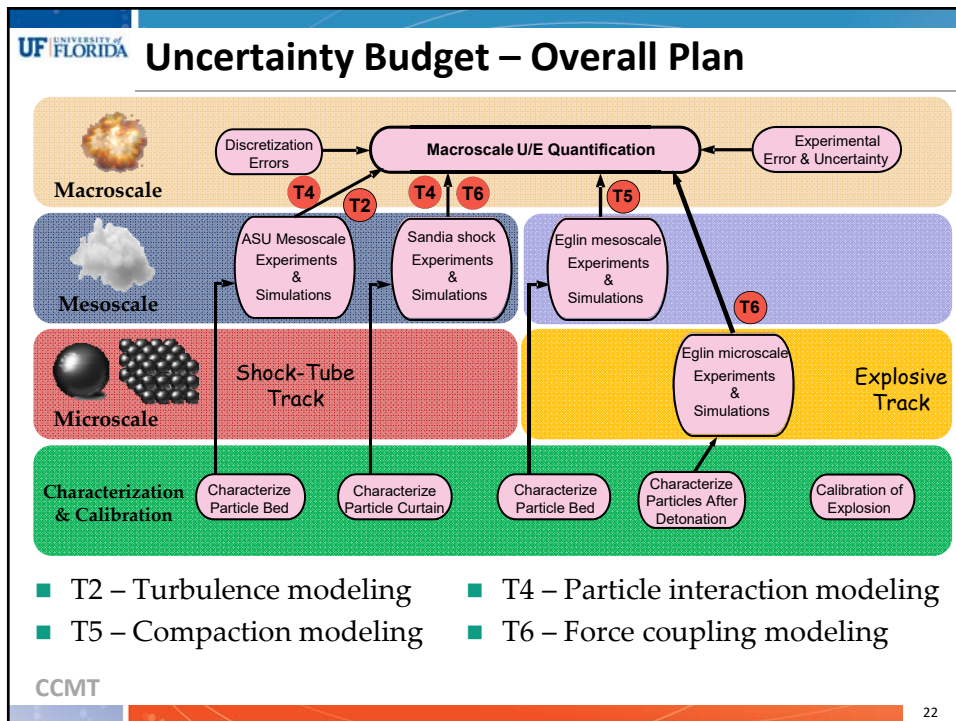
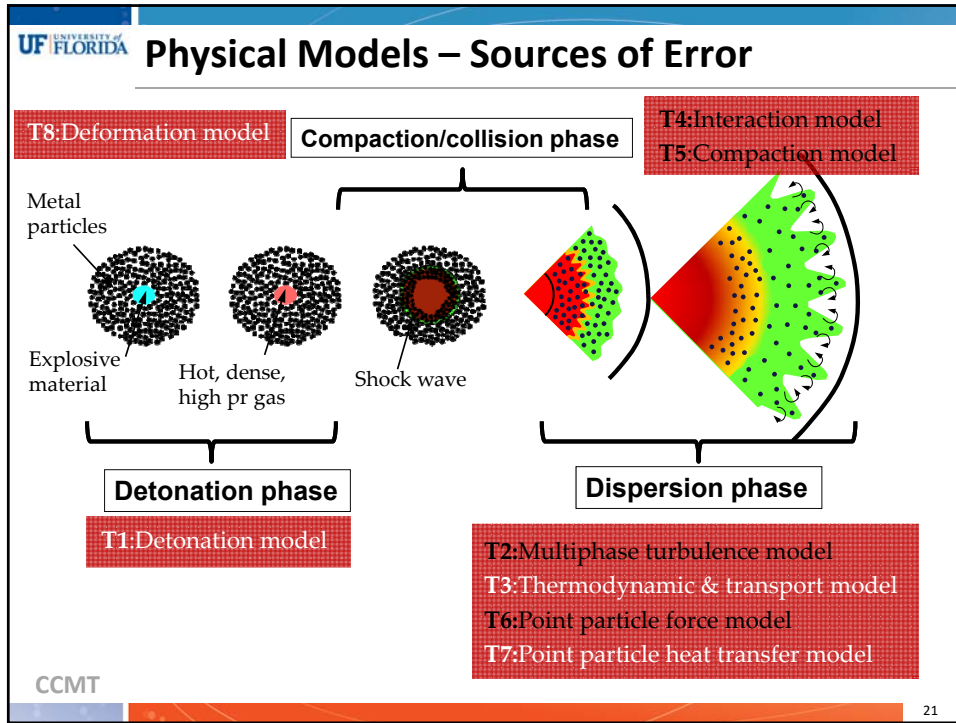
Prediction Metrics

PM-1: Blast Wave Location	PM-2: Particle Front Location	PM-3: Number of Instability Waves	PM-4: Amplitude of Instability Waves
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Major Accomplishments

- PIEP Framework for unresolved physics
- CMT-nek a versatile multiphase flow code (scalable to $>O(10^6)$ core)
- UQ integration and propagation across campaigns (+ Forensic UQ)
- BE framework for CMT-nek performance prediction
- Dynamic load balancing for multiphase flow
- $O(10^6)$ DSE acceleration using multiple FPGA
- Micro/meso understanding of shock-particle interaction physics

CCMT PIEP: Pairwise Interaction Expended Point-particle; DSE: Design Space Exploration

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Other Highlights

1. Macroscale – Yearly hero runs leading to full-physics simulations
2. Blastpad & other validation experiments
3. CMT-nek development and transition
4. Tight integration between teams
5. Student/Staff empowerment and engagement

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1: Demonstration Problem (Macroscale)

Goal

- Yearly perform the largest possible simulations of the demonstration problem and identify improvements to be made in predictive capability

Year 1-3 (Rocflu)

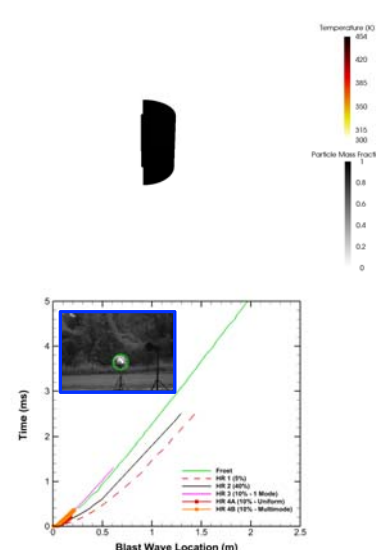
- Used existing code
- Qualitative comparison against experimental data of Frost (PM1 & PM2)
- Develop capabilities: real gas EOS, reactive burn, collision modeling
- Only 5 – 10% volume fraction
- O(10K) MPI threads

Presentation


- Bertrand Rollin

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Hero-3 demonstration simulation (Rocflu)



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1: Demonstration Problem (Macroscale)

Goal

- Yearly perform the largest possible simulations of the demonstration problem and identify improvements to be made in predictive capability

Year 4-5 (CMT-nek)

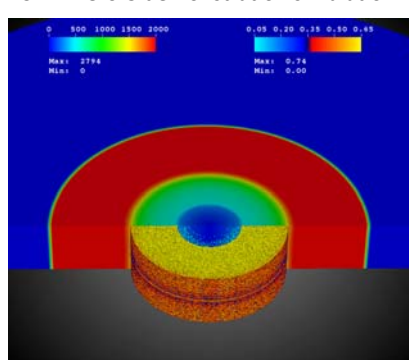
- 60% volume fraction
- O(130K) MPI threads
- Detonation condition - very strong shock
- Real gas EOS
- 4-way coupling
- Reactive burn initial condition
- Longer simulation in a bigger domain
- Extensive UQ & exploration

Presentation

- Bertrand Rollin

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CMT-Hero-3 demonstration simulation

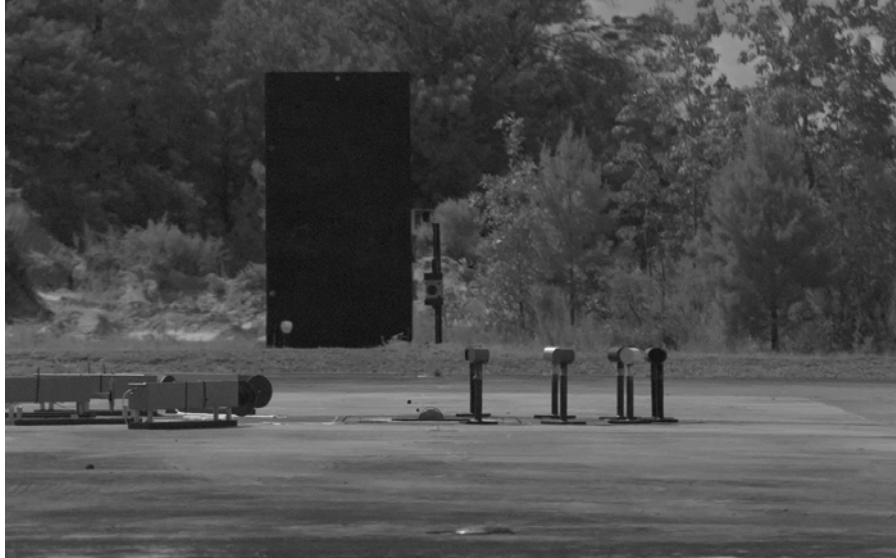


▪ **Features:**

- 32 M Degrees of freedom
- 2 M computational particles
- 0.3 x 0.3 x 0.0015 m
- $t_{\max} = 45 \mu\text{s}$
- 131,072 MPI ranks

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UNIVERSITY OF FLORIDA **2. Demonstration Problem - Blastpad**



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UNIVERSITY OF FLORIDA **2: Blastpad Experiments**

Goals

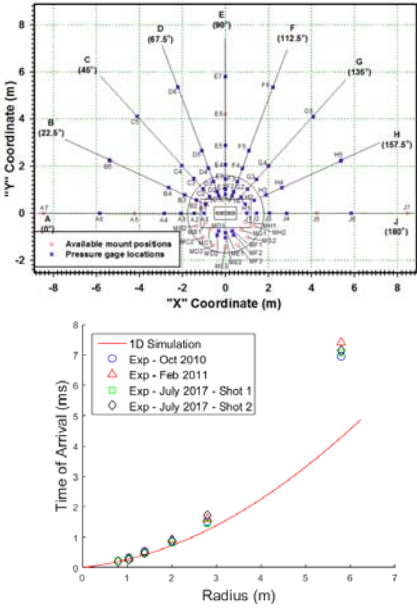
- Obtain validation-quality experimental measurements of the demonstration problem

Year 5

- Performed at Eglin AFB
- Detailed instrumentation for validation
- Simulation informed experiments
- Integrated UQ

Presentation

- Angela Diggs, Kyle Hughes, Bertrand Rollin



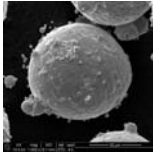
CCMT

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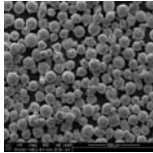
2: Blastpad Experiments

Input Parameter	Source
Explosive length [mm]	AFRL measurement
Explosive diameter [mm]	AFRL measurement at 5 locations
Explosive density [kg/m ³]	AFRL calculation
Explosive quality	AFRL X-ray
Particle diameter [mm]	CCMT measurement
Particle density [kg/m ³]	CCMT measurement
Particle volume fraction	AFRL calculation
Ambient pressure [kPa]	AFRL weather station
Ambient temperature [C]	AFRL weather station
Probe locations [m]	CCMT measurement

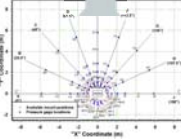
SEM of single steel particle at 1000x zoom.




SEM of several steel particles at 100x zoom.




Cam 3





Tungsten Liner



Steel Liner

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3: CMT-nek Development

Goals

- Co-design an exascale code (CMT-nek) for compressible multiphase turbulence
- Perform micro, meso and demonstration-scale simulations

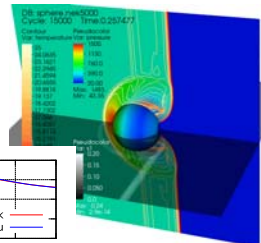
Year 5

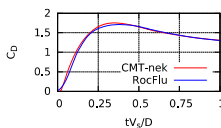
- Microscale simulations with CMT-nek
- Mesoscale simulations with CMT-nek
- Demo simulations with CMT-nek
- CMT-nek in nek5000 repository

Presentation


- Jason Hackl and David Zwick

Mach 3, $\gamma = 1.4$,
Color Pressure, contour temperature

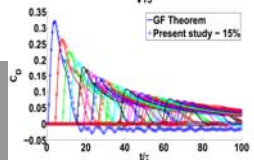





Mach number



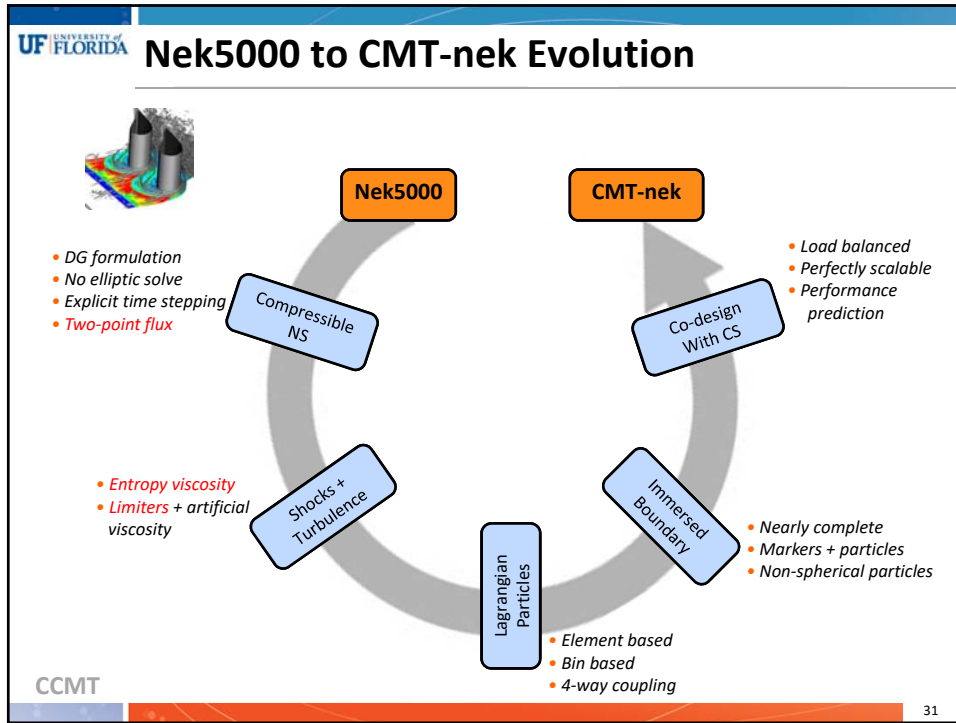
V15





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A Scalable & Versatile Multiphase Code

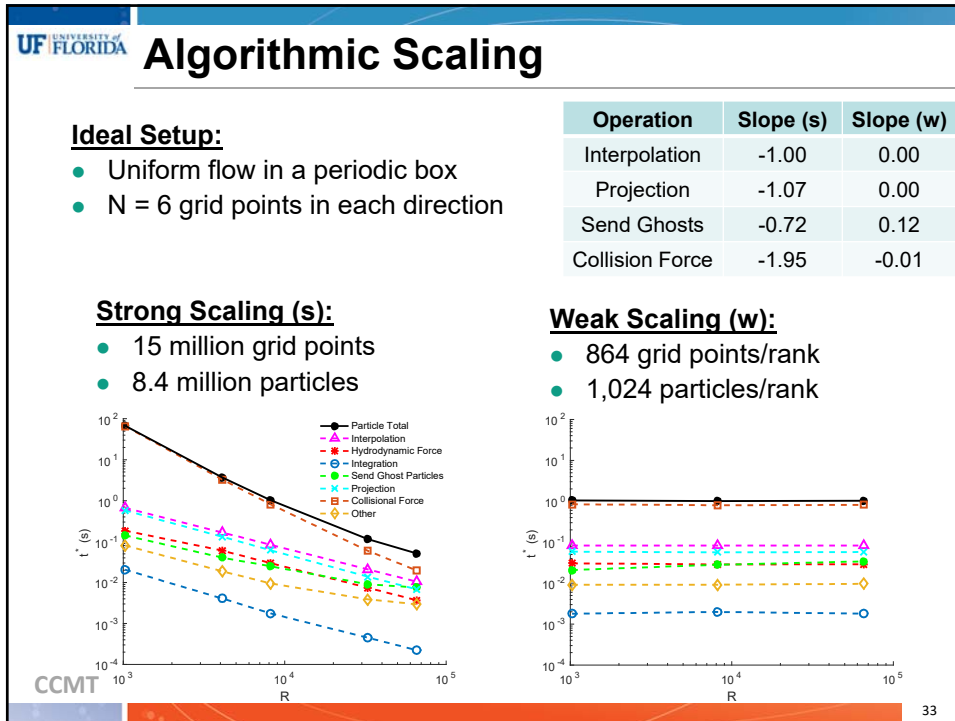
Existed

New

		nek5000	CMT-nek
Single-phase ↑	Fluid-Fluid	Incompressible	
		Low-Mach-number	
		Shock waves	
Increased multiphase coupling ↓	Tracer Particles		
	Particle-Fluid	Dilute	
		Dense	
	Particle-Particle		

Multiphase exascale problems rely on efficient communication and computations to maintain both accuracy and scalability

CCMT 32



CMT-nek Transition Workshop

A Boot Camp on CMT-nek

November 29, 2017
Organizers: Bertrand Rollin and Jason Hackl

Agenda

- **CMT-nek: Anatomy of the Beast**
Speaker: Jason Hackl | 1:30 pm – 2:15 pm
- **Lagrangian Particles in CMT-nek**
Speaker: David Zwick | 2:15 pm – 2:45 pm
- :

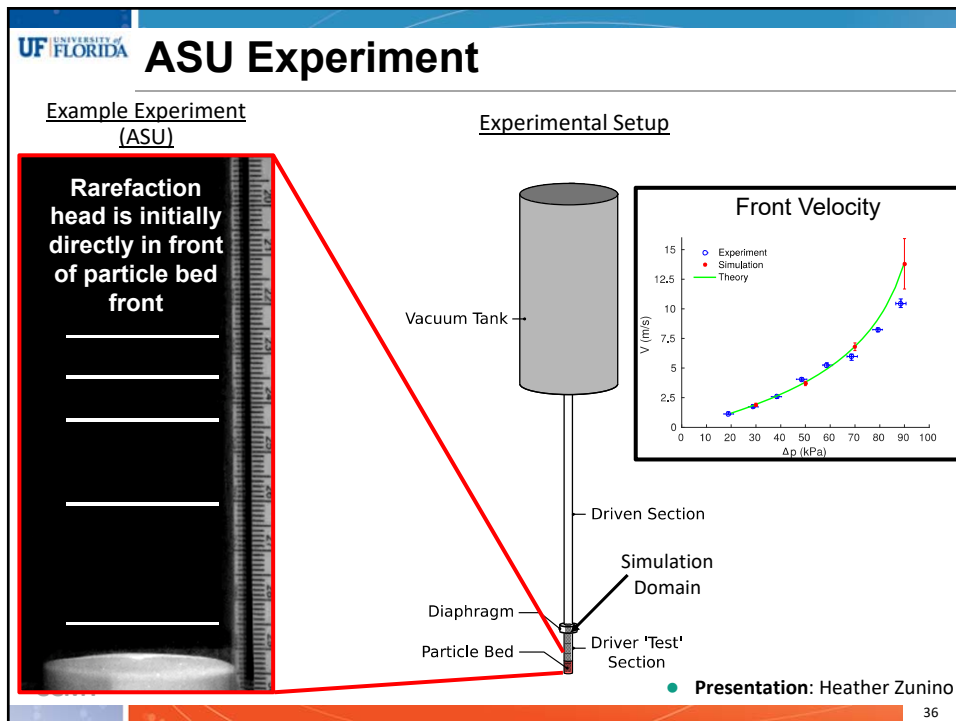
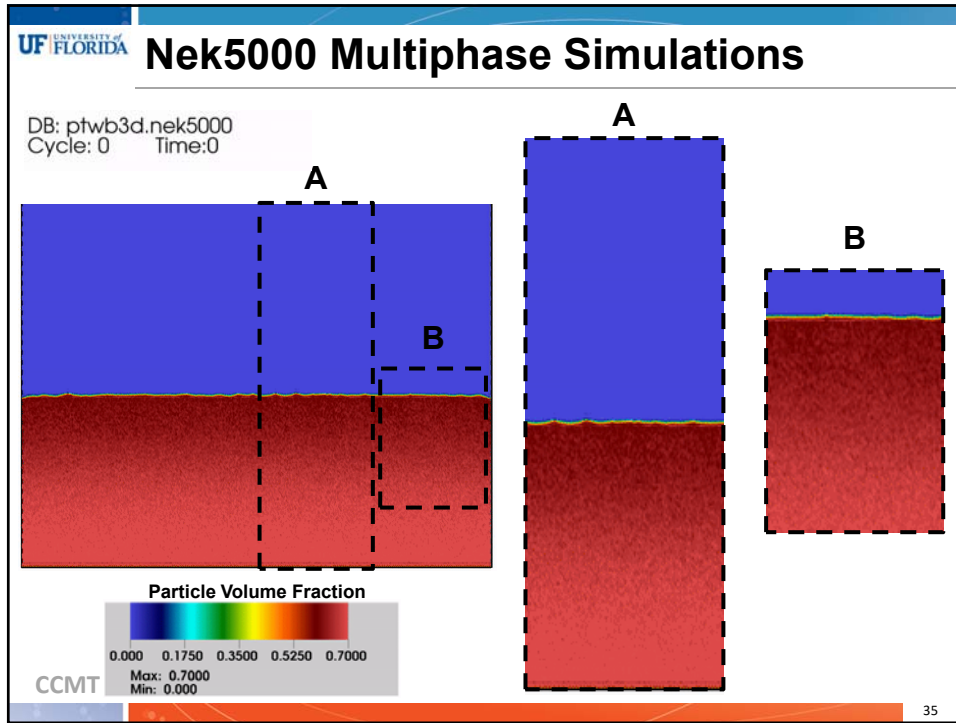
Nek5000 Developers/Users Meeting

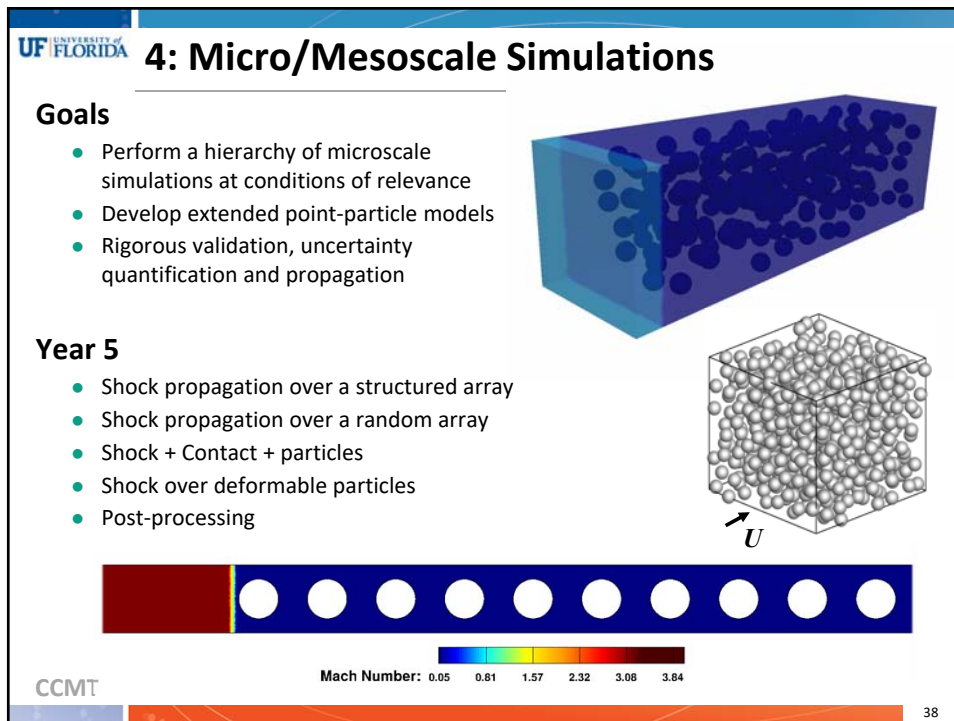
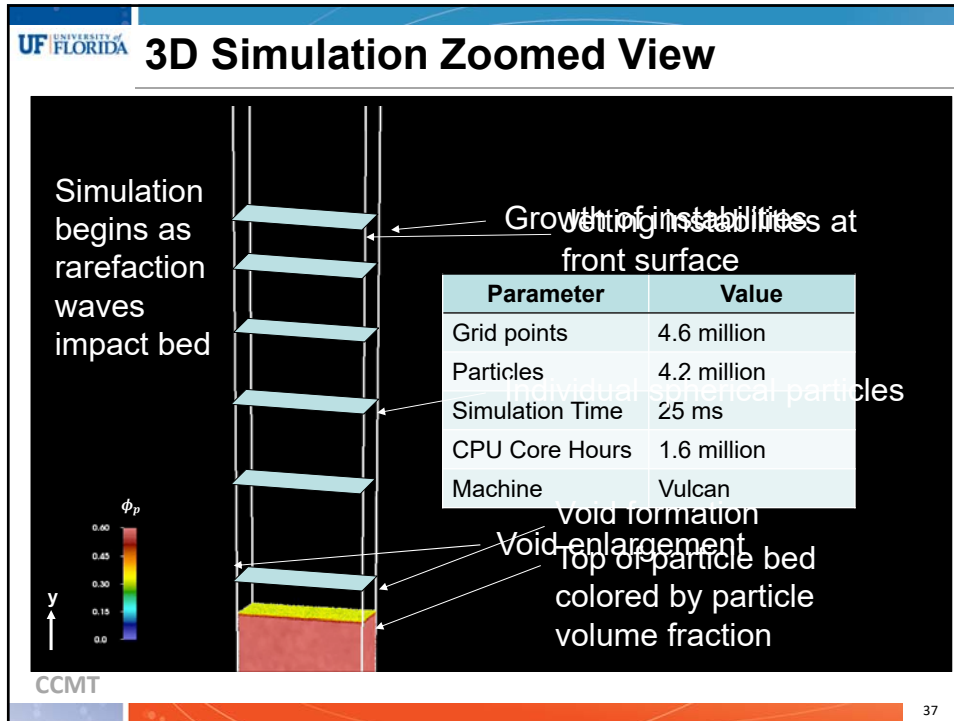
April 17-18, 2018
Organizers: T. Jackson, S. Balachandar and Paul Fischer

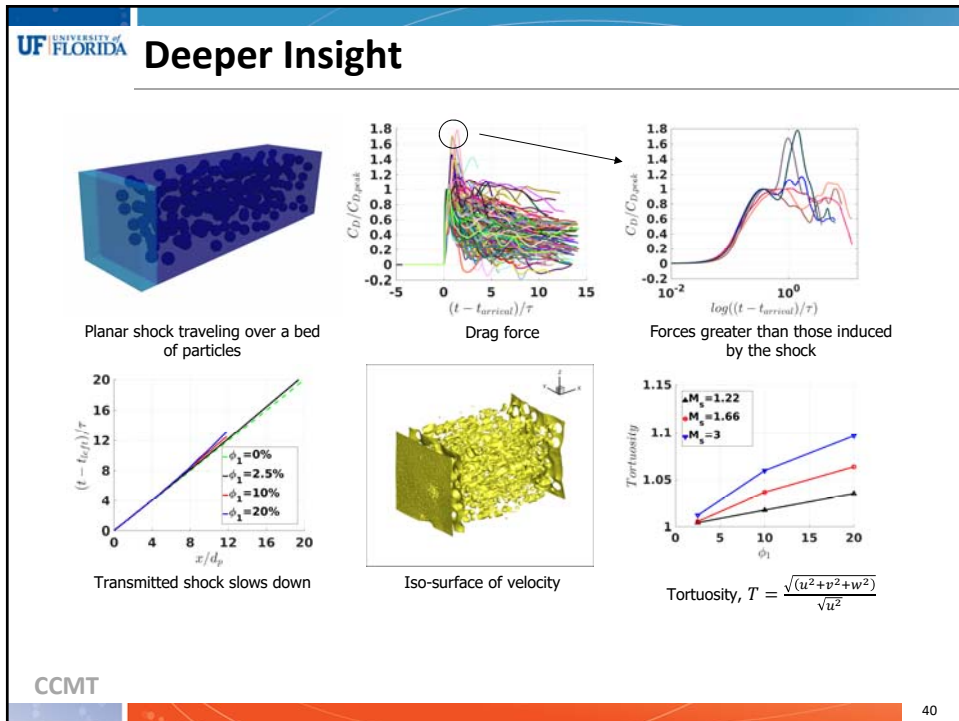
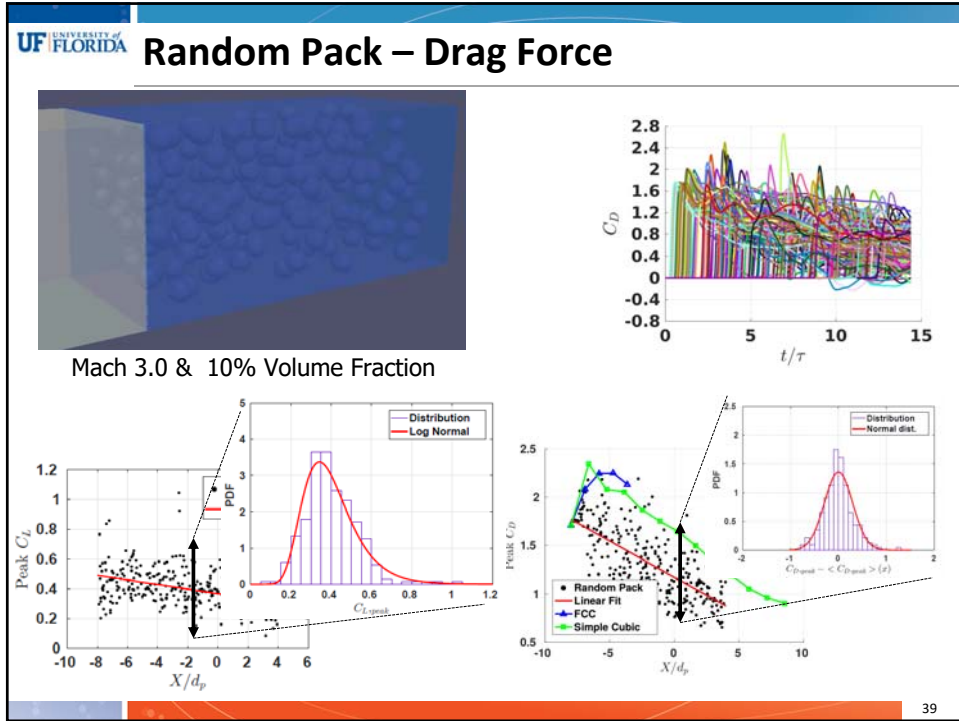
Agenda

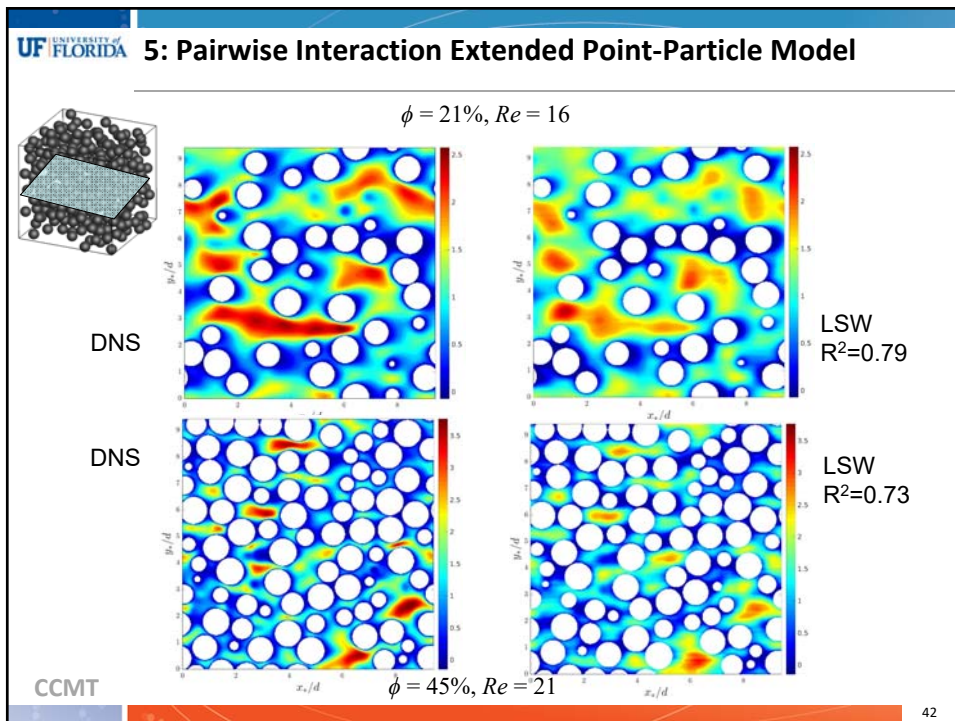
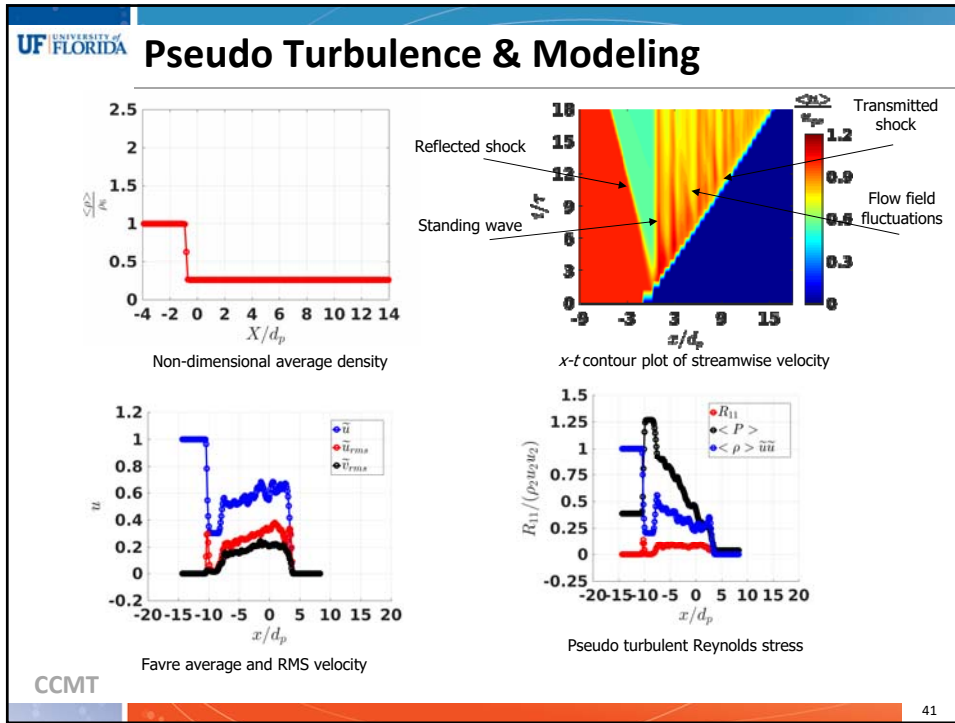
- **Nek5000 & CMT-nek presentations**
Speakers: | 9:00 am – 10:30 am
- :

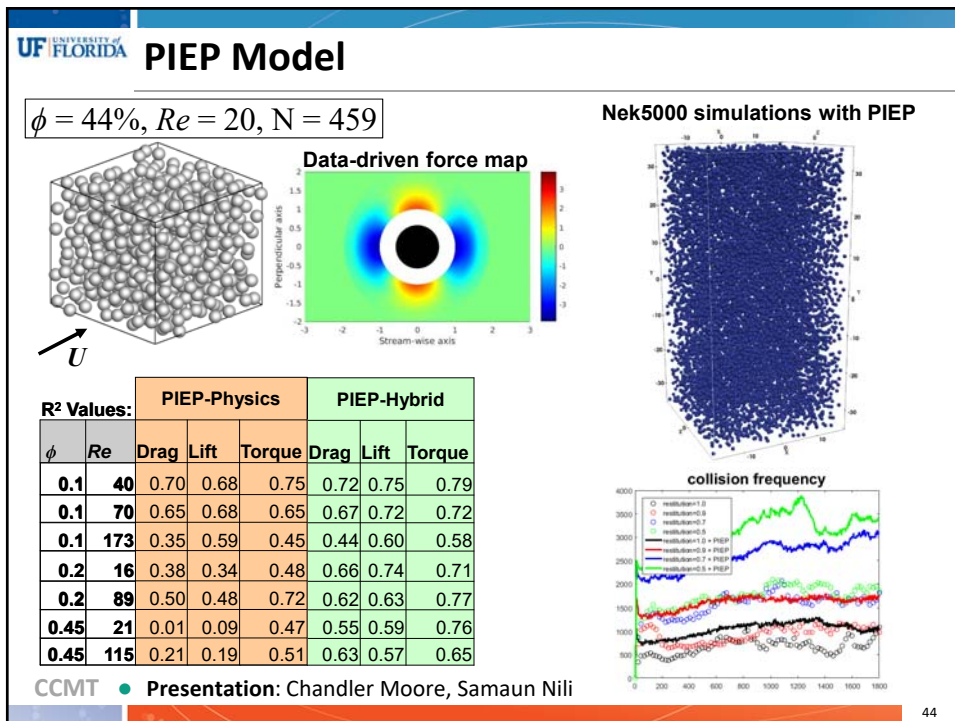
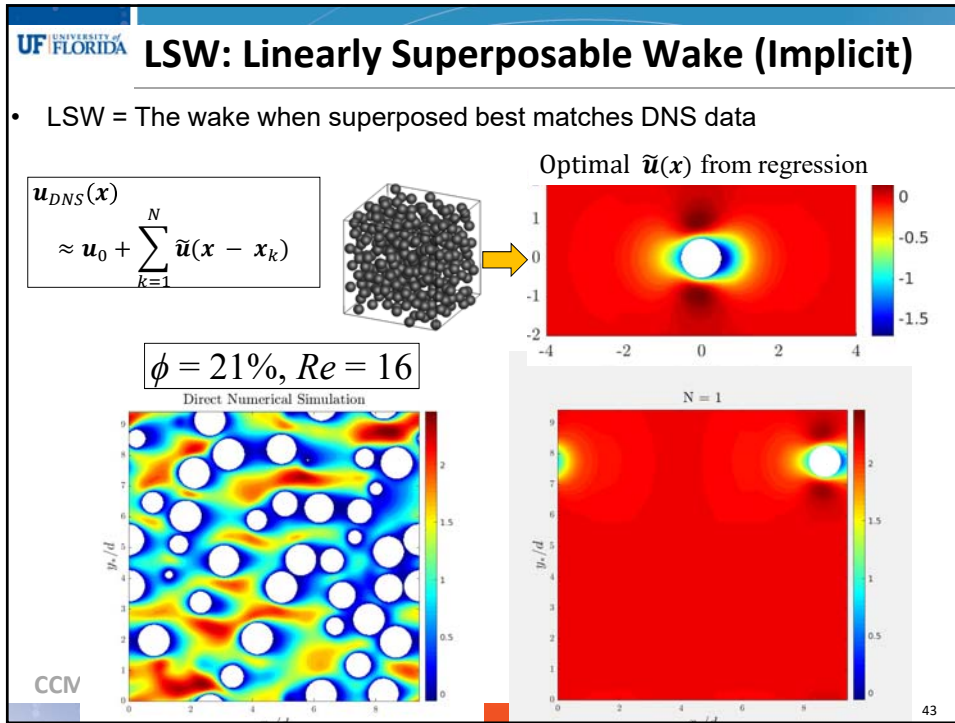
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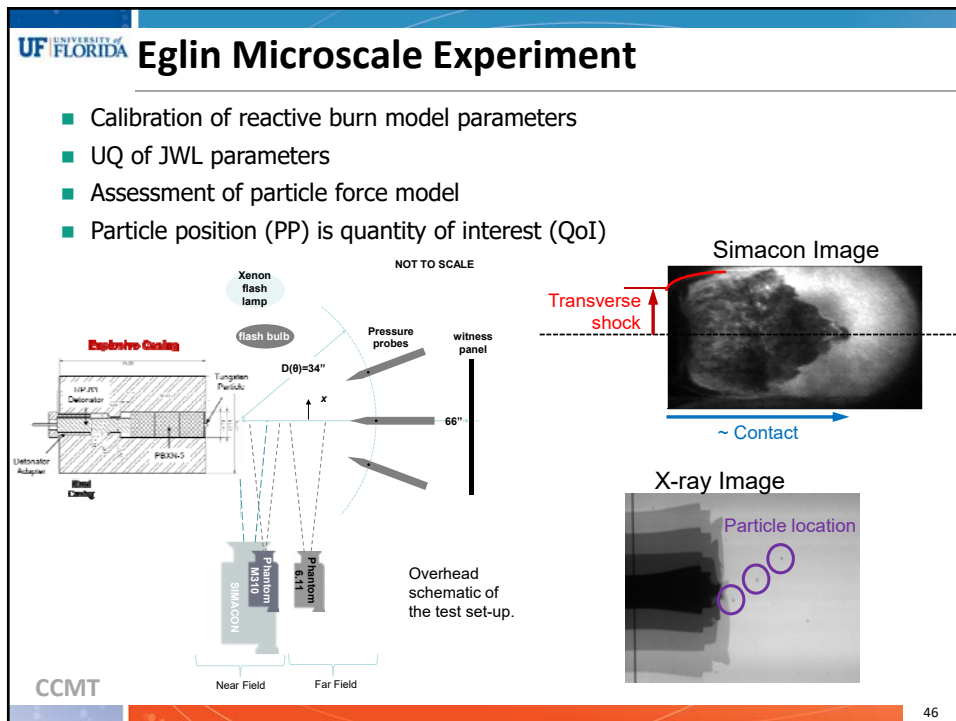
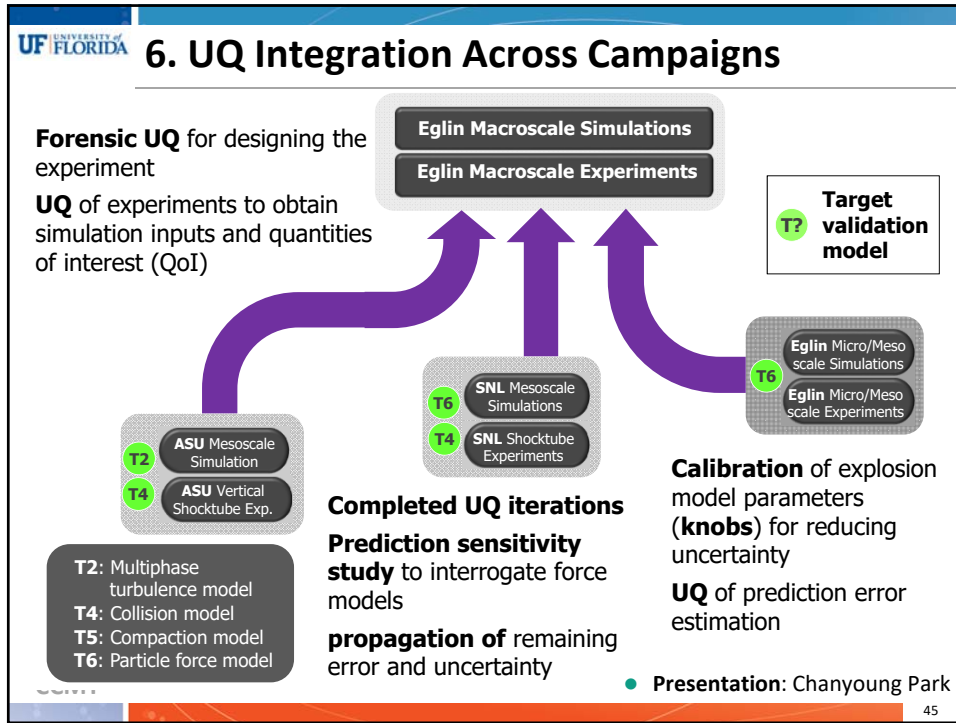












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Calibration - Uncertainty Reduction

Parameter	Quantity	Method
U1: Explosive density	mean = 1795 kg/m ³ , std = 2.9 kg/m ³ , Bi-modal	Derived
U2: RZ thickness	[0.365, 1.4] mm	Expert opinion
U3: Heat release Q	[11.6, 14.2] MJ/kg	Literature and opinion
U4: Particle diameter	mean = 2.0156 mm, std = 0.0073 mm, Weibull	Direct measurement Micrometer, 52 samples
U5: Particle density	mean = 15540 kg/m ³ , std = 250 kg/m ³ , Normal	Pycnometer Gas Pycnometer, 12 samples
U6: Initial radial position	[0, 0.254] mm	
U7-11: JWL	A, B, w, R ₁ , R ₂	Literature and opinion
...

- Including uncertainties in simulation parameters (knob)
- **Kyle** actively involved in measurements and uncertainty estimation (forensic UQ)

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Two Most Probable Detonation Parameter

- Predict **transverse shock position** and **particle position** with the micro simulation
- Transverse shock position and particle position **do not show meaningful difference** for the two most probable parameter estimates
- **Presentation:** Josh Garno

Transversers shock position

Time (sec) $\times 10^{-5}$

1796 kg/m³
0.88 mm
12.5 MJ/kg

1787 kg/m³
1.7 mm
12.9 MJ/kg

Particle position (m)

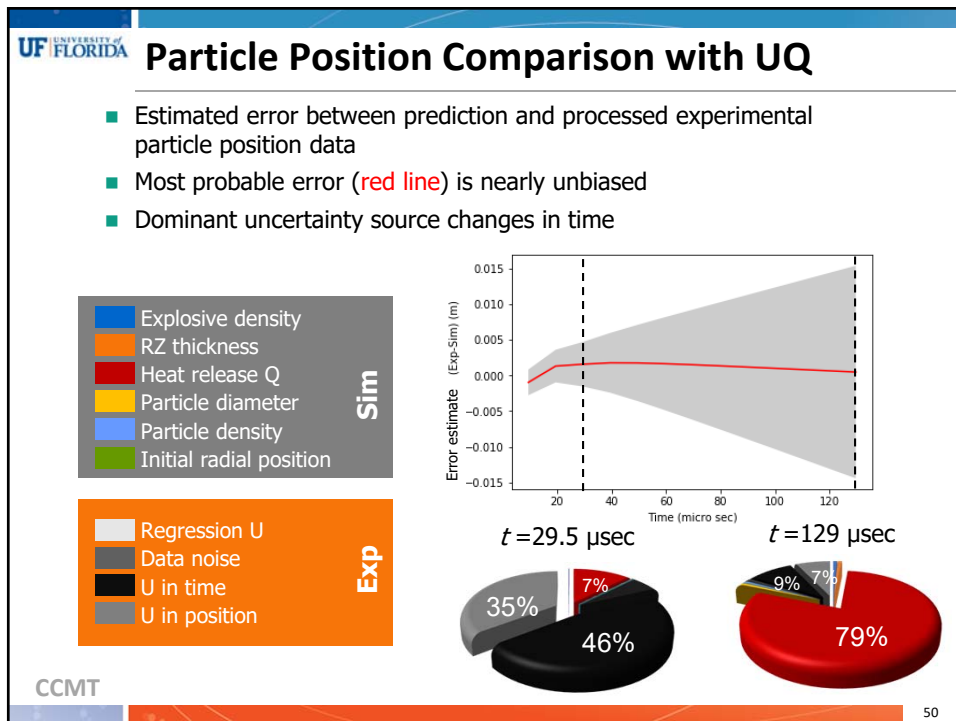
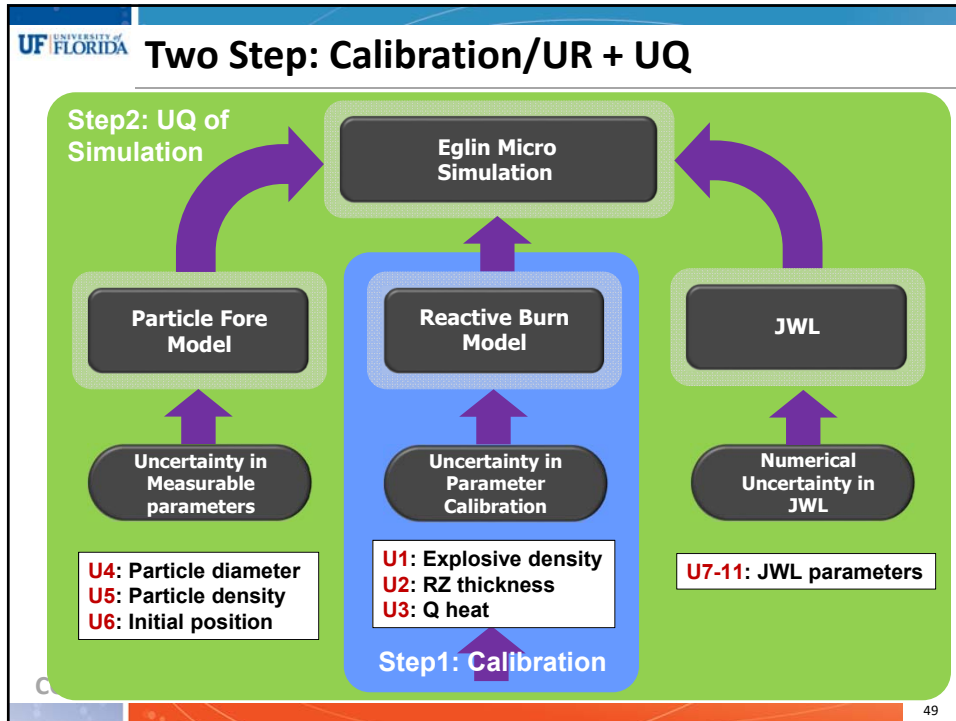
Time (micro sec)

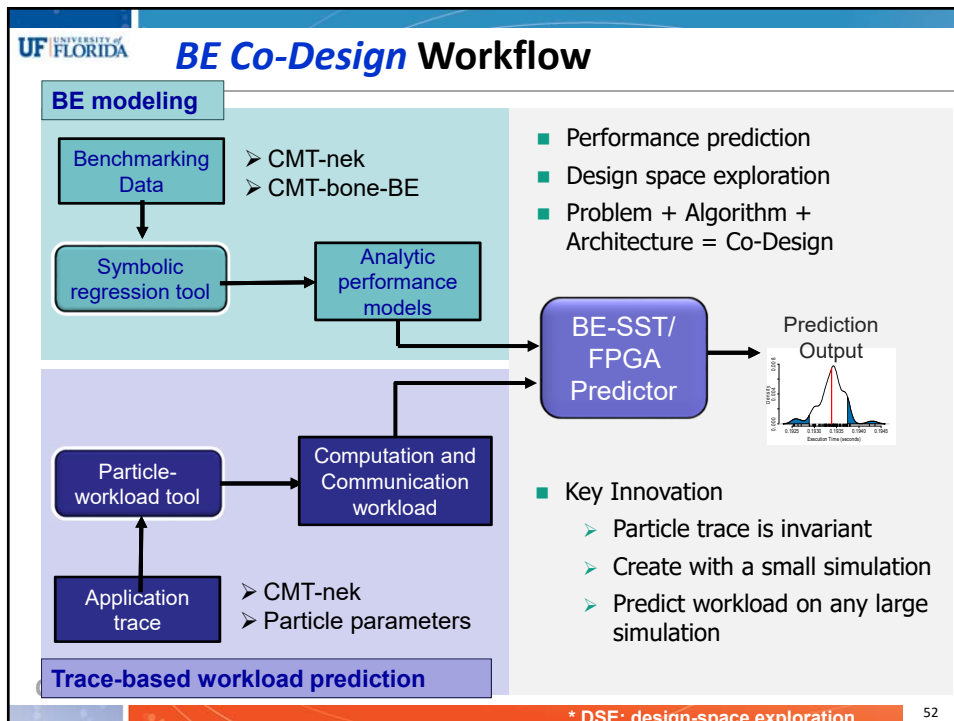
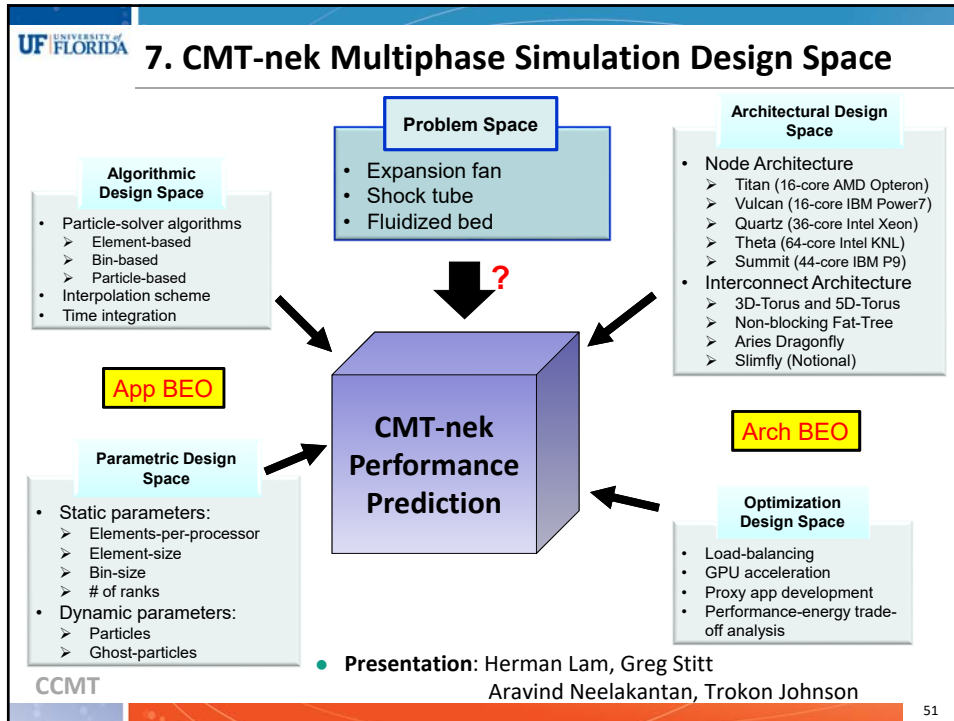
1796 kg/m³
0.88 mm
12.5 MJ/kg

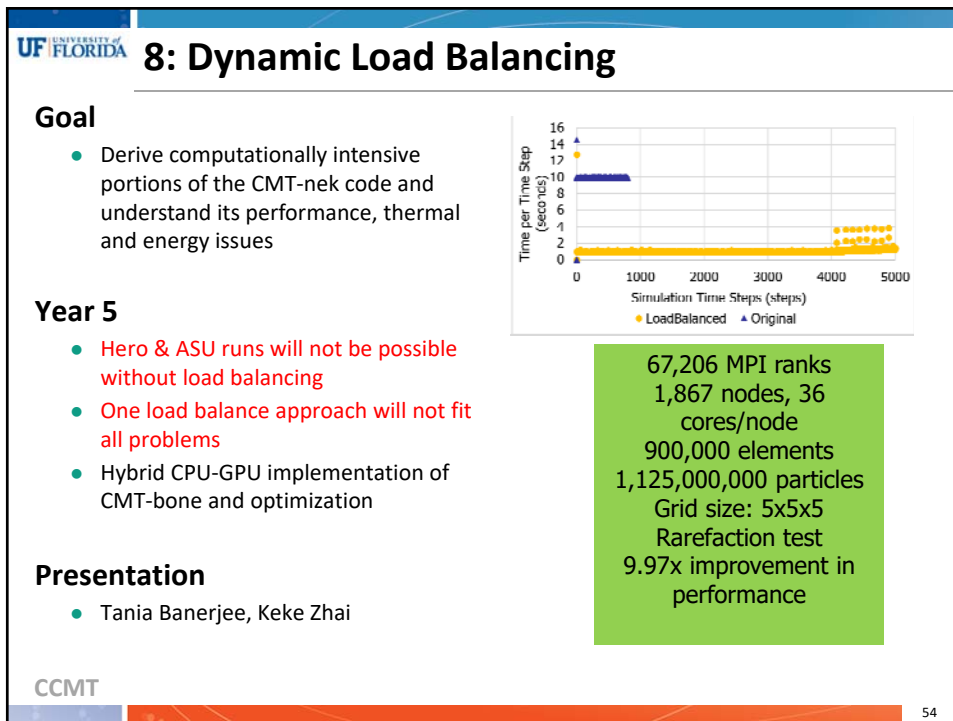
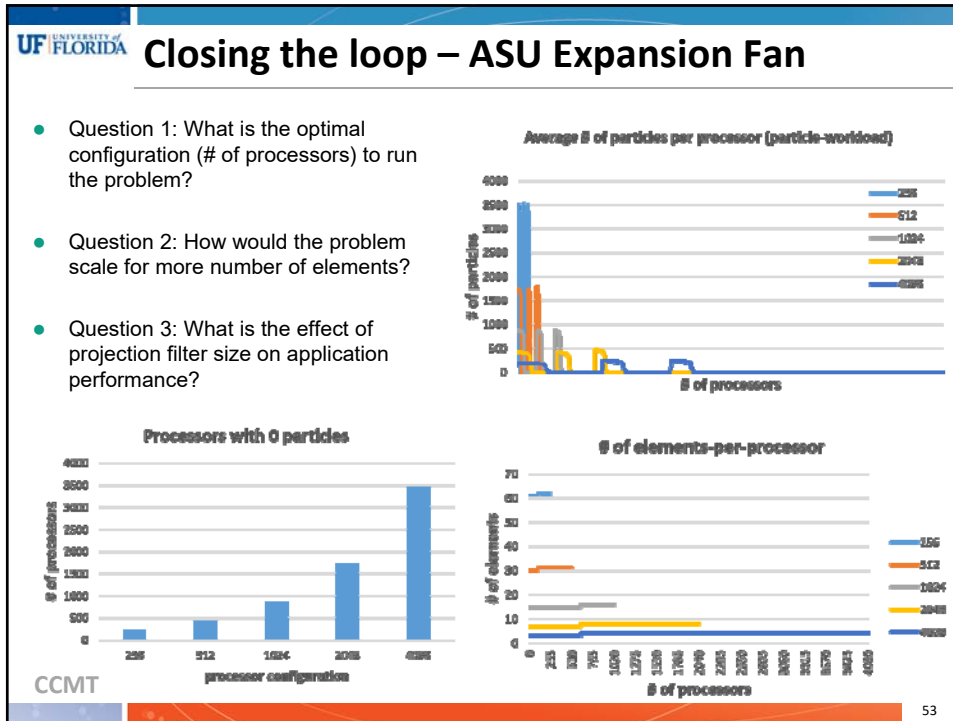
1787 kg/m³
1.7 mm
12.9 MJ/kg

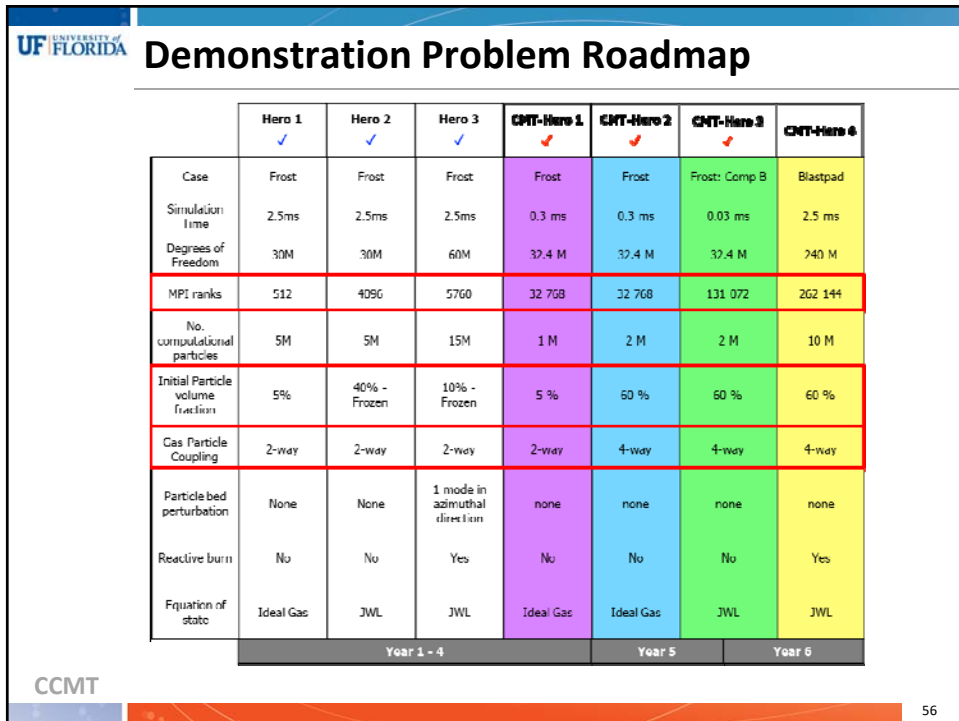
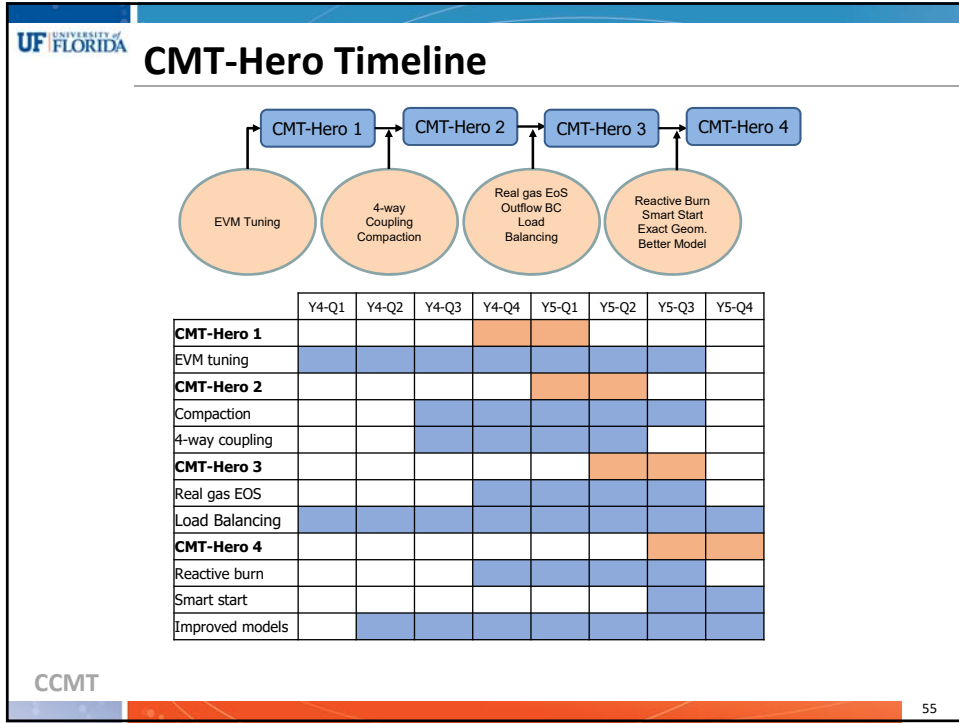
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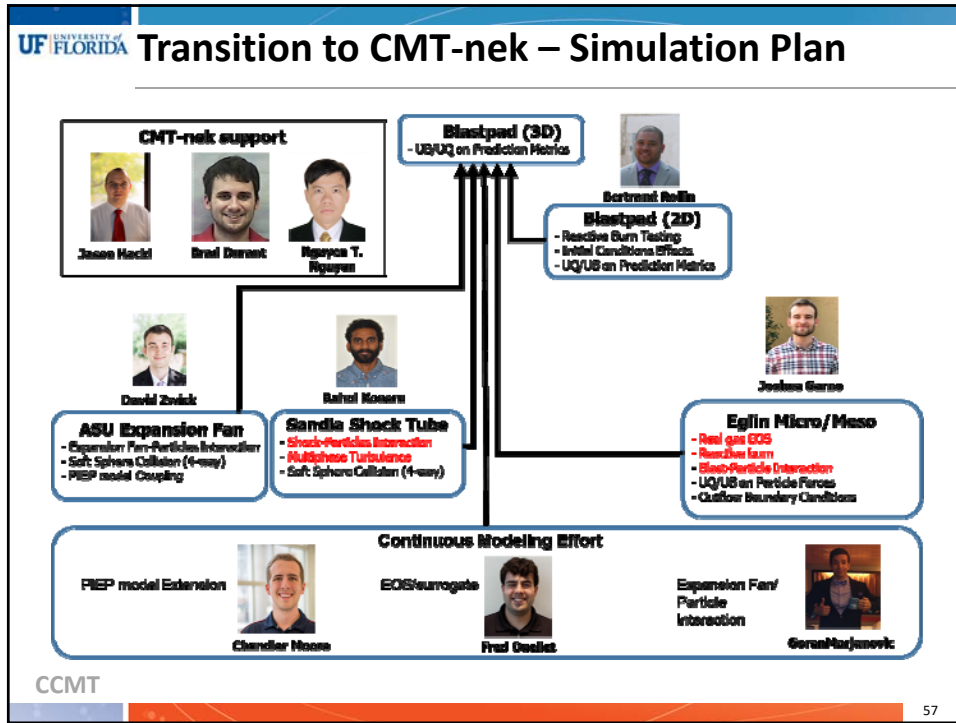
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Full System Integrated Simulations

Bertrand Rollin



The slide features a blue vertical bar on the left with the text 'CCMT'. The main title 'Full System Integrated Simulations' is centered in a large, bold, black font. Below the title, the name 'Bertrand Rollin' is centered. At the bottom, there is an orange horizontal bar containing three logos: the University of Florida logo, the Department of Aerospace and Astronautics logo, and the NASA logo.

UF UNIVERSITY OF FLORIDA **Integrated Team Members**

 Bertrand Rollin	 Fred Ouellet	 Rahul Koneru	 Joshua Garno	 Giselle Fernandez
 Jason Hackl	 Nguyen Tri Nguyen	 Goran Marjanovic	 David Zwick	 Brad Durant
 Yash Mehta	 Prashanth Sridharan	 Paul Crittenden	 Chandler Moore	

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The slide is titled 'Integrated Team Members' and features a grid of 15 portrait photos of team members. Each photo is accompanied by the member's name. The names are: Bertrand Rollin, Fred Ouellet, Rahul Koneru, Joshua Garno, Giselle Fernandez, Jason Hackl, Nguyen Tri Nguyen, Goran Marjanovic, David Zwick, Brad Durant, Yash Mehta, Prashanth Sridharan, Paul Crittenden, and Chandler Moore. The photos of Giselle Fernandez, Goran Marjanovic, Yash Mehta, and Prashanth Sridharan are overlaid with a red 'GRADUATED' stamp. The slide includes the University of Florida logo and the text 'CCMT' in the top left and bottom left corners.


UF UNIVERSITY OF FLORIDA **From Rocflu to CMT-nek**

	Rocflu	CMT-nek
Method	Finite volume	Discontinuous Galerkin
Mesh	Mixed cells (tetrahedra, hex)	Spectral elements (hex)
Shock capturing	AUSM+up	Artificial viscosity
Time marching	RK3, RK4	TVDRK3
Point particle modeling	2-way coupled	4-way coupled
Parallel capability	$O(10^4)$ MPI ranks, No load-balancing	$O(10^6)$ MPI ranks, load-balanced

➤ [Jason Hackl and David Zwick](#)

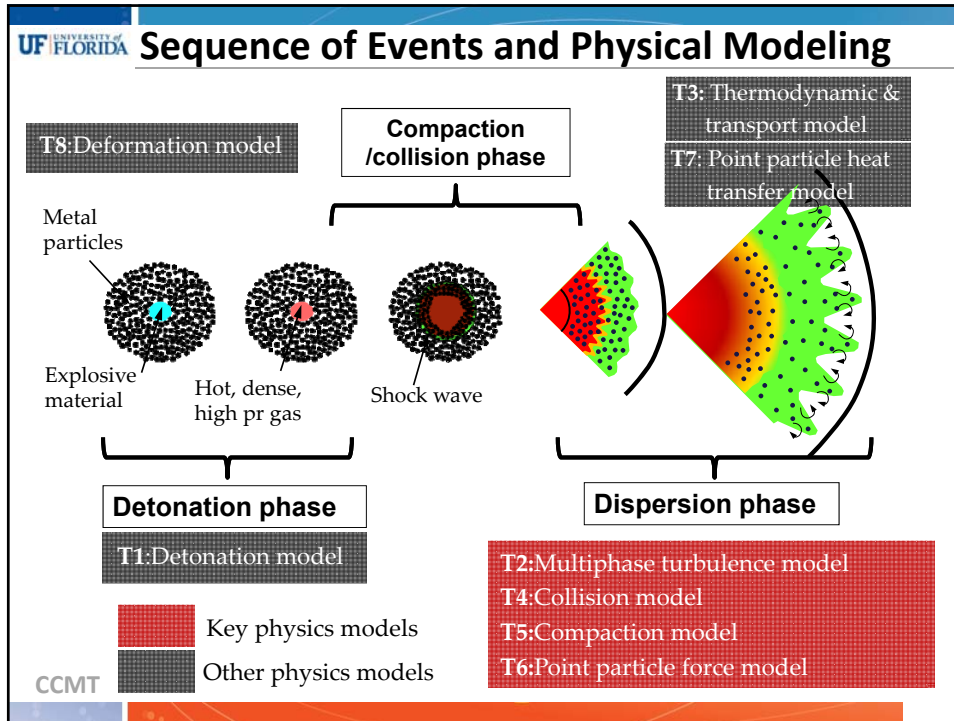
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UF UNIVERSITY OF FLORIDA **Motivation**




Dispersal of a layer of particles from the detonation of a Comp-B charge during the blast pad experiments performed at AFRL (2017)

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- 4 Micro/Meso Campaigns & Target Models**
- Sandia shock-tube
 - T6: Force coupling and T4: Particle-particle interaction
 - ASU expansion fan
 - T2: Multiphase turbulence and T4: Particle-particle interaction
 - Eglin microscale
 - T6: Force coupling
 - Eglin mesoscale gas-gun
 - T5: Compaction
- ↓**
- Demonstration problem
 - Yearly hero run
- CCMT



Rocflu Governing Equations

Compressible Multiphase Euler Equations:

$$\frac{\partial(\rho^g \phi^g)}{\partial t} + \nabla \cdot (\rho^g \phi^g \mathbf{u}^g) = 0$$

$$\frac{\partial(\rho^g \phi^g \mathbf{u}^g)}{\partial t} + \nabla \cdot (\rho^g \phi^g \mathbf{u}^g \mathbf{u}^g) + \nabla p^g = -\delta p \nabla \phi^g - \frac{1}{V_{cell}} \sum_i \zeta_i \mathbf{F}_i^{gsp}$$

$$\frac{\partial(\rho^g \phi^g E^g)}{\partial t} + \nabla \cdot (\rho^g \phi^g \mathbf{u}^g E^g) + \nabla \cdot (\phi^g p^g \mathbf{u}^g) = -\delta p \mathbf{u}^g \cdot \nabla \phi^g - p^{int} \frac{\partial \phi^g}{\partial t} - \frac{1}{V_{cell}} \sum_i \zeta_i (\mathbf{G}_i^{gsp} + \mathbf{Q}_i^{gsp})$$

Lagrangian Equations:

$$\frac{d\mathbf{x}_i^p}{dt} = \mathbf{u}_i^p$$

$$m_i^p \frac{d\mathbf{u}_i^p}{dt} = \mathbf{F}_i^{gsp} + \mathbf{F}_i^{ppp}$$

$$m_i^p C_i^p \frac{dT_i^p}{dt} = \mathbf{Q}_i^{gsp}$$

$$\mathbf{F}_{qs,i} = 3\pi \mu_i^g d_i^p |\mathbf{u}_i^g - \mathbf{u}_i^p| \frac{Re_i^p}{24} C_D(Re_i^p, M_i^p, \phi_i^p)$$

$$\mathbf{F}_{ps,i} = -V_i^p (\nabla p^g)_i$$


$$\mathbf{F}_{om,i} = V_i^p C_M(M_i^p, \phi_i^p) \left[\left(\frac{D\rho_i^g u_i^g}{Dt} \right)_i - \frac{d(\rho_i^g u_i^p)}{dt} \right]$$

$$\mathbf{F}_{vu,i} = 3\pi \mu_i^g d_i^p \theta(\phi_i^p) \int_{-\infty}^t K_{vu}(t-\chi, Re_i^p, M_i^p, \phi_i^p) \left[\left(\frac{D\rho_i^g u_i^g}{Dt} \right)_i - \frac{d(\rho_i^g u_i^p)}{dt} \right]_{t=\chi} d\chi$$

$$\mathbf{F}_i^{ppp} = -\frac{V_i^p}{\phi_i^p} \nabla \tau_{pp,i} \quad \tau_{pp,i} = \frac{P_s(\phi_i^p)^\beta}{\phi_i^p - \phi_i^g}$$

$$\mathbf{Q}_{qs,i} = \pi d_i^g \kappa^g Nu(T_i^g - T_i^p)$$

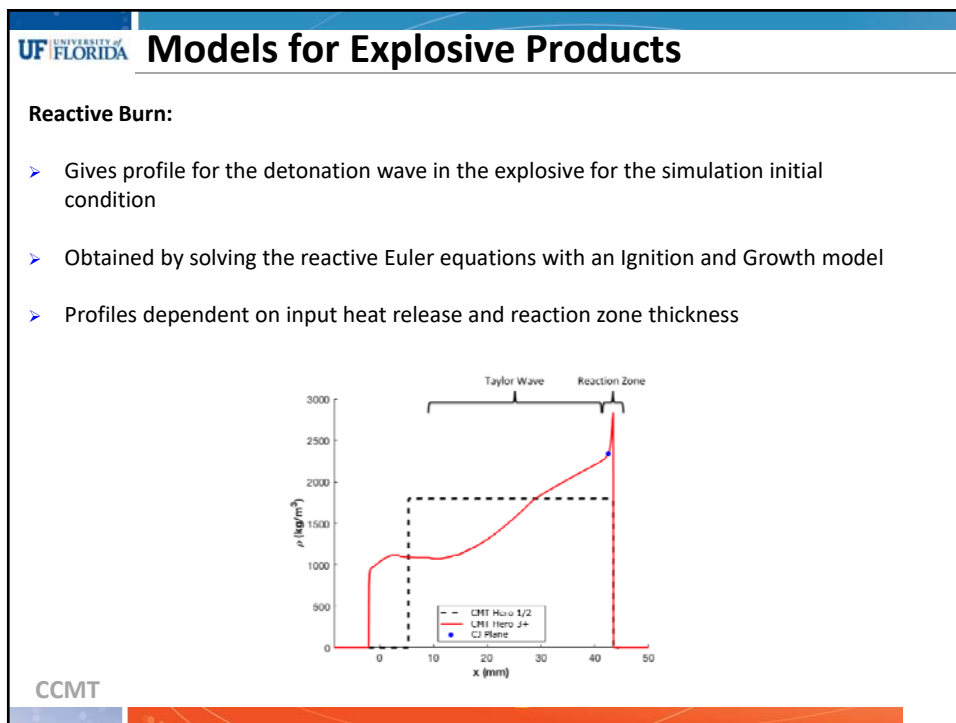
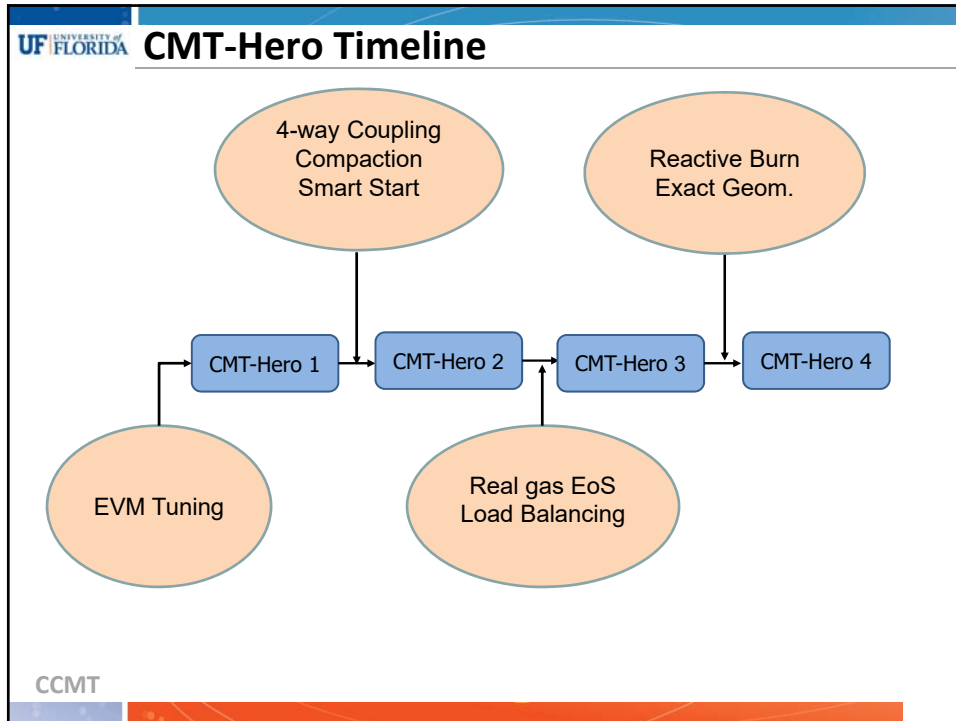
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Demonstration Problem Roadmap

	Hero 1	Hero 2	Hero 3	CMT-Hero 1	CMT-Hero 2	CMT-Hero 3	CMT-Hero 4
	✓	✓	✓	✓	✓	✓	
Case	Frost	Frost	Frost	Frost	Frost	Frost: Comp B	Blastpad
Simulation Time	2.5ms	2.5ms	2.5ms	0.3 ms	0.3 ms	0.03 ms	2.5 ms
Degrees of Freedom	30M	30M	60M	32.4 M	32.4 M	32.4 M	240 M
MPI ranks	512	4096	5760	32 768	32 768	131 072	262 144
No. computational particles	5M	5M	15M	1 M	2 M	2 M	10 M
Initial Particle volume fraction	5%	40% - Frozen	10% - Frozen	5 %	60 %	60 %	60 %
Gas-Particle Coupling	2-way	2-way	2-way	2-way	4-way	4-way	4-way
Particle bed perturbation	None	None	1 mode in azimuthal direction	none	none	none	none
Reactive burn	No	No	Yes	No	No	No	Yes
Equation of state	Ideal Gas	JWL	JWL	Ideal Gas	Ideal Gas	JWL	JWL
	Year 1 - 4			Year 5		Year 6	

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UF UNIVERSITY OF FLORIDA Models for Explosive Products

JWL Equation of State:

$$P_{JWL}(\rho, e) = A\left(1 - \frac{\omega}{R_1 V}\right)e^{-R_1 V} + B\left(1 - \frac{\omega}{R_2 V}\right)e^{-R_2 V} + \omega \rho e$$

$$T_{JWL}(P, \rho) = \left(\frac{1}{\rho \omega C_v}\right)(P - Ae^{-R_1 V} - Be^{-R_2 V})$$

Mixture Equation of State:

- > Linear functions of gas density and internal energy used for JWL parameters A, B, ω and C_v
- > Parameters take product values when above reported density and energy of the explosive material
- > They decay to the values for air at atmospheric conditions to recover the ideal gas equations

$\rho/e = \rho/e_{exp}$	$\rho/e = \rho/e_{air}$
A	0
B	0
ω	$\gamma - 1$
$C_{v,HE}$	$C_{v,air}$

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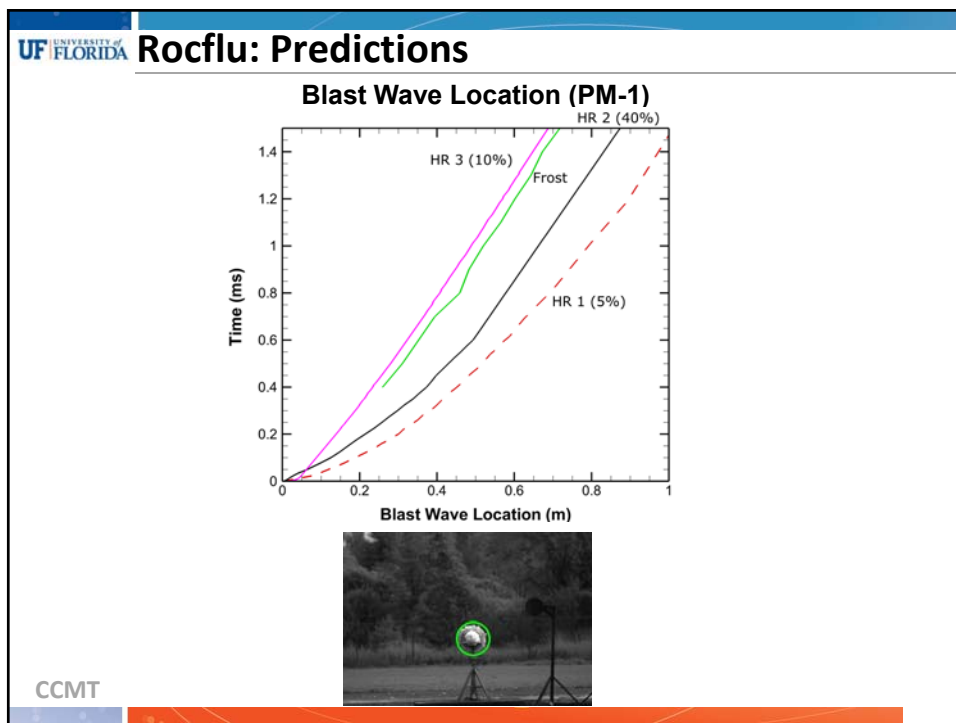
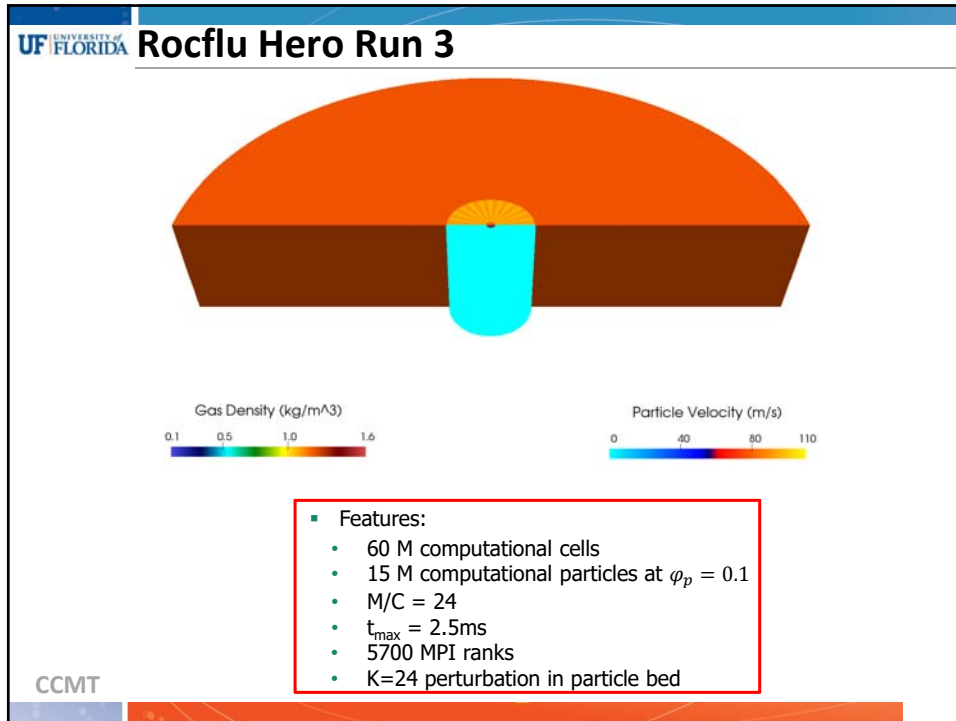
UF UNIVERSITY OF FLORIDA Simulation Setup

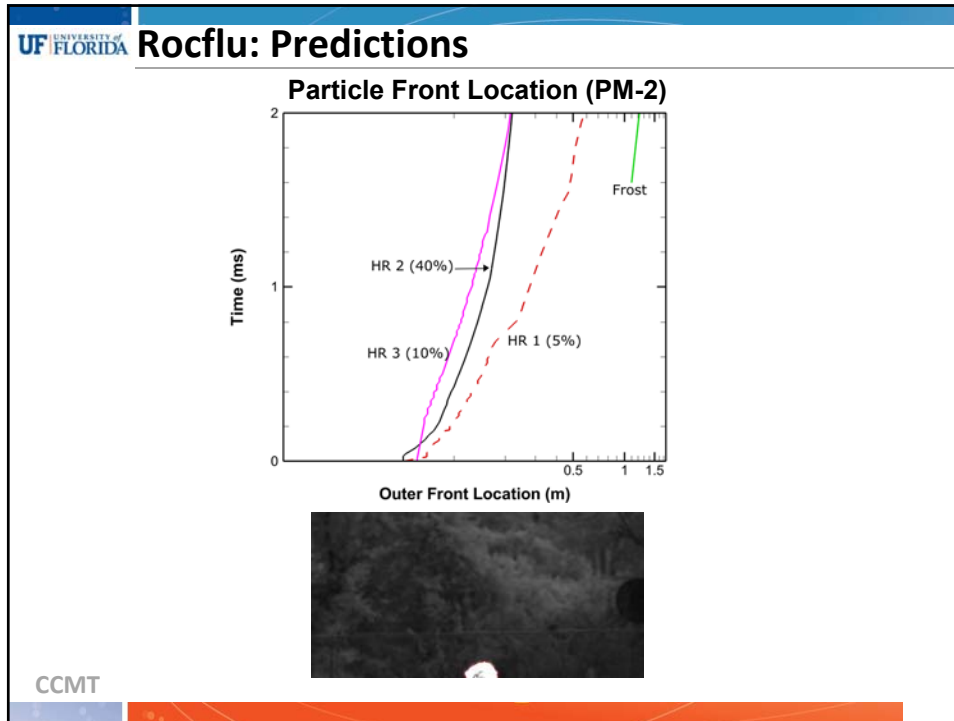
Particle cloud
Ambient air
Explosive Charge

$2r_0$, $2r_1$, L_z

Parameter	Value
ρ_B^{HE}	1712 kgm ⁻³
ρ_B^{Air}	1.203 kgm ⁻³
ρ^p	8050 kgm ⁻³
r_0	3.8 mm
r_1	20 mm
L_z	1.5 mm

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UF UNIVERSITY OF FLORIDA CMT-nek Transition Workshop

A Boot Camp on CMT-nek

November 29, 2017

Organizers: Bertrand Rollin and Jason Hackl

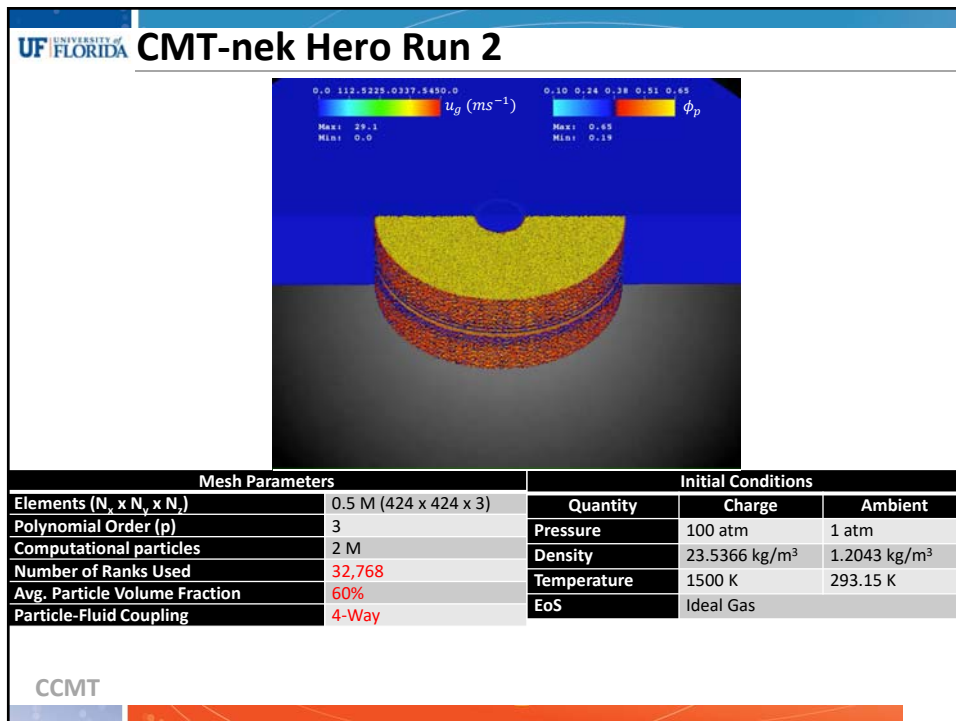
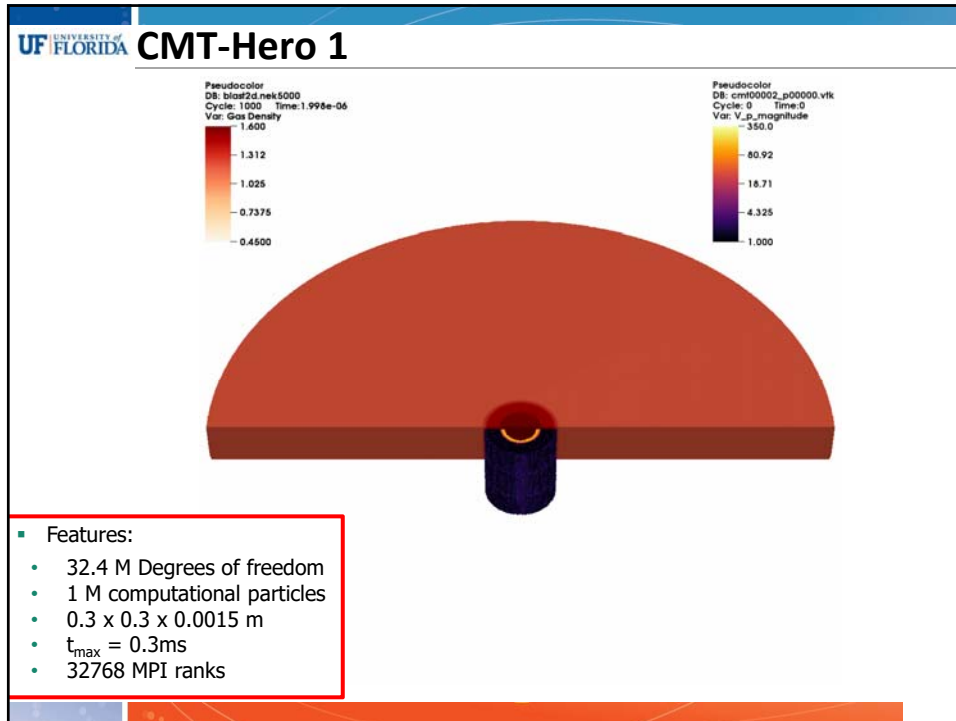
Agenda

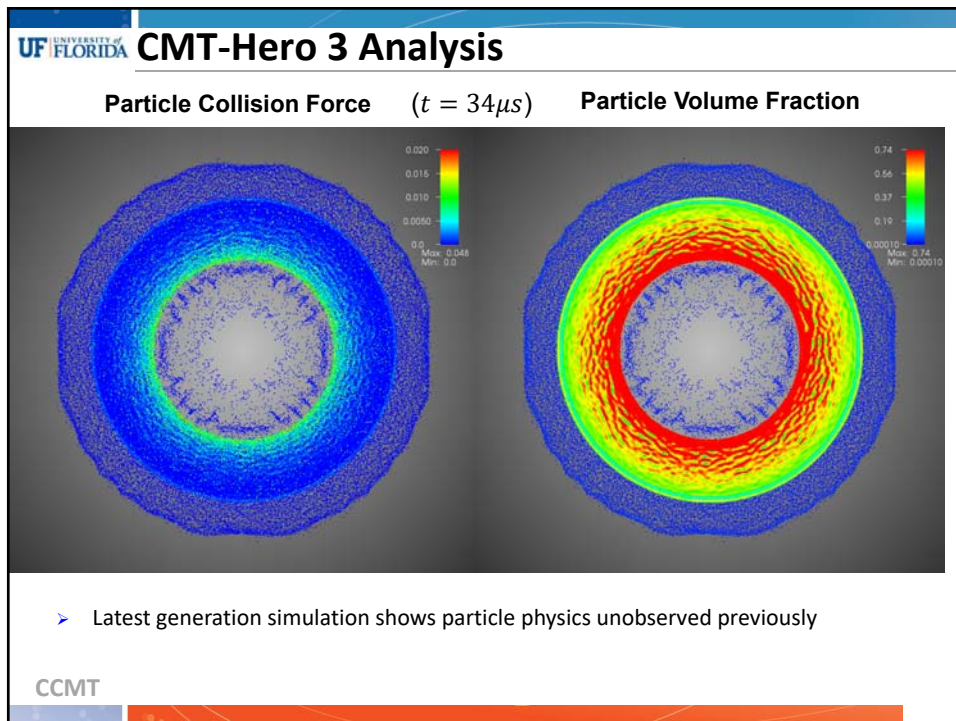
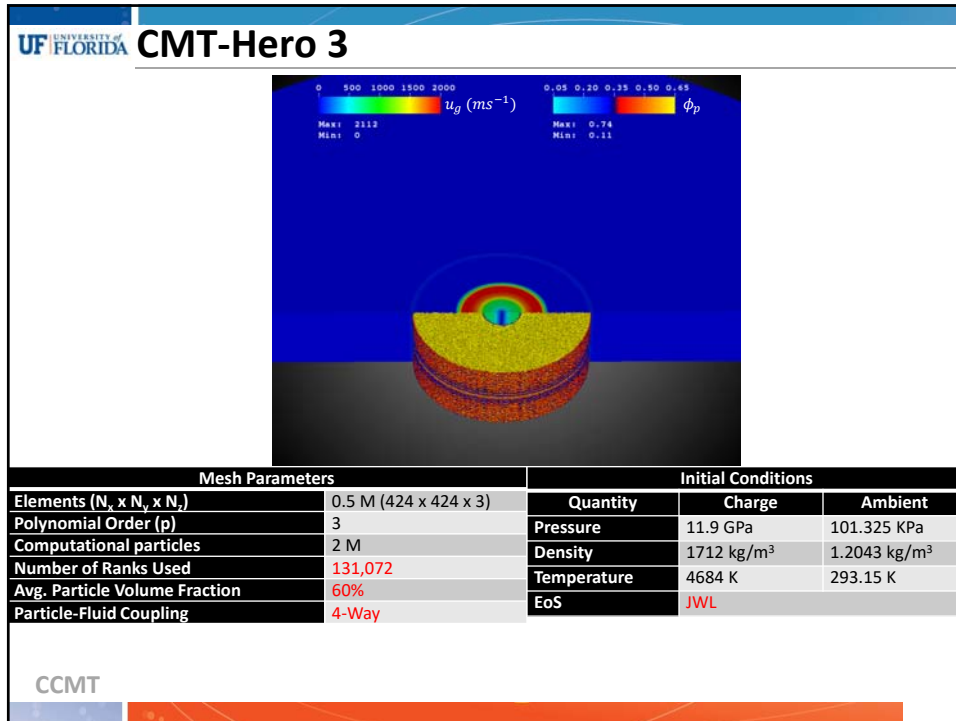
- **CMT-nek: Anatomy of the Beast**
Speaker: Jason Hackl | 1:30 pm – 2:15 pm
- **Lagrangian Particles in CMT-nek**
Speaker: David Zwick | 2:15 pm – 2:45 pm

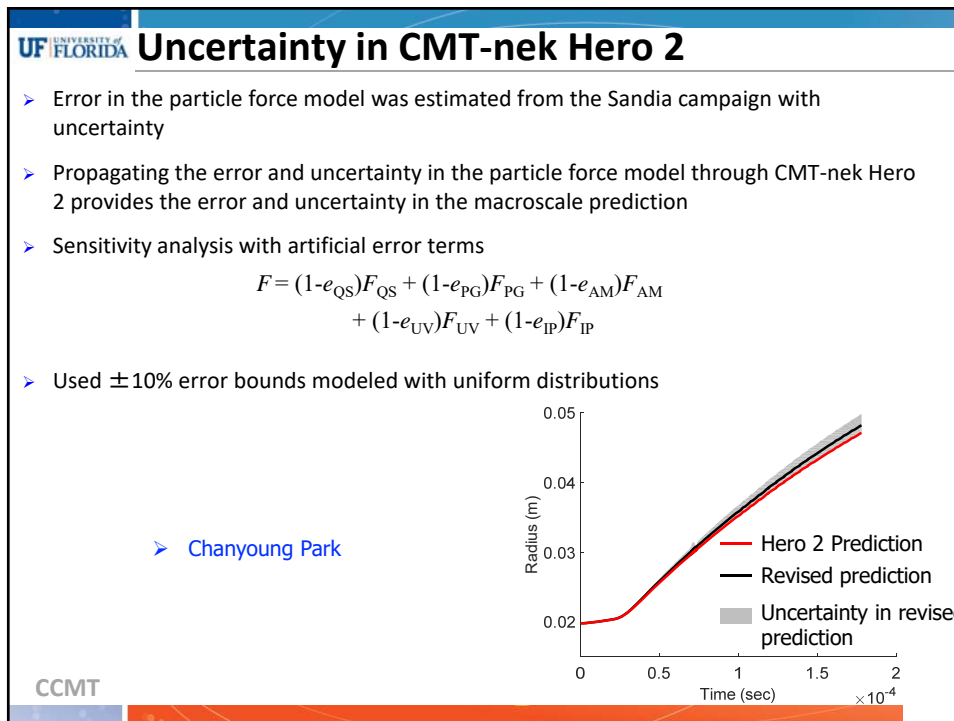
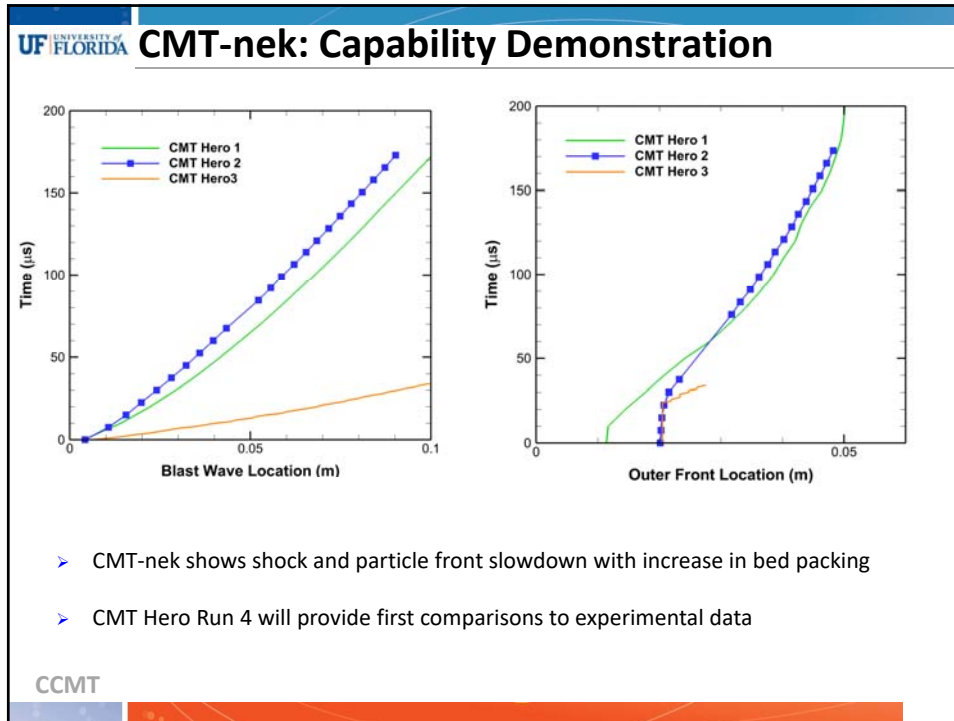
Break

- **Running CMT-nek**
Speakers: Goran Marjanovic and Brad Durant | 3:00 pm – 3:45 pm
- **Post-Processing and visualization in CMT-nek**
Speakers: David Zwick and Brad Durant | 3:45 pm – 4:15 pm
- **Lesson Learnt from CCMT's Simulations**
Speakers: Fred Ouellet and Yash Mehta | 4:15 pm – 4:45 pm

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UNIVERSITY OF FLORIDA **The Channeling Instability: Particle Speed**

- ▶ Particles in radial sectors with low particle volume travel faster
- ▶ At later times the outer front of particles location flips the initial particle volume profile
- ▶ The number of instabilities forming is directly dependent on the initial modal perturbation

Gas velocity (left) and particle volume fraction (right) contours at final time ($t=500\mu s$) for the unimodal case $k=10$. Black contour represents initial ($t=0$) unimodal particle volume contours. White contours are schematics of the final ($t= 500\mu s$) outer particle front location

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UNIVERSITY OF FLORIDA **The Channeling Instability: Particle Migration**



- ▶ Animations below show particle volume contours for three different initial perturbations
- ▶ The animations show the migration of particles from sectors with lower PV to sectors with higher PV as time goes

Unimodal Perturbation
 $k = 8$

Trimodal Perturbation
 $(A_1, A_2, A_3, k_1, k_2, k_3, \Phi_{12}, \Phi_{13})$
 $(0.04, 0.12, 0.07, 1, 12, 8, 6.00, 0.38)$

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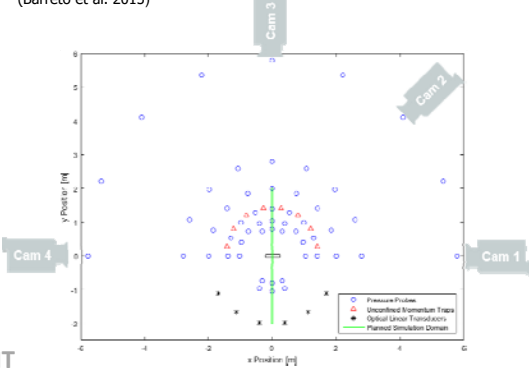
UF UNIVERSITY OF FLORIDA **Eglin Blastpad Experimental Setup**

(Barreto et al. 2015)

- Six shots
 - 2 bare charges
 - 1 charge w/tungsten (n)
 - 2 charges w/steel (n)
 - 1 charge w/steel (u-n)

➤ Angela Diggs and Kyle Hughes

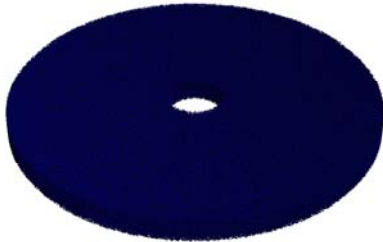


- Instrumentation:
 - 54 pressure probes
 - 8 Momentum traps
 - 4 high speed video cameras
 - 6 Linear optical transducers

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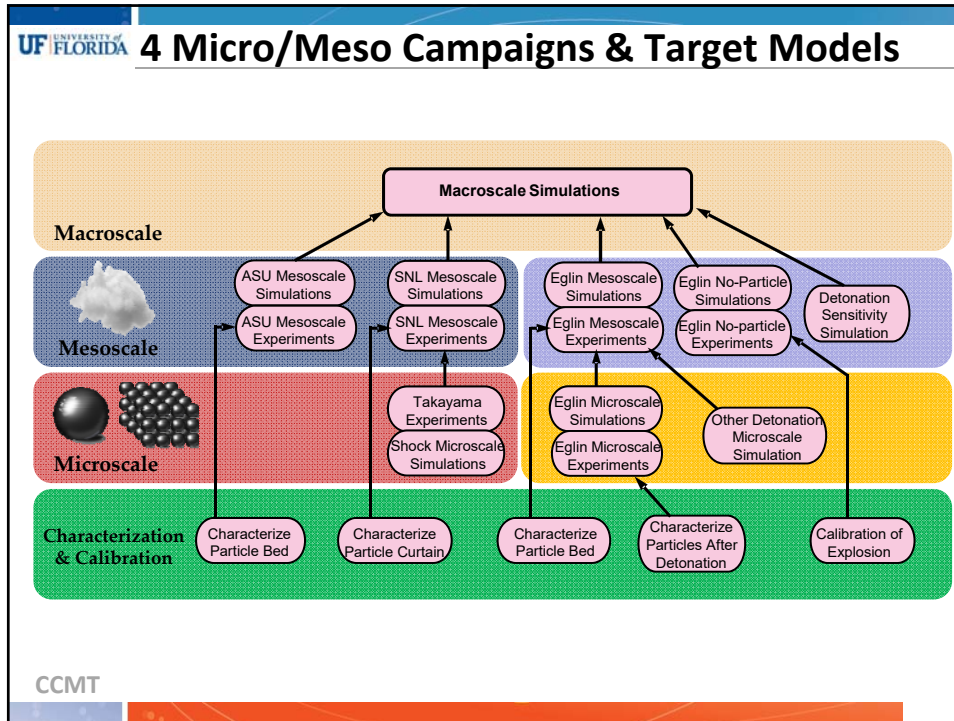
UF UNIVERSITY OF FLORIDA **Demonstration Problem – A Summary**

- ✓ Last Rocflu Hero Run completed
- ✓ UQ analysis continues
- ✓ Study of Physics of the Demonstration problem continues
- ✓ CMT-Hero 1 completed
- ✓ CMT-Hero 2 completed
- ✓ CMT-Hero 3 completed
- ✓ Blastpad tests in progress



➤ David Zwick

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Eglin Microscale/Mesoscale Experiments

- Eglin Microscale campaign is a joint effort between UQ, experimental and simulation teams
- More details in upcoming talks:
 - Kyle Hughes (Experiments)
 - Chanyoung Park (UQ)
 - Josh Garno (Simulations)

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Eglin Microscale Simulations

Overhead schematic of the test set-up

Simulation Parameters	
Cells	5 M
Barrel Radius	6.35 mm
Charge length	3.81 cm
Particle-Fluid Coupling	1-Way

Particle Parameters	
Diameter	1.992 to 2.04 mm
Density	15.59 to 16.01 g/cc
Vertical position from axis	0, 0.125, 0.254 mm

Single Particle: • Current Work

Ring of Particles: ••••• Future Work

Grid of Particles: ••••• Future Work

➤ Josh Garno

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Mesoscale Simulation of Mesoscale Experiment

P-Rad Mesoscale Experiment (x-ray)

Simulation Details:

- Real gas equation of state (JWL)
- Point-particle model
- Explosive modeled with a reactive burn initial profile

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***Do you have any
questions?***



Experimental Studies of Gas-Particle Mixtures Under Sudden Expansion at ASU

Heather Zunino

PhD Candidate

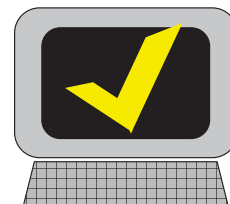
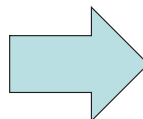
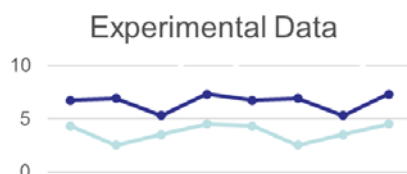
Dr. Ronald Adrian

Advisor



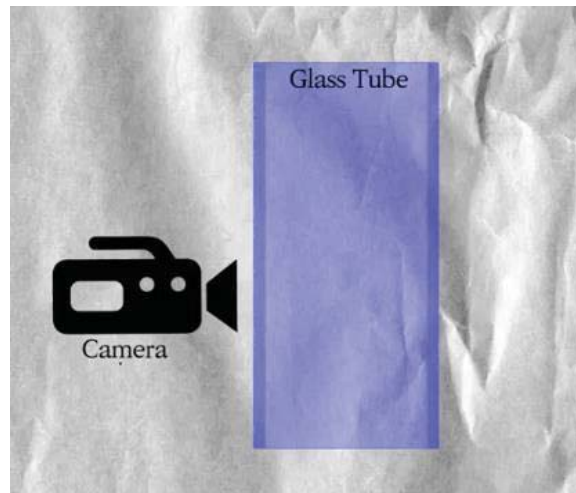
Problem Statement and Goals

- Experimental multi-phase studies involving compressible flow are complicated
 - Air and solid particles may move separately
 - Particles generate turbulence
- Need for a simple 1D flow experiment that can be used for early validation of the computational codes developed by the PSAAP center.
- Simpler physics involved than the PSAAP capstone experiment
- Perform experiments on existing shock tube setup
- Examine expansion fan, flow structures, turbulence, and instabilities
- Provide data for early-stage validation of computational codes developed by the PSAAP Center



Experiment Description

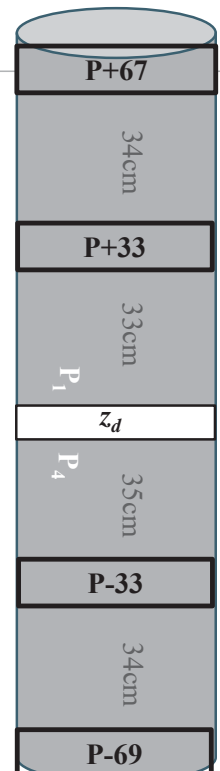
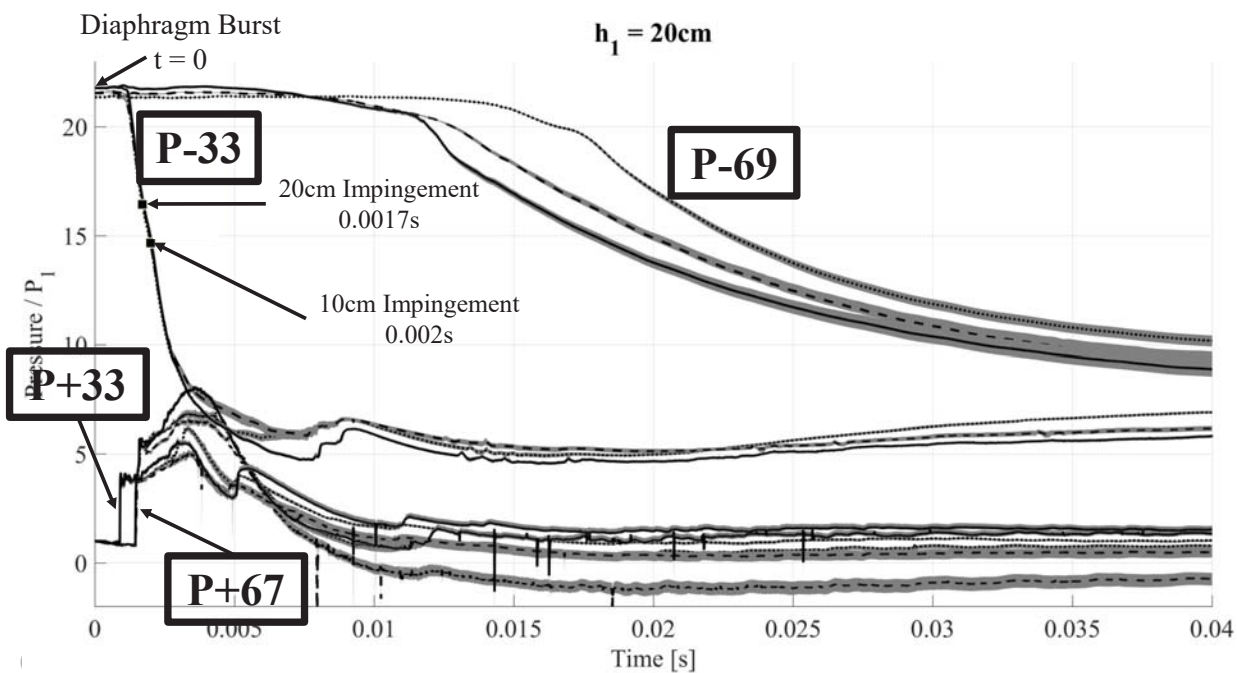
- 1 meter glass tube
 - Cylindrical footprint
 - Inner diameter: 3.9cm
- Particle bed
- Diaphragm
 - Tape
- High-speed camera
- Measurements
 - Gas velocity
 - Particle volume concentration
 - Particle interface
- Parameters: particle size, bed height, and pressure ratio



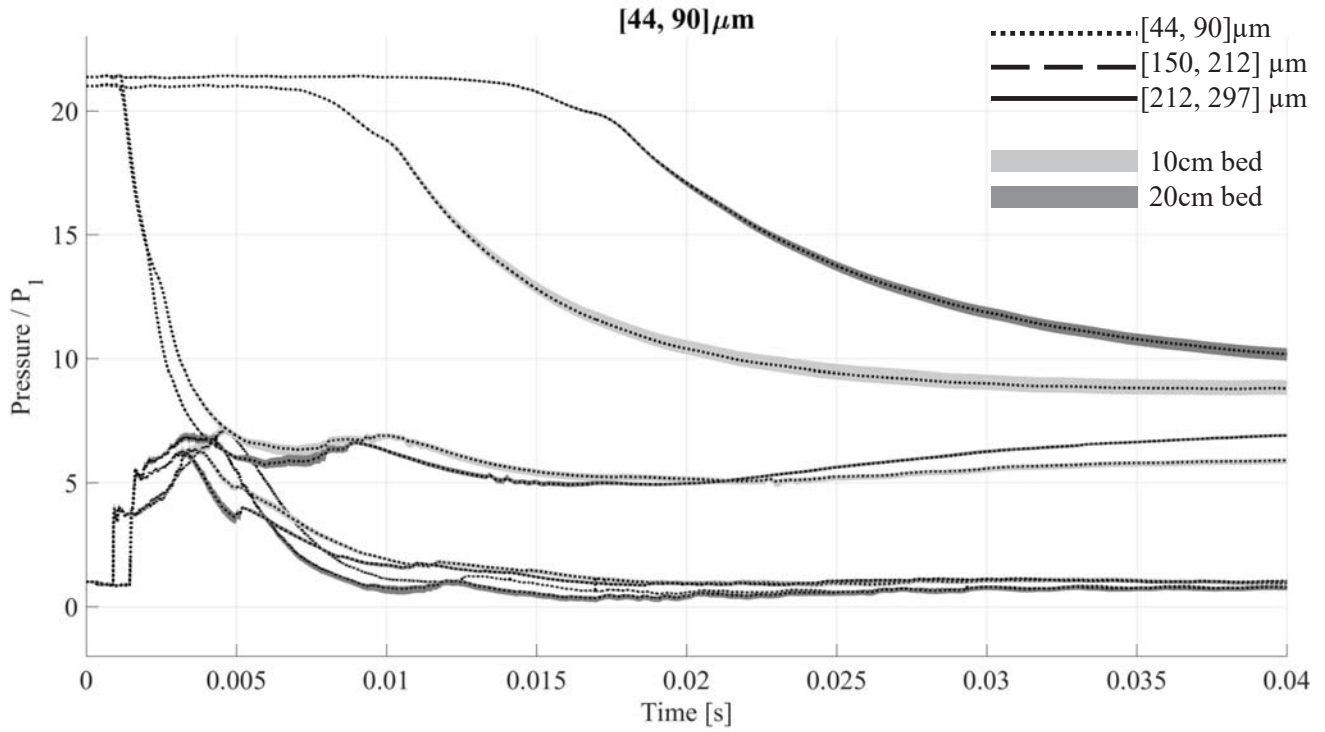
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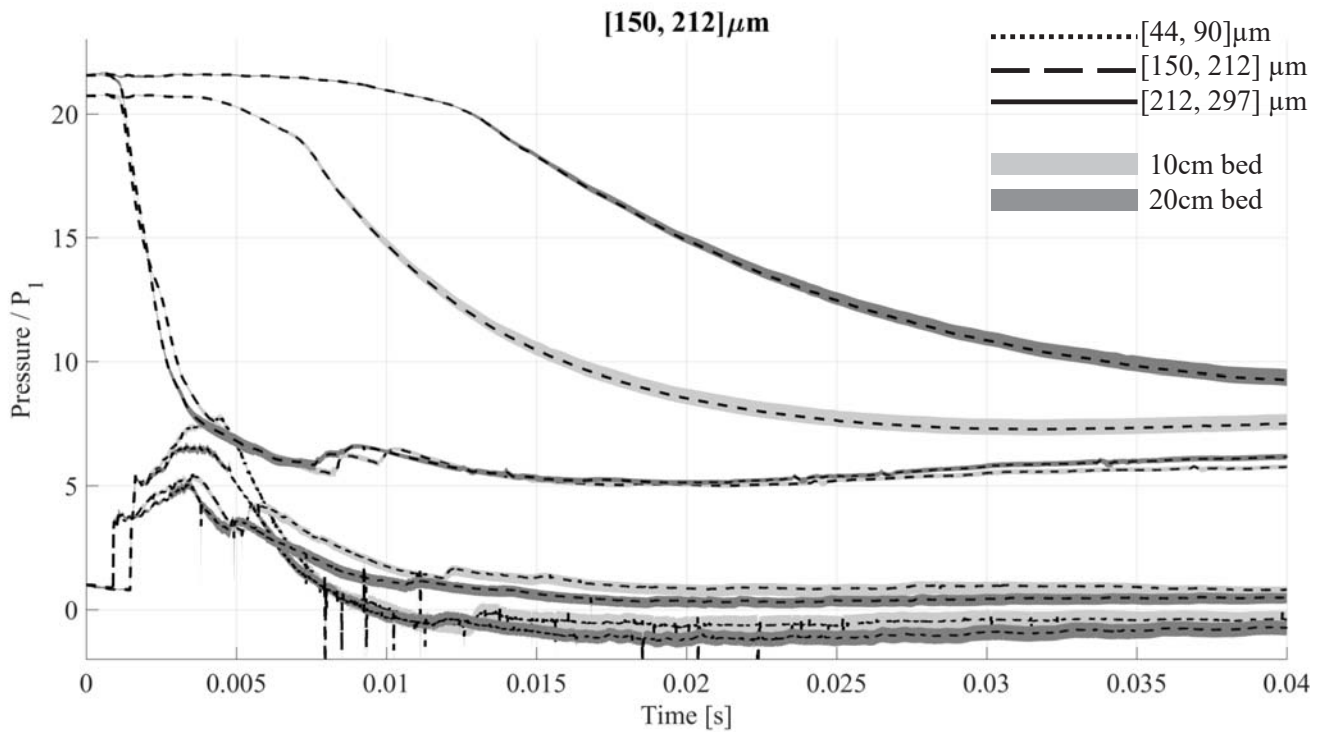
Expansion Waves and Timing Diagram



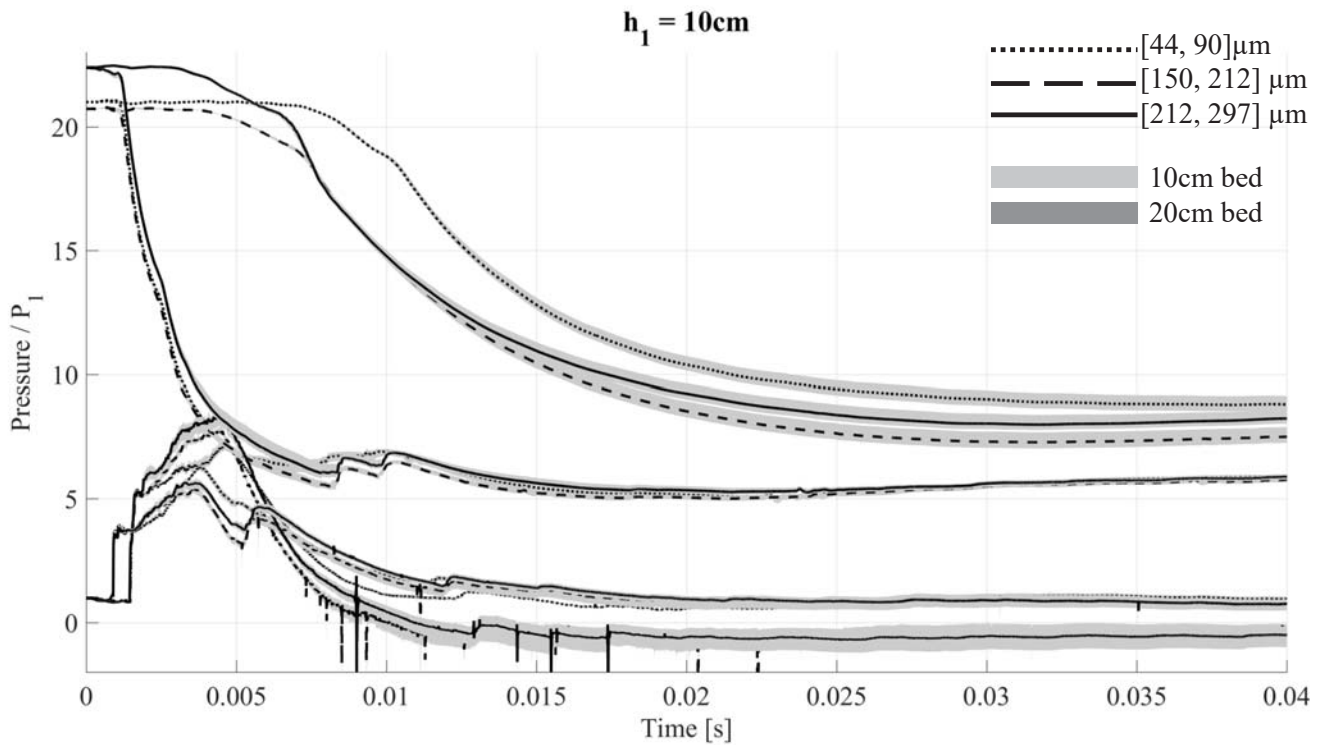
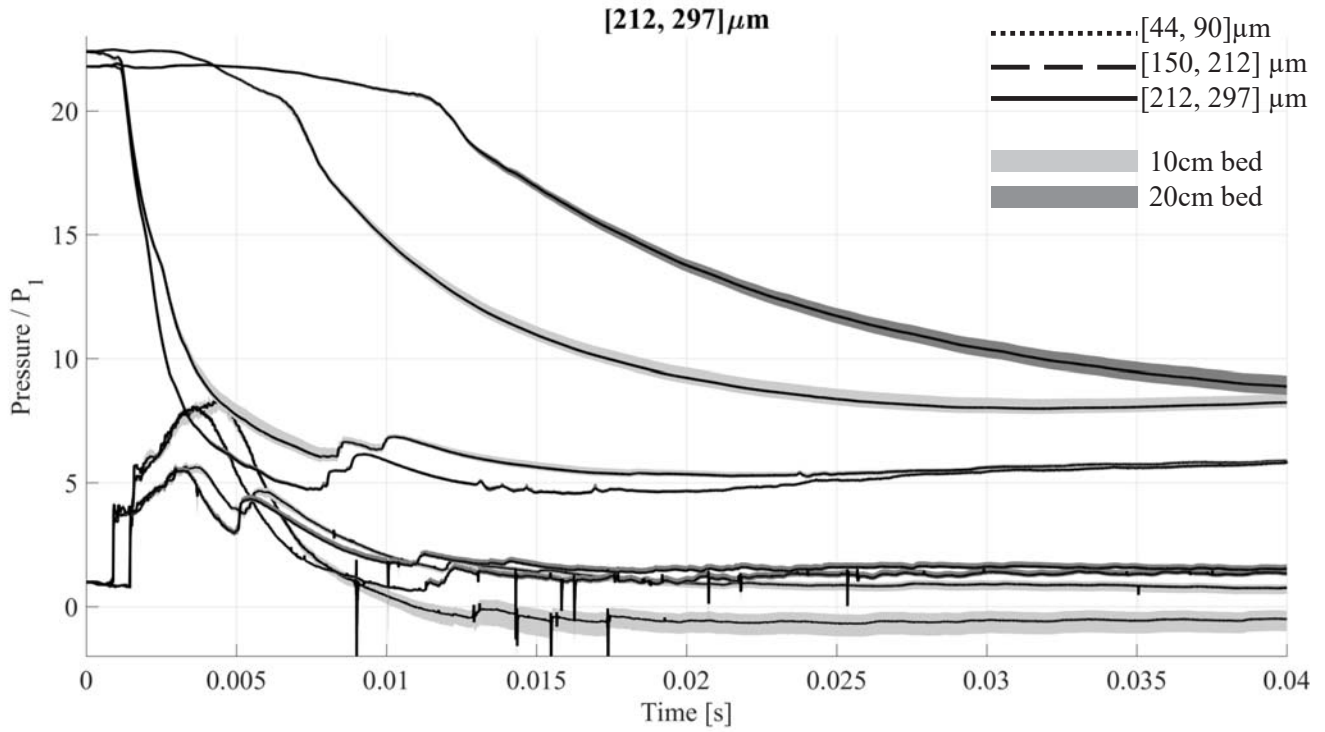
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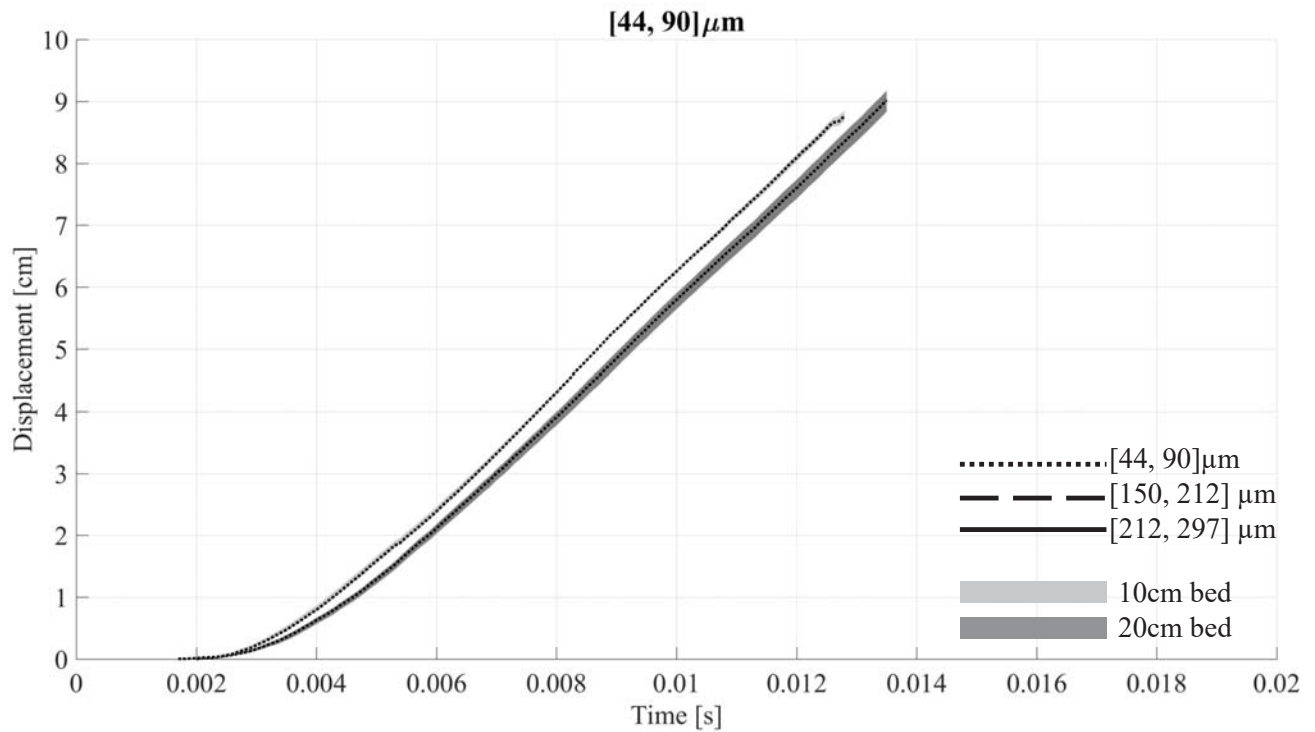
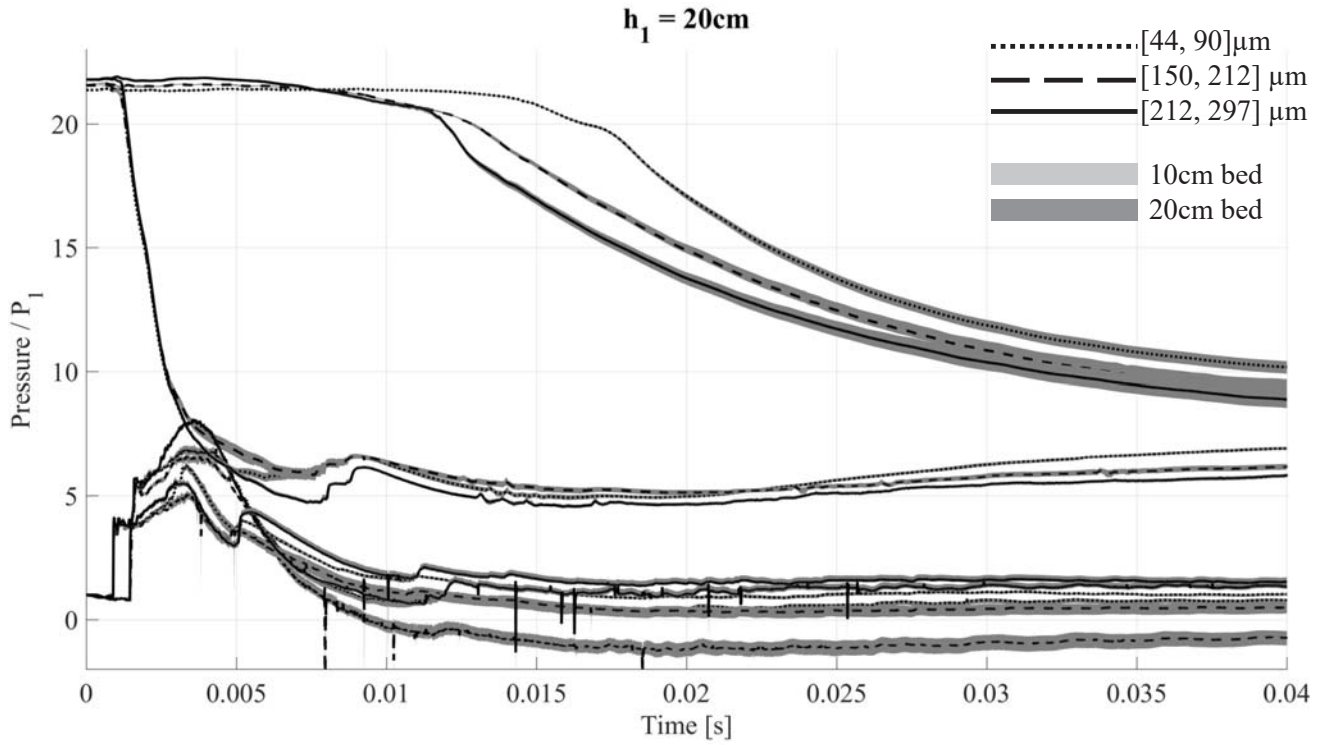


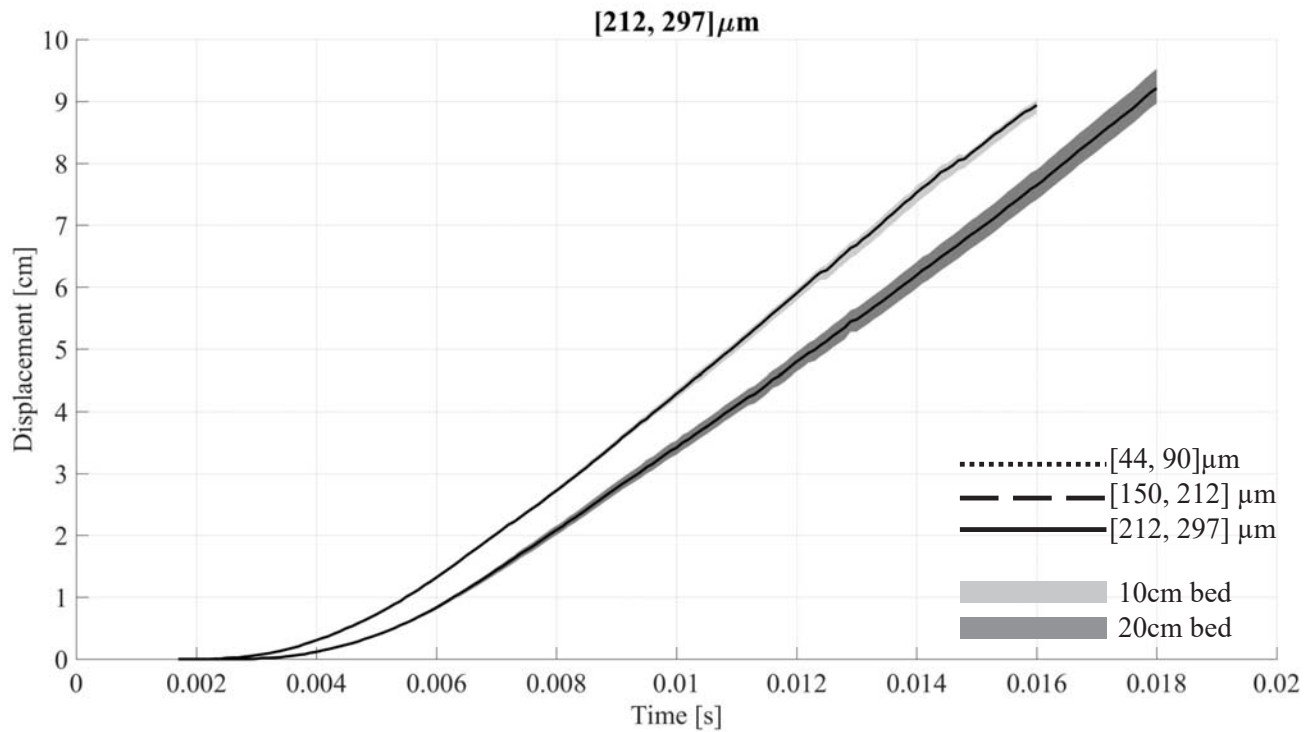
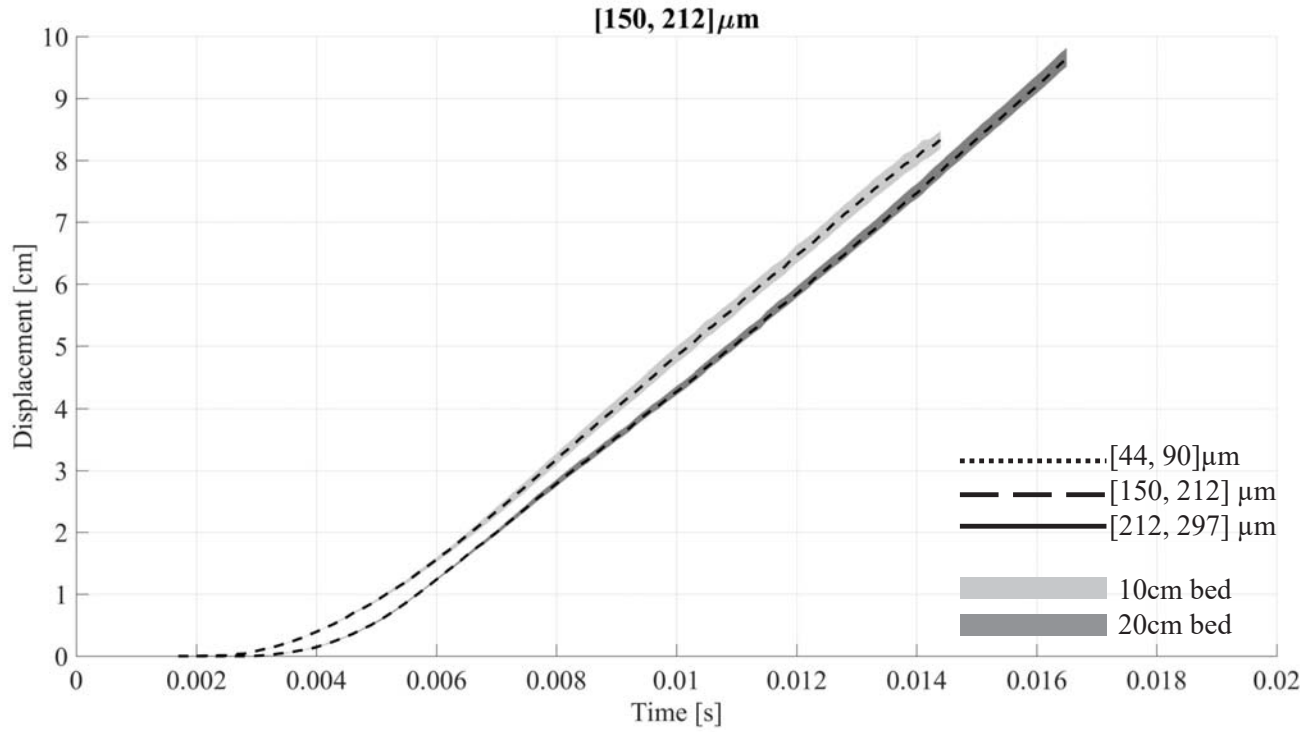
CCMT

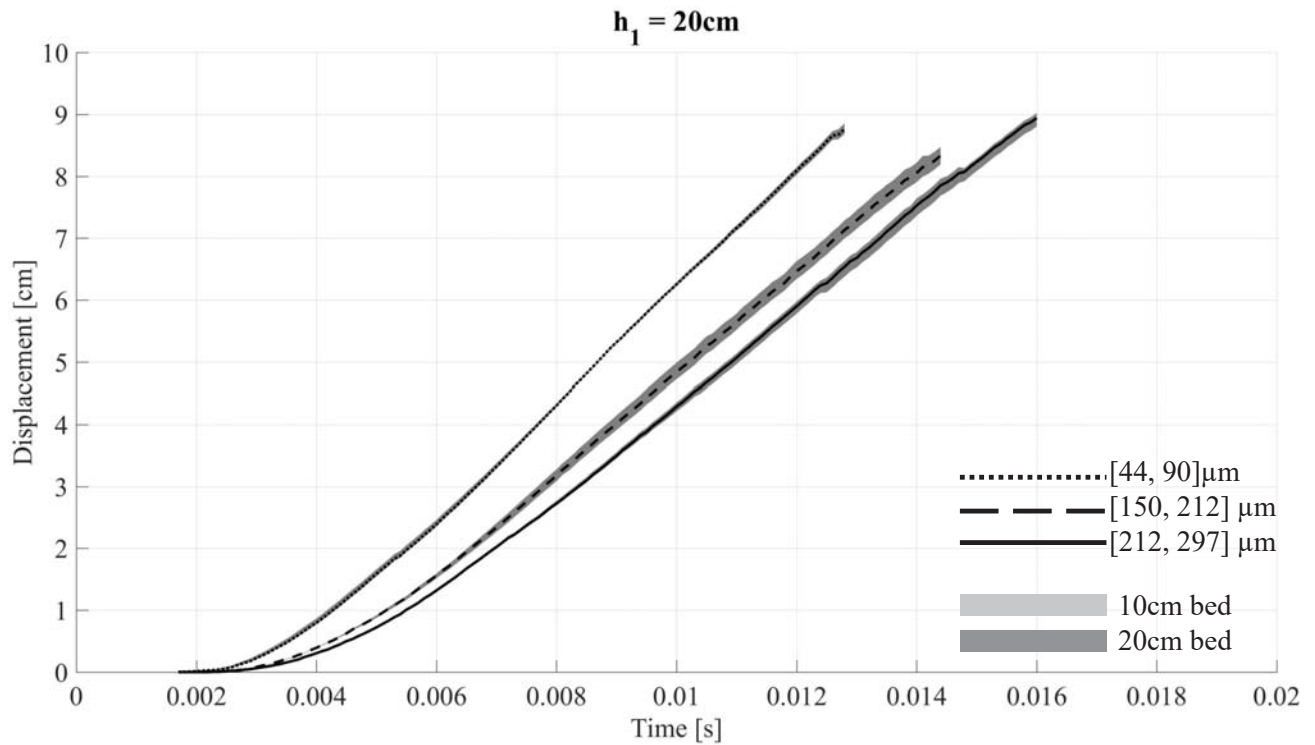
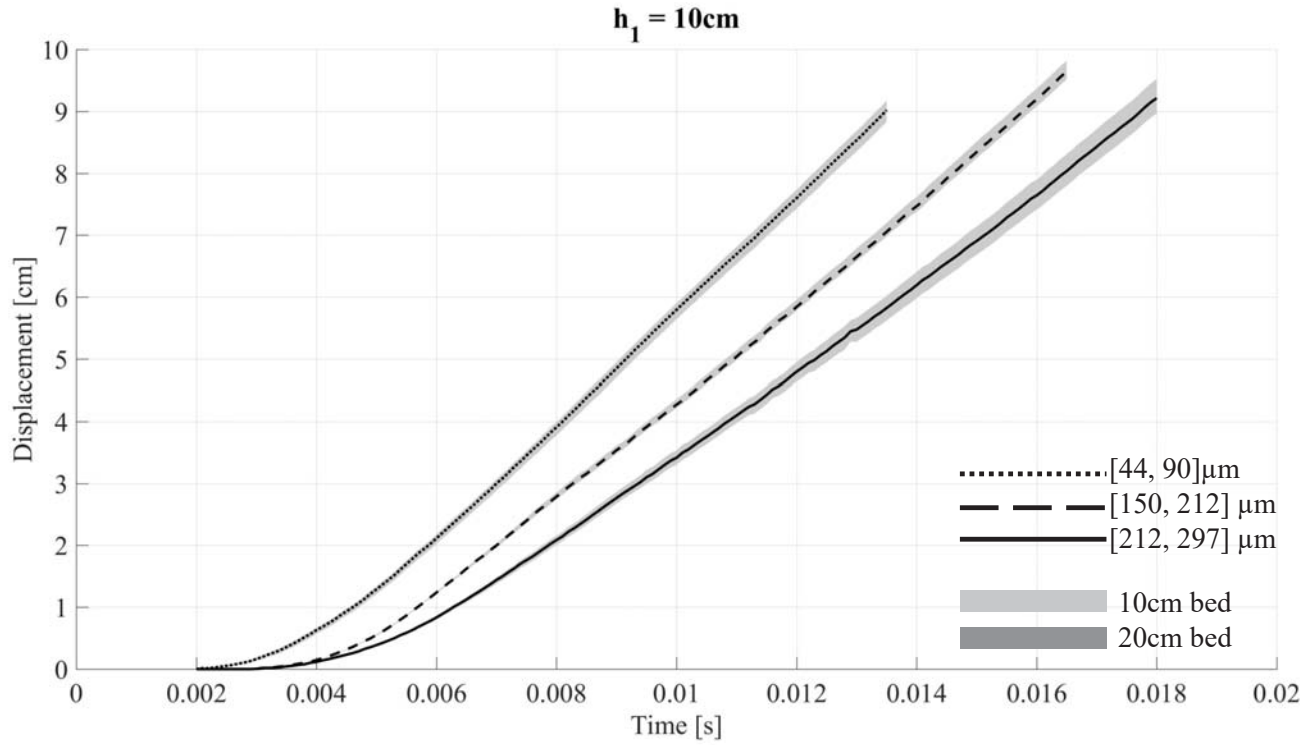


CCMT



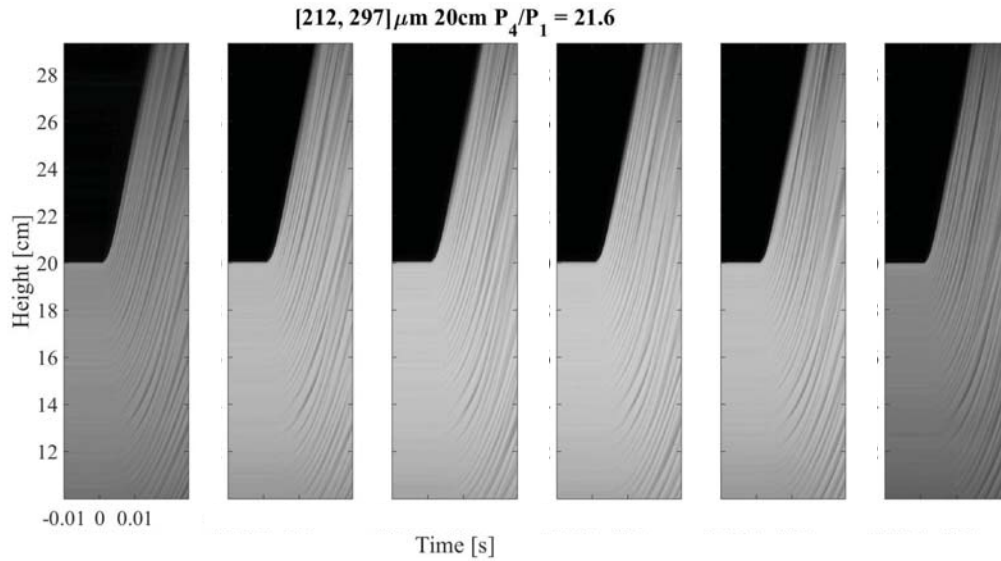






Streak Images [212, 297] μm

- Width of tube: 6 sections, each column is intensity averaged over 20 pixels

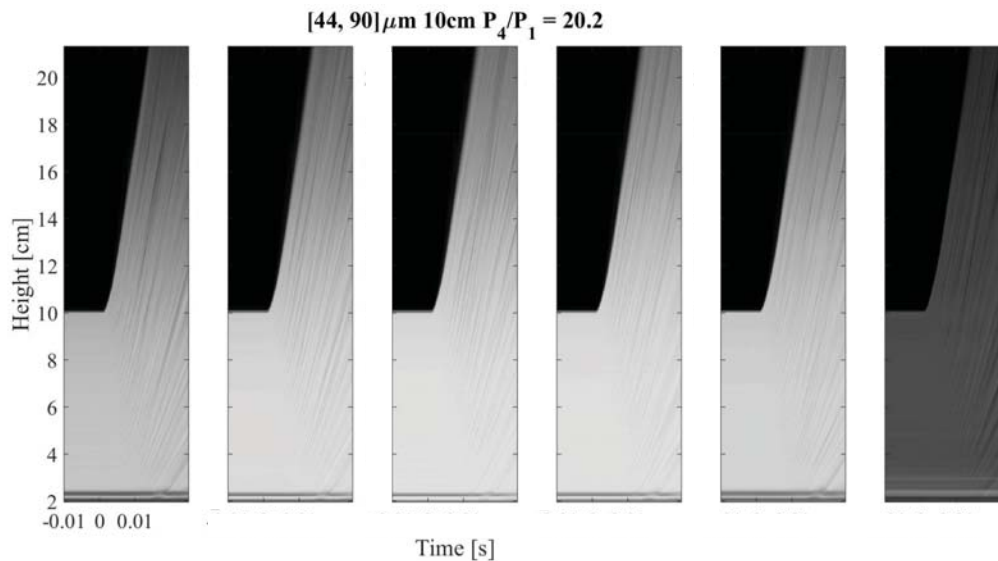


CCMT

15

Streak Images [44, 90] μm

- Width of tube: 6 sections, each column is intensity averaged over 20 pixels



CCMT

16

To complete the story...

- Explore various non-dimensionalizations to try to collapse the curves presented
- Develop a more complete description of the horizontal void cracks and void formations
- Compare data to available models
- Should more data be required...
 - A fourth bead diameter is available
 - A third bed height
 - For some experiments, can add in an additional pressure ratio
 - Already have some experiments with varying pressure ratios
 - Varying pressure ratio may reveal different regimes or dominating effects

CCMT

Validation Experiments: Microscale, Mesoscale, and Macroscale Experiments

Kyle Hughes, Kathy Prestridge
Los Alamos National Laboratory

Angela Diggs, Don Littrell, Mike Jenkins, Chi Mai
Munitions Directorate, Eglin AFB, FL

Nam-Ho Kim, Raphael Haftka, Chanyoung Park
University of Florida

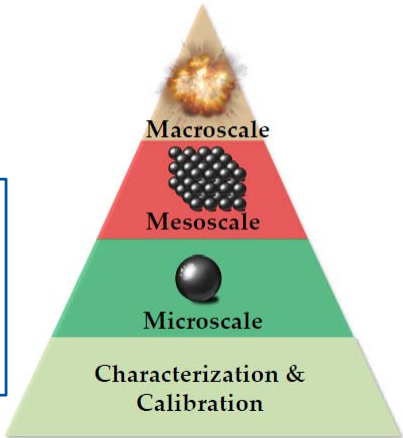
December 13, 2018



UF UNIVERSITY OF FLORIDA

Validation Hierarchy

- Explosive dispersal is explored at three different regimes of complexity based on the number of particles
- Microscale** ($O(1-10^2)$ particles)
 - Validation of microscale drag models
 - Explosive tests conducted October 2014 and February 2015
- Mesoscale** ($O(10^3 - 10^6)$ particles)
 - Examination of instabilities, volume fraction effects, explosive dispersal at reduced scales
 - Eglin AFB explosive tests conducted November 2015
 - LANL tests conducted October 2017 and November 2018
- Macroscale** ($O(10^7)+$ particles)
 - Validation of full simulation
 - AFRL blastpad conducted July 2017




CCMT

2

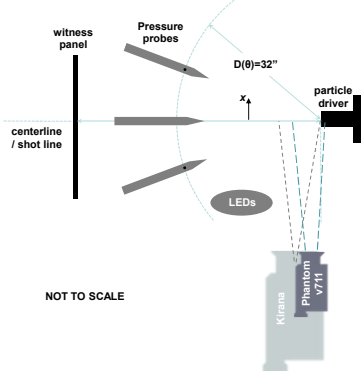
UF UNIVERSITY OF FLORIDA **First Iteration: Eglin AFB November 2015**

PI: Mike Jenkins



Side view photograph of the experimental setup.

Source: Mike Jenkins, Shot Log, 11/5/2015



Overhead schematic of the test set-up.


Summary of mesoscale shots of interest.

Test #	Driver	Particles
Nov15-1	RP-80 + 1 N5	0.5 g Tungsten powder
Nov15-2	RP-80 + 1 N5	0.5 g Tungsten powder
Nov15-3	RP-2 + 1 N5	0.5 g Tungsten powder
Nov15-4	RP-80 + 1 N5	0.5 g Tungsten powder
Nov15-5	RP-80 + 1 N5	0.5 g Tungsten powder
Nov15-6	RP-80 + 1 N5	0.5 g Tungsten powder
Nov15-7	RP-80 + 2 N5	0.5 g Tungsten powder

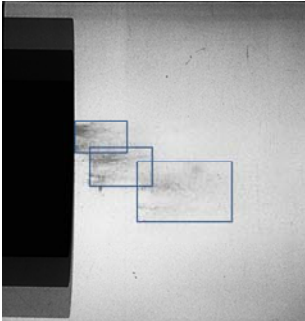
3

UF UNIVERSITY OF FLORIDA **First Iteration: Forensic Results**

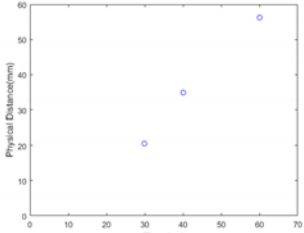
- 4"-diameter casing and smaller amount of explosive prevented fragmentation like in the microscale explosive tests (approximately 15% increase in bore diameter)
- 0.5 g of tungsten particles, sized 100-130 μm , is loosely packed within tissue paper and placed flush with the casing
 - The packet had gaps around the edges that may have caused some jetting
 - Unknown geometry and unknown volume fraction ($6\% < \phi < 60\%$)
- Particle front positions difficult to discern from multiple exposure x-ray



Tungsten particles packed into a tissue and placed within the casing
CCMT



Multiple exposure x-ray film



Downstream particle front position and velocities

Time (μs)	Position (mm)	Velocity (m/s)
30	20.6	-
40	34.8	1423
60	56.3	1074

4

Second Iteration: LANL October 2017

- Tests used same casing and explosive train as used by the Eglin experiments
- Examination of detonation physics with a reduced number of particles (to reduce computational burden)
- Tests performed in vacuum to isolate the effect of contact discontinuity
- Proton radiography acts an improved diagnostic and volume fraction reduced with hollow glass microspheres

The diagram shows an expanded cross-sectional view of the test article. A proton beam (x3 magnifier) enters from the left through a field of view (FOV) into a steel casing. Inside the casing, there is an adapter nut, a particle packet, PBX-9501 explosive, and an RP-80 detonator. The entire assembly is in a vacuum environment. To the right, a photograph shows the assembled test article with alignment fiducials.

Expanded cross-sectional view of the test article with proton radiography field of view.

Assembled test article with alignment fiducials

Particle Packet

- Steel particles characterized through SEM, helium gas pycnometer, and CT scans
- Hollow glass microspheres (HGM) are used to artificially lower the volume fraction
 - The released energy of the explosive is approximately 15.4 kJ
 - Back-of-the-envelope calculations estimate the HGM bed is crushed with 0.7 J, (<0.01% of released explosive energy)

The diagram illustrates the construction of a particle packet. It shows three stages of assembly, each involving a layer of particles and a paper divider. The first stage uses steel particles (115 ± 23 μm) and results in a 66% volume fraction (2 shots) with a mass of 1.29 g. The second stage uses a combination of steel particles and hollow glass microspheres (HGM) and results in a 46% volume fraction (1 shot) with a mass of 0.82 g. The third stage uses only hollow glass microspheres (107 ± 12 μm) and results in a 26% volume fraction (2 shots) with a mass of 0.41 g. SEM images of the particles are shown on the left and right.

Steel particles (115 ± 23 μm)

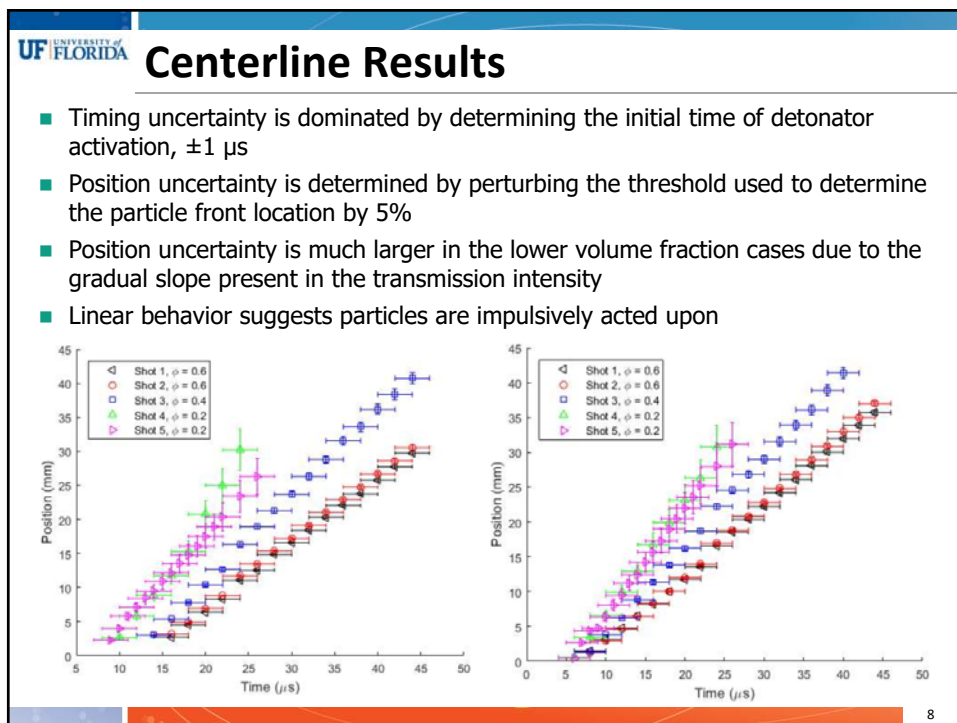
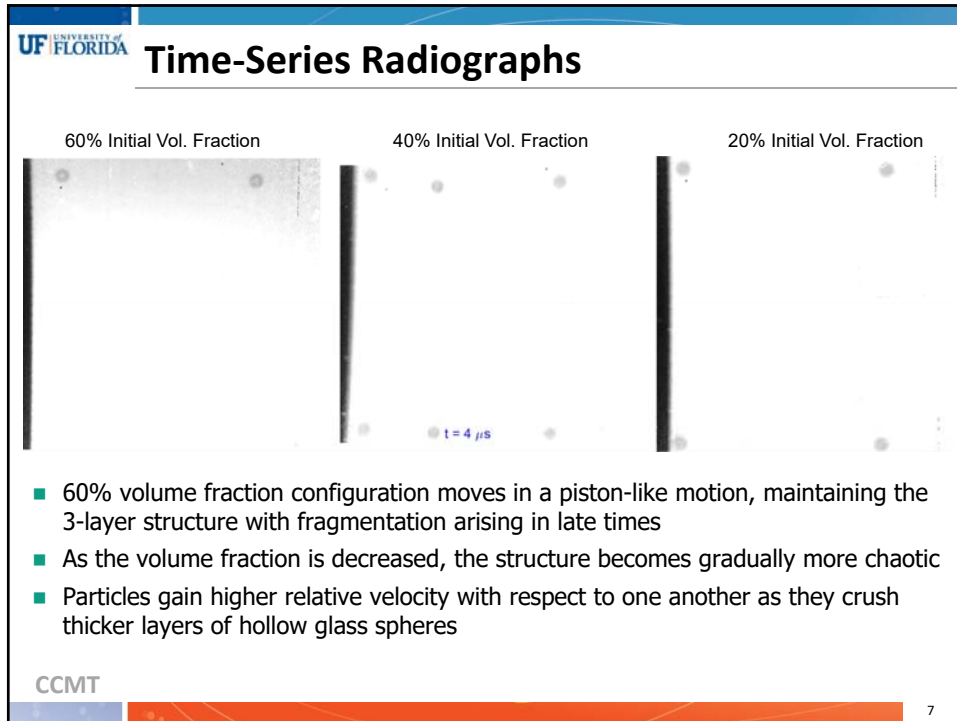
Hollow glass microspheres (107 ± 12 μm)

66% Volume Fraction (2 shots) 1.29 g

46% Volume Fraction (1 shot) 0.82 g

26% Volume Fraction (2 shots) 0.41 g

CCMT



Third Iteration: LANL November 2018

- **Goal:** Investigate the physics of shock-contact interactions with high-volume fraction bed of particles by varying carrier fluid
- **Unique contribution:** Well-characterized bed is subjected to both shock and contact wave. P-rad provides high-fidelity measurement of particle front positions and volume fraction.

Previous Experiments

Poorly-characterized particle bed
Carrier Fluid: Air

Explosive products contact
Shock

Well-characterized particle bed
Vacuum

Explosive products contact
Shock

New Experiments

Well-characterized particle bed
Varying carrier fluid: Xenon, Air, SF₆

Explosive products contact
Shock

Volume fraction fiducial to allow extraction of time-varying volume fraction

CCMT 9

Preview: LANL November 2018

- Previous setup is now surrounded by an 1/8" aluminum cannister to allow varying the ambient fluid
- Particle packet is packed at 60% volume fraction
- Gas pressure and temperature are approximately STP for all three gases

Gas: Air
 $a_{air} = 346 \text{ m/s}$
 $\rho_{air} = 1.19 \text{ kg/m}^3$

Gas: Xenon
 $a_{xenon} = 174 \text{ m/s}$
 $\rho_{xenon} = 5.29 \text{ kg/m}^3$

Gas: SF₆
 $a_{SF_6} = 123 \text{ m/s}$
 $\rho_{SF_6} = 5.86 \text{ kg/m}^3$

CCMT 10

Validation Hierarchy

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- Macroscale** ($O(10^7)+$ particles)
 - Validation of full simulation
 - AFRL blastpad conducted July 2017

CCMT

11

Test Facility and Planned Simulation Domain

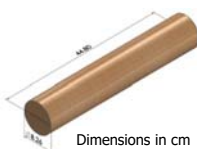
- Eglin Blastpad chosen as the test site due to additional camera views and large number of pressure transducers (Barreto et al. 2015)
- Instrumentation suite consisting of 54 in-ground pressure transducers (sampled at 1 MHz), 6 optical linear encoders, 8 unconfined momentum traps, and 4 high-speed cameras
- Phantom v1212 sampled at 12000 fps (Camera 1/4) and Phantom v711 (Camera 2/3) sampled at 7500 fps

CCMT


12

Design of the Test Article

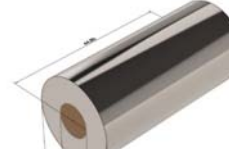
- Literature data consulted to ensure the large scale jetting instability will arise
- Test article designed to match Eglin legacy blast pad data to increase the statistical significance



a) Bare charge
(Mass = 4.1 kg)



b) Charge w/ tungsten particles (M/C = 10)

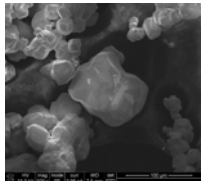


c) Charge w/ steel particles (M/C = 13)

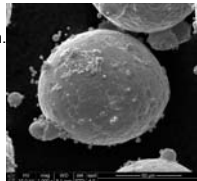
Dimensions in cm

- Steel particles were selected from possible vendors based on sphericity and tight size distribution.

SEM of single tungsten particle, 500x zoom.



SEM of single steel particle at 1000x zoom.




CCMT 13


Minimizing the Casing Influence

- Case fracture may be a possible mechanism for jetting instability (Zhang et al. 2001, Xu et al. 2013)
- Case influence was minimized by using thin phenolic tubing with no inner casing or struts (estimated to require only 0.06% of the released explosive energy to break)
- Notches used to attempt to control the failure mechanism in some of the tests
- Note endcaps had the unexpected effect of slowing the end shocks

Shot	Liner	Notched?
1	-	-
2	-	-
3	Tungsten	Y
4	Steel	Y
5	Steel	Y
6	Steel	N

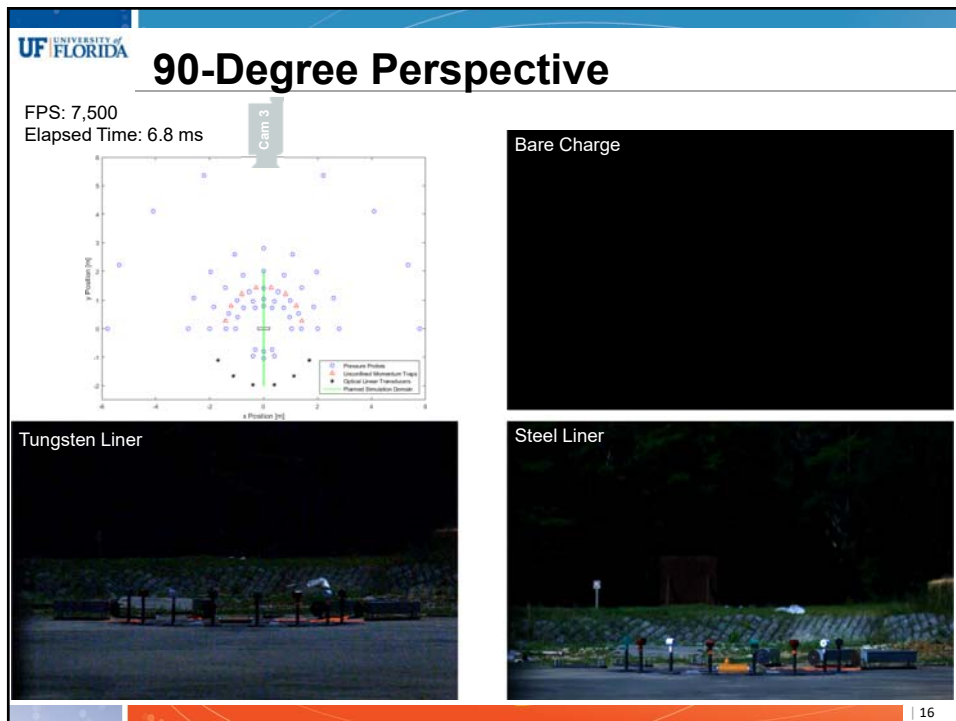
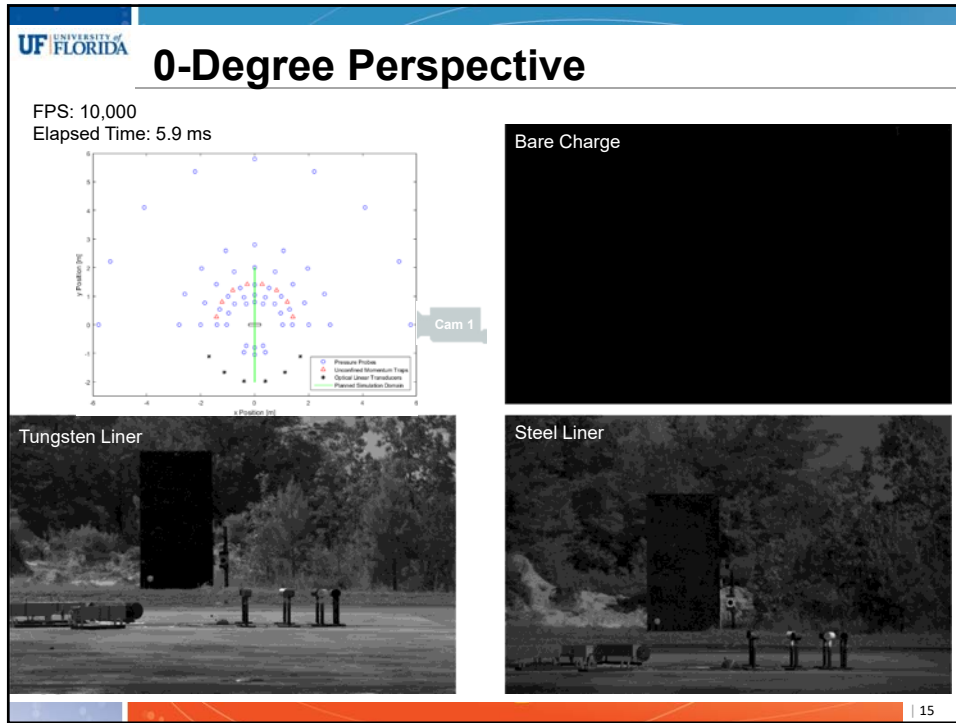


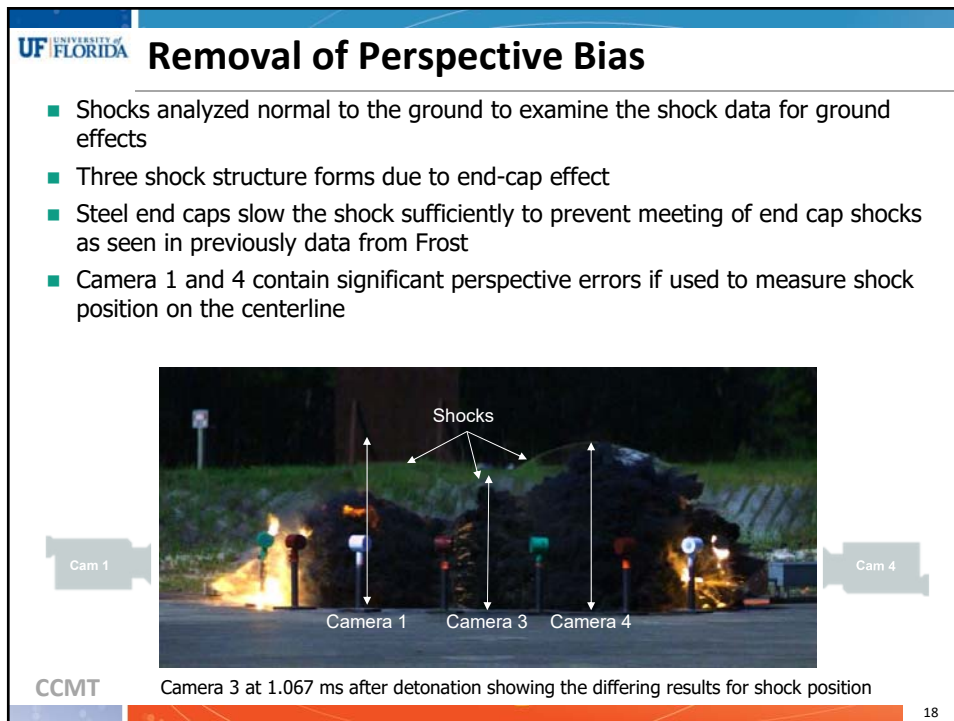
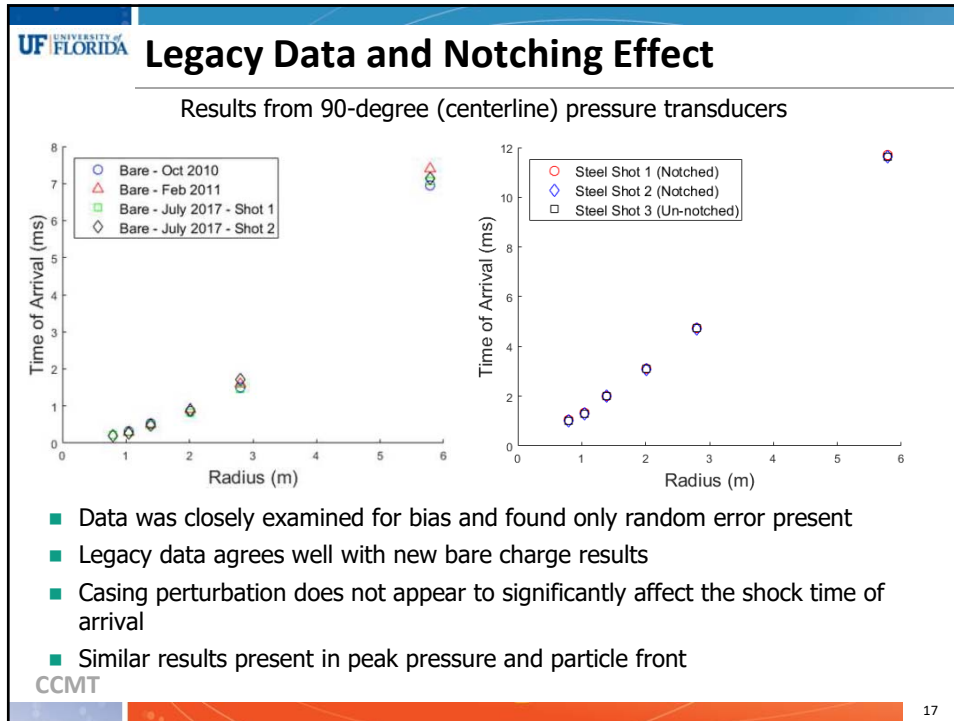
Top view of notched casing (steel liner)

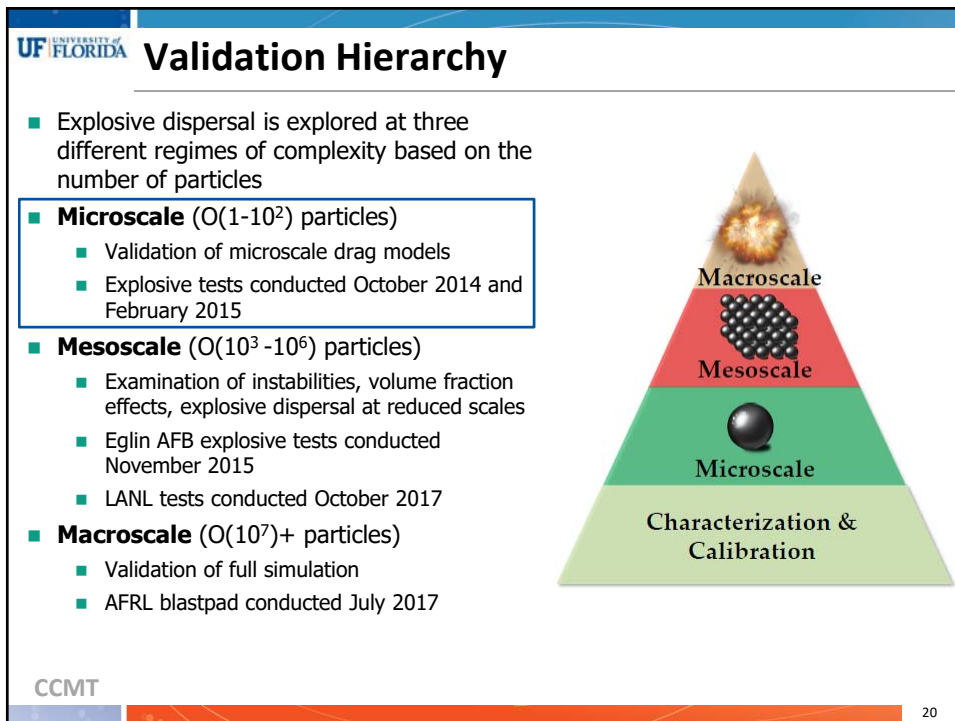
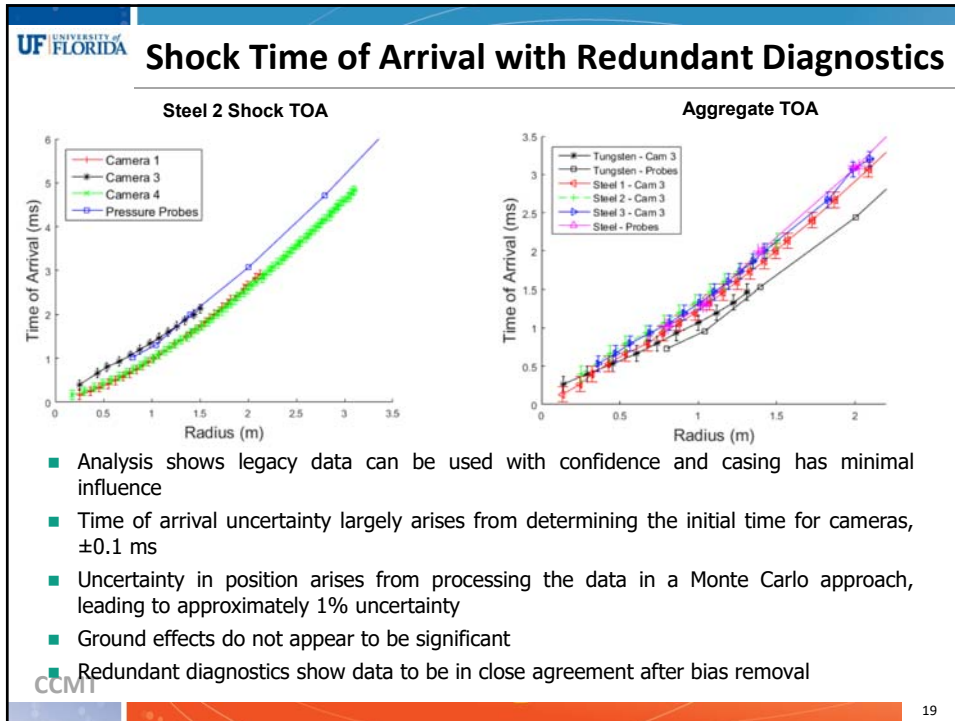


Casing with steel particle liner aligned with test plane

CCMT 14







Third Microscale Iteration: Setup

- Previous feedback from AST/TST shows concern with modeling the microscale tests due to fracture of the casing and low number of repeated tests
- Explosive casing diameter has been increased to 4" to prevent large deformation and fracture

Shot	Items Type
1	Bare
2	Bare
3	Bare
4	Single
5	Single
6	Single
7	Square 2x2
8	Square 3x3
9	Square 4x4
10	Ring of 7
11	Ring of 7
12	Ring of 7

(1) 1/2" x 1/2" booster;
 (2) particle(s);
 (3) detonator adapter nut;
 (4) detonator

CCMT | 21

Third Microscale Iteration: Instrumentation

- High-Speed Videos: Capture shock and explosive contacts
- Pressure Transducers: Moved closer to limit computational domain
- X-ray Imaging: Particle position
- Shadowgraph Imaging: Shock position

NOT TO SCALE

CCMT | 22

UF UNIVERSITY OF FLORIDA **Conclusions**

Microscale Experiments

- Previous forensic investigations allow identification of large uncertainty sources: casing fracture, difficulty in identifying the shock, etc.
- New microscale experiments provide additional diagnostics and design modifications to significantly reduce the uncertainty

Mesoscale Experiments

- Mesoscale experiment were iterated to eliminate large sources of uncertainty in the particle bed and diagnostics
- Hollow glass microspheres lower the overall volume fraction
- P-rad shows a linear behavior in particle front from 22 time series of "x-ray" images


Macroscale Experiments

- Test design significantly considers simulation capability:
 - Charge geometry and material match legacy data
 - Casing effect was further investigated with small perturbations to casing
 - Particles selected for sphericity and monodispersity
- High-speed video shows the casing fragmented into approximately 44 strips for ALL steel liner tests, far more than the number of notches present
- Shock time of arrival, peak pressure, and particle front shows no notching affect

CCMT | 23

CCMT

Do you have any questions?

UF UNIVERSITY OF FLORIDA  **NNSA**

CCMT

CMT-nek Development and Application

Center for Compressible Multiphase Turbulence
Tri-Lab Services Team Review













December 12, 2018

Jason Hackl
David Zwick



UF UNIVERSITY OF FLORIDA

CMT-nek users and developers

 Prof. Paul Fischer, UIUC	 Jason Hackl	 David Zwick	 Nguyen Tri Nguyen	Method & code developers
 Tania Bannerjee	 Keke Zhai	 Aravind Neelakantan	 Sai Chenna	
 Brad Durant	 Goran Marjanovic	 Rahul Koneru	 Fred Ouellet	Student users

2

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Accomplishments in Year 5

- Positivity-preserving solution limiting
 - Scale conserved variables to catch undershoots
- Particle 4-way coupling
- Compressible multiphase flows with shocks
- GPU and load-balanced ports
- Transition to CMT-nek
 - Microscale: shock-sphere interactions
 - Mesoscale: ASU expansion
 - Macroscale: **CMT-Hero 3**: blast with particles

CCMT 3

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Outline

1. CMT-nek gas dynamics
 - a. Solution limiting strategy
 - b. V&V of shock-capturing and microscale flow
2. CMT-nek multiphase
 - a. Lagrangian particles and scaling
3. Code development and future work
 - a. Integration
 - b. Future simulations

CCMT 4

Discontinuous Galerkin Spectral Elements

$$\frac{\partial U_i}{\partial t} + \nabla \cdot \mathbf{H}_i(\mathbf{U}, \nabla \mathbf{U}) = R_i$$

flux = convective + diffusive

$$\mathbf{H}_i = \mathbf{H}_i^c(\mathbf{U}) + \mathbf{H}_i^d(\mathbf{U}, \nabla \mathbf{U})$$

Gauss-Lobatto-Legendre quadrature on tensor product of N points

- Mass matrix \mathbf{B}

$$\left[\frac{\partial \mathbf{U}}{\partial t} \right] = \mathbf{B}^{-1} \left(I_{vol}^{(c)} - I_{sfc}^{(c)} \right)$$

Inviscid Euler equation fluxes \mathbf{H}^c

$$-\mathbf{B}^{-1} (I_{GU} - I_{G\tau U} - I_{KU} - \mathbf{R})$$

Artificial viscosity \mathbf{H}^d + source terms

$$\mathbf{H}_1^d = -\kappa_s \nabla \rho$$

$$H_{i+1,j}^d = - \left(\rho \nu_s \sigma_{ij} + \kappa_s u_j \frac{\partial \rho}{\partial x_i} \right)$$

$$\mathbf{H}_5^d = - \left[\rho \nu_s \mathbf{u} : \underline{\sigma} + \kappa_s \left(\nabla(\rho e) + \frac{1}{2} |\mathbf{u}|^2 \nabla \rho \right) \right]$$

Guermont & Popov (2014) *J. Appl. Math.* **74**:284-305

U+ = neighbor's EU

U- = element's EU

U+ = neighbor's EU

Right-hand-side for fully explicit TVDRK3

CCMT

Numerical flux $\mathbf{H}^*(\mathbf{U}^-, \mathbf{U}^+)$

- restriction \mathbf{E}
- nek5000 gather-scatter operation $\mathbf{Q}\mathbf{Q}^T$

Hackl, Shringarpure, Koneru, Delchini & Balachandar (2018) in revision for *Computers & Fluids*

Shocks via Artificial Viscosity

Entropy viscosity method¹

$$\nu_* = c_E h^2 \frac{|R_s|}{\|s - \langle s \rangle\|_\infty}$$

$$\nu_{\max} \equiv c_{\max} h \max_{\Omega_e} (|\mathbf{u}| + c)$$

$$\nu_s = S(\min(\nu_*, \nu_{\max}))$$

$$\kappa_s = \mathcal{P} \nu_s$$

$$R_s \equiv \partial s / \partial t + \nabla \cdot (\mathbf{u} s)$$

- Successes
 - Ideal-gas validation
 - First-order convergence in h, N
 - Microscale and mesoscale
- Limitations
 - Three tuneable parameters
 - Saturates in strong blasts (viscosity is not localized)
 - Entropy residual R_s is expensive to compute for surrogate and other non-ideal EOS

Macroscale runs defaulted to first-order wave-speed viscosity. We sought alternatives...

¹Guermont, Pasquetti & Popov (2011) *J. Comp. Phys.* **230**:4248-4267

Page 69 of 164

Internal energy limiter

¹Zhang and Shu (2010) *JCP* **229**:8918-8934

²Lv & Ihme (2015) *JCP* **295**:715-739

- Rescale conserved variables ($\mathbf{U} \rightarrow \mathcal{L} \mathbf{U}$) every RK stage.
 - to preserve positivity of density¹

$$\mathcal{L} \rho(r_i, s_j, t_k) = (\rho(r_i, s_j, t_k) - \bar{\rho}) \theta + \bar{\rho}$$

$$\theta = \min \left(\frac{\bar{\rho} - \epsilon}{\bar{\rho} - \rho_{\min} + \epsilon}, 1 \right)$$

$$\rho_{\min} = \min_{i,j,k \in [1,N]} (\rho(r_i, s_j, t_k))$$
 - and internal energy
 - $\mathcal{L} \mathbf{U} = \mathbf{U}(r_i, s_j, t_k) + \varepsilon (\bar{\mathbf{U}} - \mathbf{U}(r_i, s_j, t_k))$
 - ε determined by negativity of a convex function, like internal energy, and Jensen's inequality².
$$\varepsilon = \frac{\tau}{\tau - \rho e(\bar{\mathbf{U}})}, \quad \tau = \min \left(0, \min_{i,j,k \in [1,N]} \rho e(r_i, s_j, t_k) \right)$$

$$\bar{\rho} = \int_{\Omega_e} \rho dV \approx \sum_{i,j,k=1}^N \omega_i \omega_j \omega_k \rho(r_i, s_j, t_k)$$

| 7

Shock-sphere drag coefficient

- Mach 3 shock-sphere interaction, $N = 5$

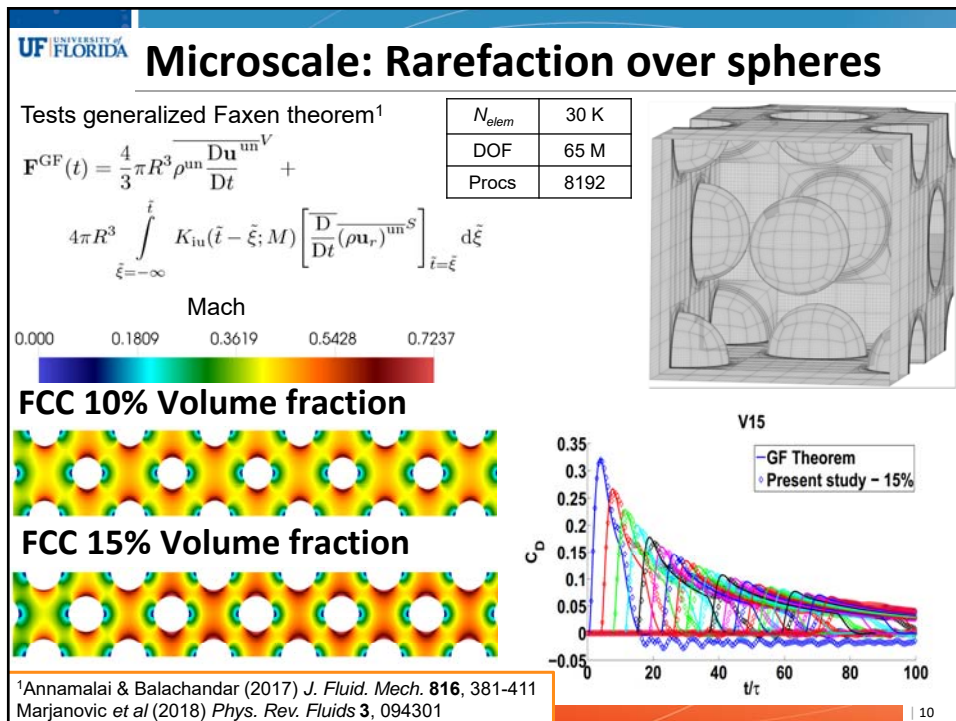
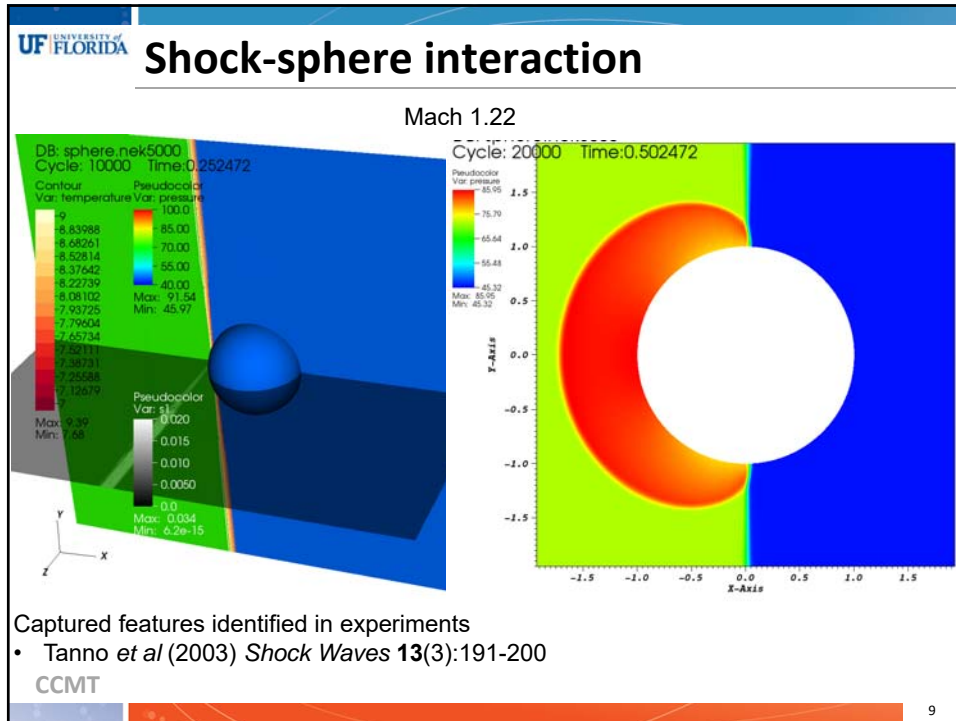
	Mach 1.22	Mach 3
c_{\max}	0.3	0.75
c_E	40	85
P	0.75	0.75
- Mach 1.22 shock-sphere, two resolutions

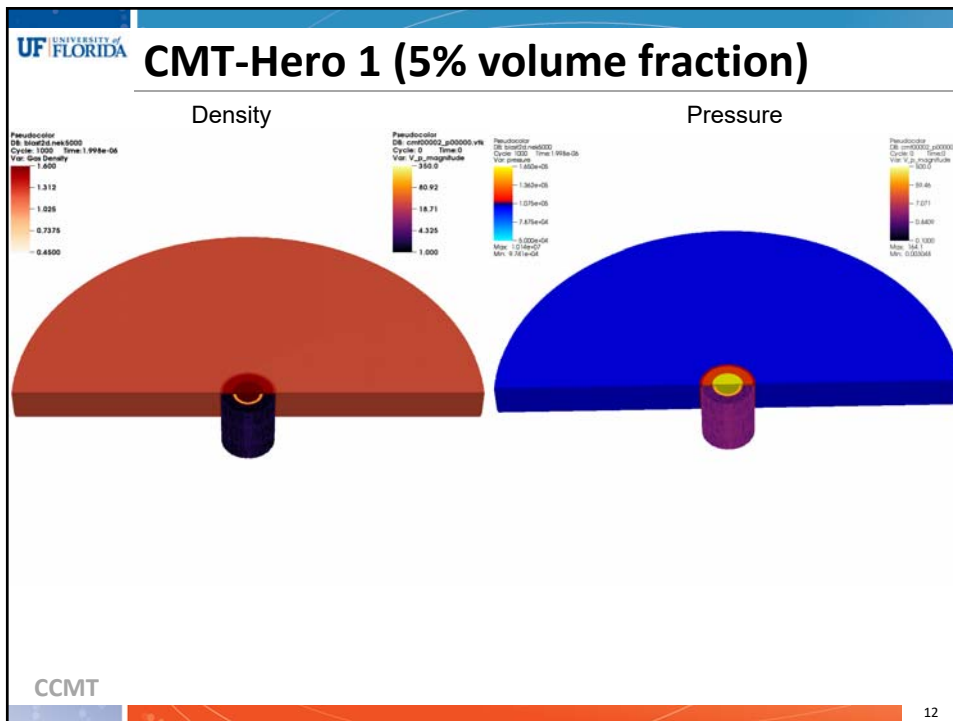
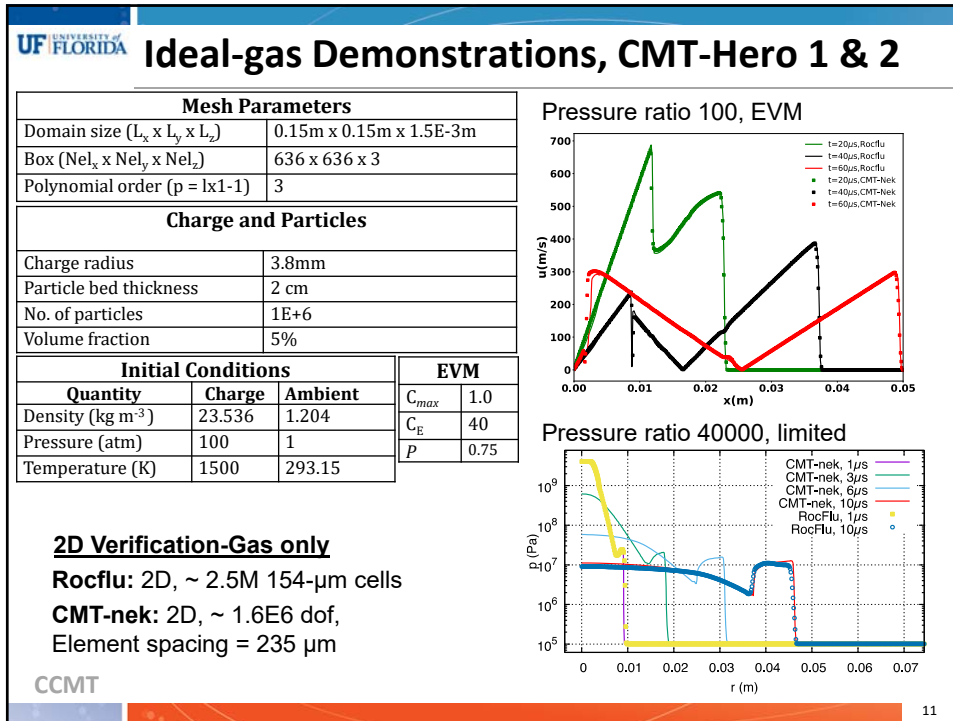
Coarse: 2048 elements, $N = 5$
Fine: 1 M elements, $N = 5$

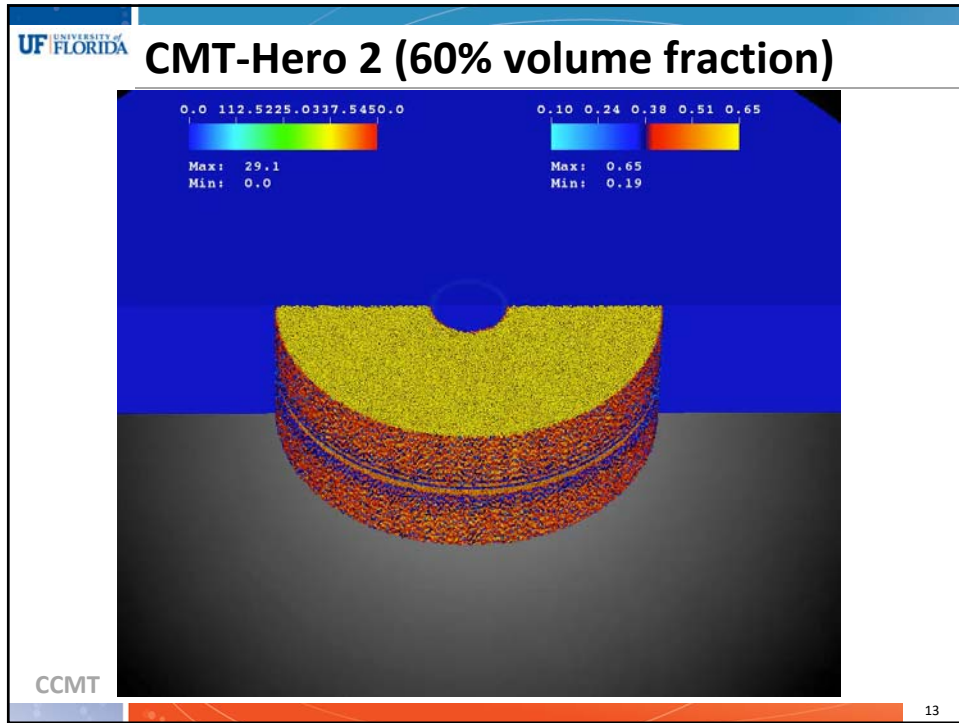
CCMT

tV_s/D | Mehta, Jackson, Zhang & Balachandar (2016) *J. Appl. Phys.* **119**:104901

8







Multiphase Capabilities

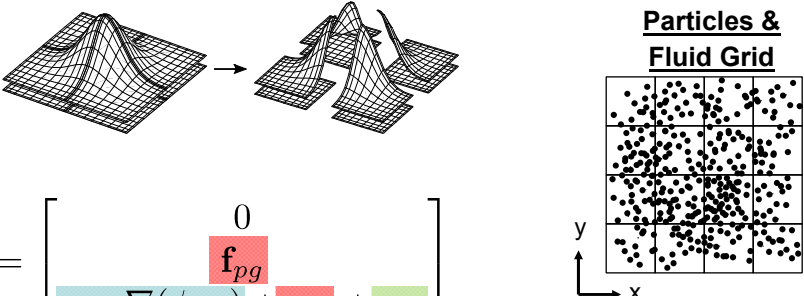
✓ Existed
 ✓ New

		nek5000	CMT-nek
↑ Single-phase	Fluid-Fluid	Incompressible	✓
		Low-Mach-number	✓
		Shock waves	✓
↓ Increased multiphase coupling	Tracer Particles	✓	✓
	Particle-Fluid	Dilute	✓
		Dense	✓
Particle-Particle	✓	✓	

Multiphase exascale problems rely on efficient communication and computations to maintain both accuracy and scalability

UF UNIVERSITY OF FLORIDA **Multiphase Source Terms**

Projection: Particles Effect Surrounding Grid



$$\mathbf{R} = \begin{bmatrix} 0 \\ \mathbf{f}_{pg} \\ \sigma_g : \nabla(\phi_p \mathbf{v}) + g_{pg} + q_{pg} \end{bmatrix}$$

Particles & Fluid Grid

Particle-fluid force coupling and energy contribution

Particle-fluid energy coupling Eulerian particle velocity

CCMT 15

UF UNIVERSITY OF FLORIDA **Governing Equations**

For a single particle:

Position:

$$\frac{d\mathbf{X}}{dt} = \mathbf{V}$$

Momentum:

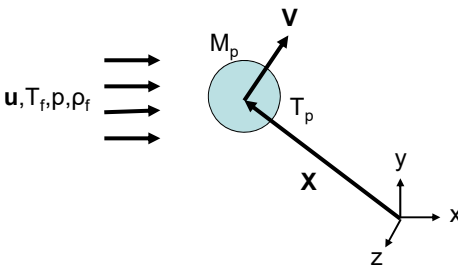
$$M_p \frac{d\mathbf{V}}{dt} = \mathbf{F}_{pg} + \mathbf{F}_{un} + \mathbf{F}_c + \mathbf{F}_b$$

Energy:

$$M_p C_{p,p} \frac{dT_p}{dt} = Q_h$$

Hydrodynamic force Collisional force

Body force Hydrodynamic heat transfer



CCMT 16

UF UNIVERSITY OF FLORIDA **Inter-Particle Collision Model**

Normal spring-damper soft-sphere collision model:

$$\mathbf{F}_{c,i} = \sum_{j=1}^n \mathbf{F}_{c,ij}^S + \mathbf{F}_{c,ij}^D$$

Spring force:

- Overlapping particles repel each other

Damping force:

- Colliding particles dissipate energy

From thousands of particles...
... to millions of particles...
... and even billions...

CCMT 17

UF UNIVERSITY OF FLORIDA **Implementation: Particle Storage**

- Data layout effects algorithm implementations
- Where particle data can be stored in memory:
 - Particle based** - evenly distributed in memory
 - Element based** - with local grid
 - Bin based** - with local particles
 - Other possibilities (Load-balancing – Keke)

Rank 1	Rank 2	Rank 3	Rank 4

CCMT 18

Element Based Storage

- Each particle is assigned to the same rank as its surrounding grid
- Pros:
 - Many operations are local on each rank
- Cons:
 - Possible load/memory imbalance, grid size restrictions

E1 ① ② ③ ④ ⑤	E2 ⑥ ⑦
E3	E4 ⑧

Rank 1	Rank 2	Rank 3	Rank 4

CCMT 19

Bin Based Storage

- Each particle is stored with nearby particles based on some type of particle bin structure
- Pros:
 - Good load/memory balance, particles know neighbors
- Cons:
 - Communication overhead for projection and collisions

E1 ① ② ③ ④ ⑤	E2 ⑥ ⑦	
E3	E4 ⑧	

Rank 1	Rank 2	Rank 3	Rank 4

CCMT 20

UF FLORIDA

Scaling

How does bin based algorithm scale?

Setup:

- Uniform flow in a 3D box
- Particles are randomly distributed in box
- Cases run on Vulcan (LLNL)
- Bin based is better than element based storage
- Scaling:
 - Parameter scaling (N_p, N, δ_f)
 - Processor scaling (strong and weak scaling)

CCMT 21

UF FLORIDA

Bin Based Processor Scaling

Strong Scaling

ASU Simulation Weak Scaling

Description:

- Problem fixed while R processors varied
- 25 million total particles
- N_p/R decreases from 24,500 to 256

Observations:

- Collision algorithm scales as $O((N_p/R)^2)$
- Creating ghost particles scales as $O(N_p/R)$

Description:

- Problem per rank fixed while R processors varied
- 3,072 particles per rank
- N_p increases from 3 to 302 million

Observations:

- Sending ghost particles scales as $O(\log_2 R)$

CCMT 22

UF FLORIDA Fluidization

- Particle bed initially at $\sim 1/4$ of height
- Particles initially randomly distributed in entire domain and allowed to settle

Mesh and Boundary Conditions

Parameter	Value	Run	Inlet Superficial Velocity [m/s]
N_p	9,240	1	0.083
D_p [m]	1,200E-6	2	0.165
ρ_p [kg/m ³]	1,000	3	0.248
L_x [m]	0.044	4	0.331
L_y [m]	0.120	5	0.414
L_z [m]	0.010	6	0.496
ρ_f [kg/m ³]	1.205		
ν_f [Pa s]	1.8E-5		

Experimental minimum fluidization velocity ~ 0.3

$L_e = 0.003$ [m]
 $N = 5$

CCMT 23


UF FLORIDA Pressure Drop

- Superficial velocity is intrinsic fluid velocity multiplied by fluid volume fraction at inlet
- Average pressure drop across bed is normalized by bed weight

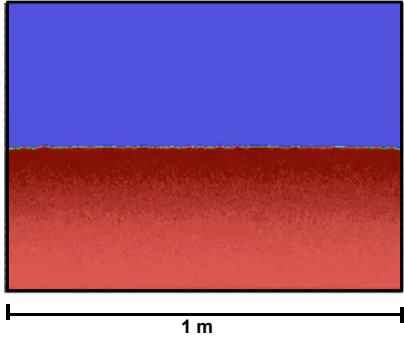
Average pressure drop vs. inlet velocity

Particle volume fraction (Run 6)

CCMT 24



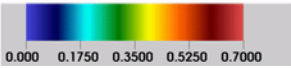
Large Scale Fluidized Bed



1 m

Parameter	Value
N_p	2.6 million
D_p [m]	1,200E-6
ρ_p [kg/m ³]	1,000
L_x [m]	1
L_y [m]	0.5
L_z [m]	0.010
N_e	180,000
N_{grid}	22.5 million

Particle Volume Fraction




0.000 0.1750 0.3500 0.5250 0.7000

- Inlet conditions of Run 6
- 65,536 MPI ranks
- 70 hours
- 10 s simulation time (played in real-time)

CCMT

25



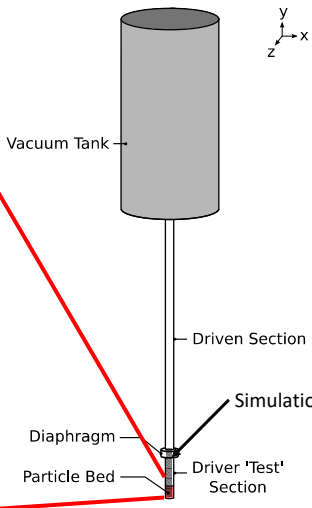
ASU Experiment

Example Experiment (ASU)
Multiphase shock tube is a surrogate used for laboratory-scale volcanic eruptions

Rarefaction head is initially directly in front of particle bed front

- Problem with glass
- Particle decomposition

Experimental Setup



Vacuum Tank

Driven Section

Simulation Domain

Diaphragm

Driver 'Test' Section

Particle Bed

lock tube section to rapid pressure bursts

CCMT

26

2D Simulation Setup

- 2D simulations
 - Effect of numerical model parameters
- Campaigns:
 - Mesh refinement
 - Projection filter width
 - Collision parameters
 - Domain width
 - Aleatory uncertainty

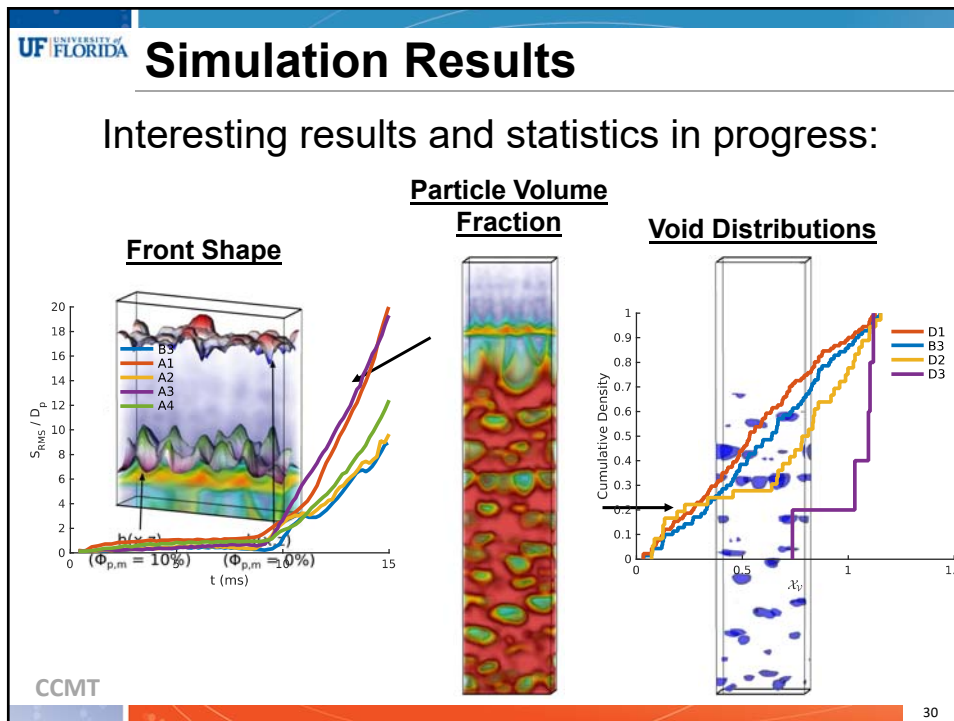
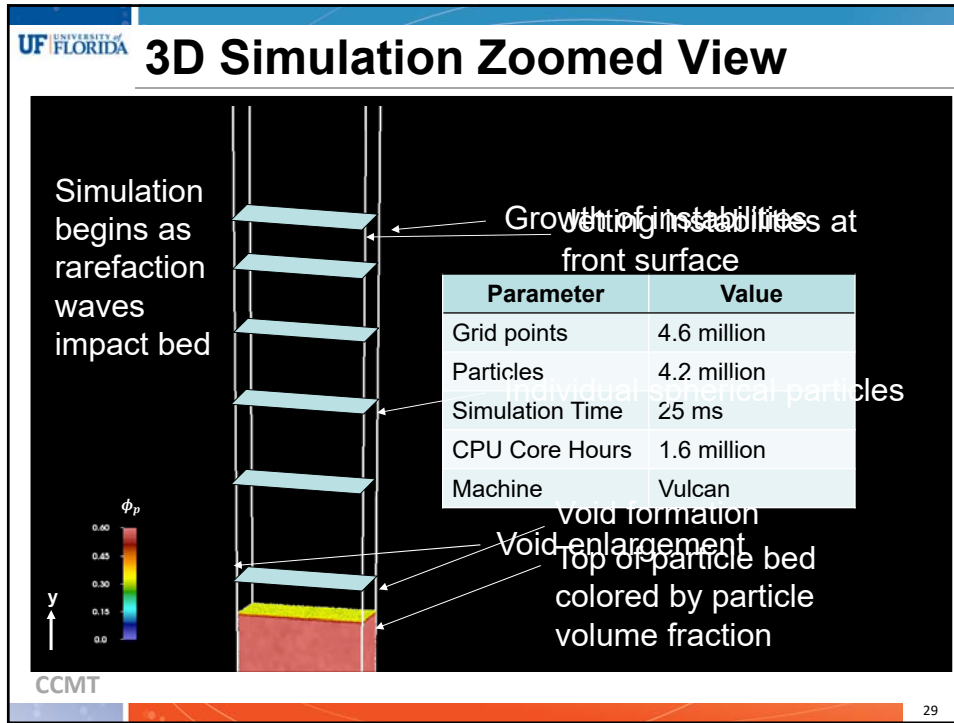
CCMT 27

Effect of Pressure Drop

Front rise velocity:

- 4 pressures and 3 levels of mesh refinement
- Leading particle is the bed height
- Linear regression to get slope
- Richardson extrapolation to estimate solution

CCMT 28



Transition and code management

- CMT-nek workshop November 29, 2017
 - Training and introduction for CS, macroscale
- Host nek5000 User/Developer Meeting
 - Tampa, Florida April 17-18, 2018
- CMT-nek developer documentation
 - Doxygen driven
- Automated regression testing via travis.ci

CCMT 31

Demonstration problem roadmap

	Hero 1 ✓	Hero 2 ✓	Hero 3 ✓	CMT-Hero 1 ✓	CMT-Hero 2 ...	CMT-Hero 3	CMT-Hero 4
Case	Frost	Frost	Frost	Frost	Frost	Blastpad	Blastpad
Simulation Time	2.5ms	2.5ms	2.5ms	0.3 ms	0.3 ms	2.5 ms	2.5 ms
Degrees of Freedom	30M	30M	60M	32.4 M	32.4 M	120 M	240 M
MPI ranks	512	4096	5760	32 768	32 768	65 536	131 072
No. computational particles	5M	5M	15M	1 M	2 M	5 M	10 M
Initial Particle volume fraction	5%	40% - Frozen	10% - Frozen	5 %	60 %	60 %	60 %
Gas-Particle Coupling	2-way	2-way	2-way	2-way	4-way	4-way	4-way
Particle bed perturbation	None	None	1 mode in azimuthal direction	none	none	none	none
Reactive burn	No	No	Yes	No	No	No	Yes
Equation of state	Ideal Gas	JWL	JWL	Ideal Gas	Ideal Gas	JWL	JWL
	Year 1 - 4				Year 5-6		

CCMT 32

Transition plan: timeline

	Y6-Q1	Y6-Q2	Y6-Q3	Y6-Q4
CMT-Hero 3				
Outflow BC				
Load balancing				
Localized artificial viscosity				
CMT-Hero 4				
Reactive burn				
Smart start				
Entropy-stable fluxes				

DGSEM is undergoing rapid change as its usage spreads!

- Better & cheaper ways to stabilize high-order methods

CCMT 33

Convective terms: past & future

Nonlinear flux functions \mathbf{H}^c

$$U_j u_i + \phi_g p \delta_{ij}$$

\mathbb{P}^{N-1}
 $\mathbf{v}^T \mathbf{D}_k^T \mathbf{B}[\mathbf{r}_{k, \mathbf{x}_i}] [U_j] [u_i]$
 \mathbb{P}^{N-1}

- Integrand has products of polynomial & **rational functions** (and thermodynamic properties)
- Most stability proofs for DG depend on exact integration, but quadrature on rational functions is not exact.

But it turns out that \mathcal{D} has the **summation-by-parts (SBP)** property

$$(\mathbf{B}\mathcal{D})^T = \text{diag}(-1, 0, \dots, 0, 1) - \mathbf{B}\mathcal{D}$$

Which means it discretely guarantees the weighted-residual DG statement for a wider range of functions than the quadrature weights in \mathbf{B} can integrate exactly

$$\int_{\Omega_e} v \nabla \cdot \mathbf{H} dV = \int_{\partial\Omega_e} v \mathbf{H}^* \cdot \mathbf{n} dA - \int_{\Omega_e} (\nabla v) \cdot \mathbf{H} dV$$

Provided \mathbf{H}^c is written in a very special way...

Gassner, Winters & Kopriva (2016) "Split form nodal DG schemes with SBP for Euler equations" *JCP* **327**:39-66
 Carpenter et al (2014) *SIAM J. Sci. Comput.* **36**(5):B835-B867

CCMT 34

UF UNIVERSITY OF FLORIDA **Convective terms: past & future**

$$\sum_{m=1}^N \mathcal{D}_{im} H(U(r_m)) = \frac{1}{\omega_i} (\bar{H}(U(r_{m+1})) - \bar{H}(U(r_m))) \approx 2 \sum_{m=1}^N \mathcal{D}_{im} H^\#(U(r_i), U(r_m))$$

To get this to behave stably \rightarrow i.e., almost like a first-order subcell (but better), \rightarrow Use entropy-stable fluxes here in $H^\#$

So our volume integral $I_{vol}^{(c)}$ has grown a **Riemann solver!**

- Still consistent, conservative, yet formally high-order
- May be applied to tensor products in 3D
- It's even good to incorporate $H^\#(U^-, U^+)$ into numerical surface flux H^*
- Leverage work in energy stability and entropy stability in **finite volume** schemes
e.g. Ismail & Roe (2009) *J. Comp. Phys.* **228**:5410-5436

$$\mathbf{H}_2^{c\#} = \begin{bmatrix} \hat{\rho} \hat{u}^2 + \hat{p} \\ \hat{\rho} \hat{u} \hat{v} \\ \hat{\rho} \hat{u} \hat{w} \end{bmatrix} \quad \mathbf{z} \equiv \sqrt{\frac{\hat{\rho}}{p}} \begin{bmatrix} 1 \\ \mathbf{v} \\ p \end{bmatrix} \quad \hat{\rho}(U^-, U^+) \equiv \left\{ \{z_1\} \right\} \frac{z_5^- - z_5^+}{\ln(z_5^-) - \ln(z_5^+)}$$

Fisher & Carpenter (2013) "High-order entropy stable finite difference schemes..." *J. Comp. Phys.* **252**:518-557
Winters et al (2018) "A comparative study...for under-resolved turbulence computations" *JCP* **372**:1-21

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UF UNIVERSITY OF FLORIDA **Inviscid vortex: "dealiased" vs split**

$$I_{vol}^{(c)} = \mathcal{M}^{-1} \mathcal{I}^\top \mathcal{D}_{i,M}^\top \mathcal{M}_M \mathbf{h}_{j,M} \quad I_{vol}^{(c)} \approx 2 \sum_{m=1}^N \mathcal{D}_{im} H^\#(U(r_i), U(r_m))$$

Cycle: 700000 Time: 699.999

Dealiased fluxes blow up

Steady CCW vortex, periodic box, $u_\infty = 1$

Split-form lasts longer with energy-preserving fluxes (need entropy-stable)

No artificial viscosity
Coarse! $\rho=24$
 $h=10/4$

H[#] Kennedy & Gruber (2008) J. Comp. Phys. 227:1676-1700

$$F_{KC}^\#(U_{ijk}, U_{mjk}) = \begin{bmatrix} \{\rho\} \{\mathbf{u}\} \\ \{\rho\} \{\mathbf{u}\}^2 + \{\mathbf{p}\} \\ \{\rho\} \{\mathbf{u}\} \{\mathbf{v}\} \\ \{\rho\} \{\mathbf{u}\} \{\mathbf{w}\} \\ \{\rho\} \{\mathbf{u}\} \{\mathbf{e}\} + \{\mathbf{p}\} \{\mathbf{u}\} \end{bmatrix}, \quad G_{KC}^\#(U_{ijk}, U_{imk}) = \begin{bmatrix} \{\rho\} \{\mathbf{v}\} \\ \{\rho\} \{\mathbf{u}\} \{\mathbf{v}\} \\ \{\rho\} \{\mathbf{v}\}^2 + \{\mathbf{p}\} \\ \{\rho\} \{\mathbf{v}\} \{\mathbf{w}\} \\ \{\rho\} \{\mathbf{v}\} \{\mathbf{e}\} + \{\mathbf{p}\} \{\mathbf{v}\} \end{bmatrix}$$

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Summary and Future Work

- CMT-nek is filling RocFlu's shoes and beyond
 - Compressible multiphase flow, scalable
 - Production simulation of microscale, mesoscale and macroscale flows
 - Behavioral emulation companion suite
- Future work (new capability)
 - Nonreflecting boundary conditions
 - Reactive burn initial conditions
 - Localizing artificial viscosity
 - Shock detectors
 - Improved numerical fluxes for discontinuous AV (Weighted interior penalty, Zunino (2009) *SIAM J. Sci. Comput.* **38**:99-126)
 - Entropy stability (Carpenter *et al* (2014) *SIAM J. Sci. Comput.***36**)

CCMT ➢ Entropy-stable fluxes for multiphase systems, non-ideal EOS

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Effect of AV on shock speed

- 1-equation mixture model
- Limited (curves)
- First-order wave-speed viscosity, diffusivity (symbols)

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***Do you have any
questions?***




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Uncertainty Budget


Validation and Uncertainty Reduction

Chanyoung Park, Raphael (Rafi) T. Haftka and Nam-Ho Kim
Department of Mechanical & Aerospace Engineering




UF UNIVERSITY OF FLORIDA


Uncertainty Budget Team




Raphael Haftka, PhD




Nam-Ho Kim, PhD




Sangjune Bae




Samaun Nili



Chanyoung Park, PhD



Kyle Hughes, PhD

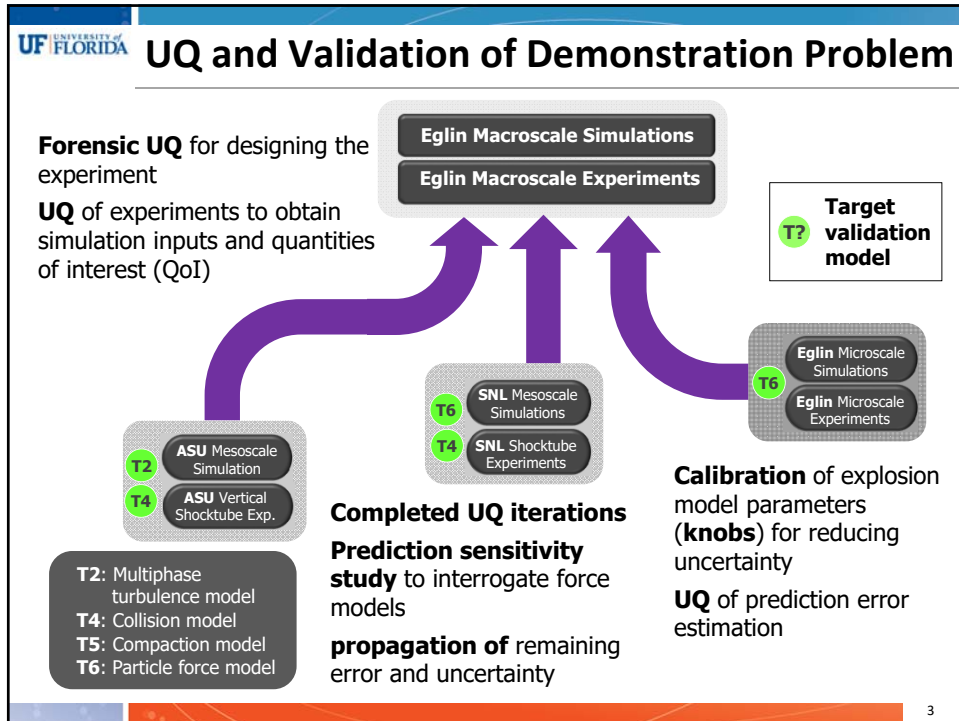


Giselle Fernandez

- PIs: Raphael Haftka and Nam-Ho Kim
- Research Scientist: Chanyoung Park (Joining **Caterpillar** at Champaign)
- Students
 - Samaun Nili (Mesoscale force model potential errors)
 - Sangjune Bae (Joining **CCMT as postdoc**)
 - Kyle Hughes (Graduated; **Postdoc** at **LANL**)
 - Giselle Fernandez (Graduated; **Postdoc** at **LANL** or **SNL**)

CCMT

2



- UF UNIVERSITY OF FLORIDA** **Outline**
- Eglin microscale
 - **Calibration** and **UQ** of reactive burn model parameters (knobs)
 - **UQ** of prediction error estimation of particle force model
 - UQ of Eglin macroscale
 - **Propagation** of error and uncertainty from Sandia meso- to Eglin macroscale based on sensitivity study
 - **Forensic UQ** based on past experiments
- CCMT
- 4

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Microscale UQ

UQ of simulation parameters and UR through calibration

Eglin Macroscale Simulations
Eglin Macroscale Experiments

Eglin Microscale Simulations
Eglin Microscale Experiments

Calibration of explosion model parameters (**knobs**) for reducing uncertainty
UQ of prediction error estimation

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Eglin Microscale Experiment

- Calibration of reactive burn model parameters
- UQ of JWL parameters
- Assessment of particle force model
- Particle position (PP) is quantity of interest (QoI)

NOT TO SCALE

Xenon flash lamp
flash bulb
Pressure probes
witness panel
centerline / shot line
66°
D(θ)=34"
Explosive Casing
Phantom
Phantom
Near Field
Far Field
Overhead schematic of the test set-up.

Explosive Casing

RP-83 Detonator
Detonator Adapter
Steel Casing
PBXN-5
Tungsten Particle

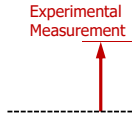
CCMT

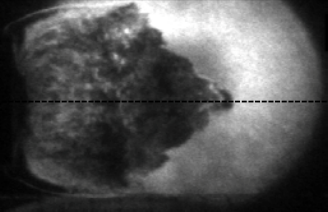
6

UF UNIVERSITY OF FLORIDA **Experimental Measurements**

- Transverse shock position (for calibration of reactive burn parameters)

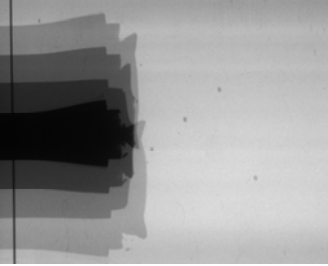
Experimental Measurement





Highspeed camera image

- Particle trajectory (for validation of particle force model)



X-ray image

CCMT

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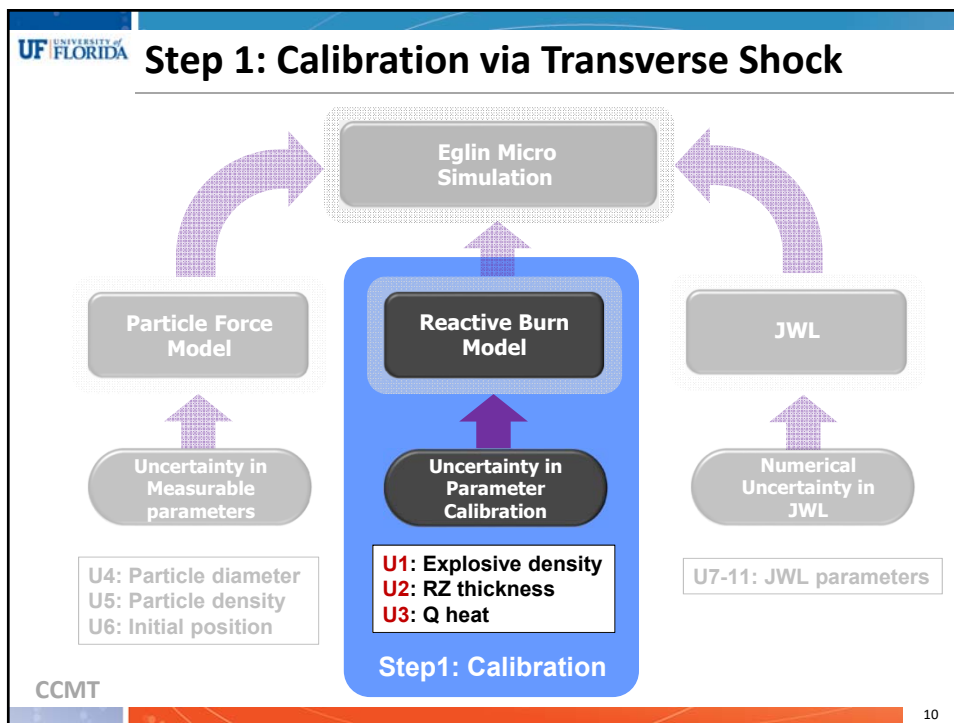
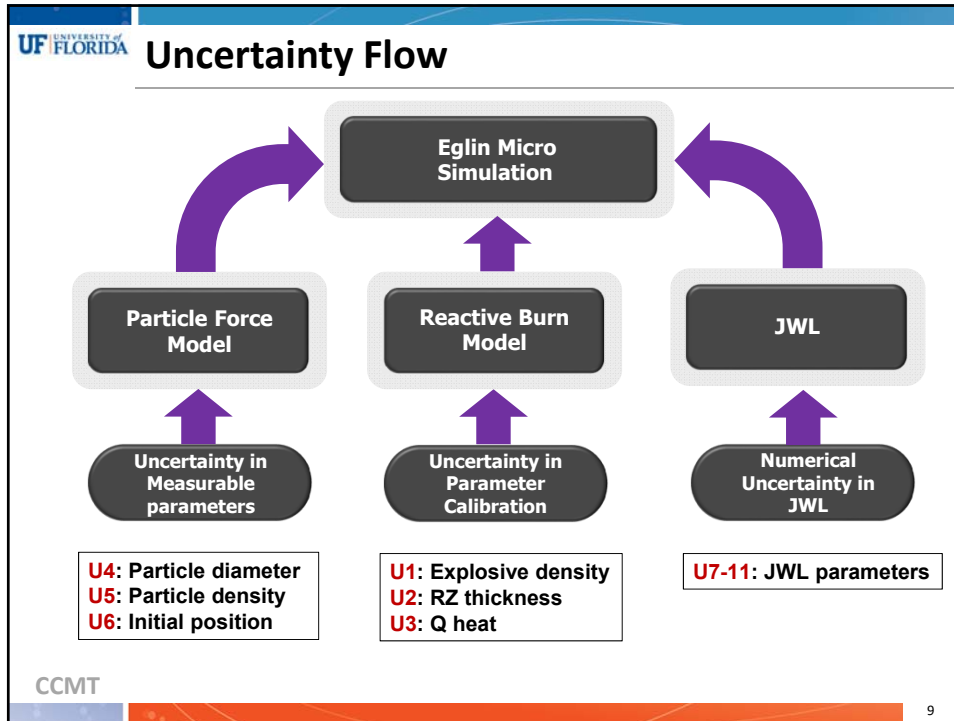
UF UNIVERSITY OF FLORIDA **Uncertain Parameters**

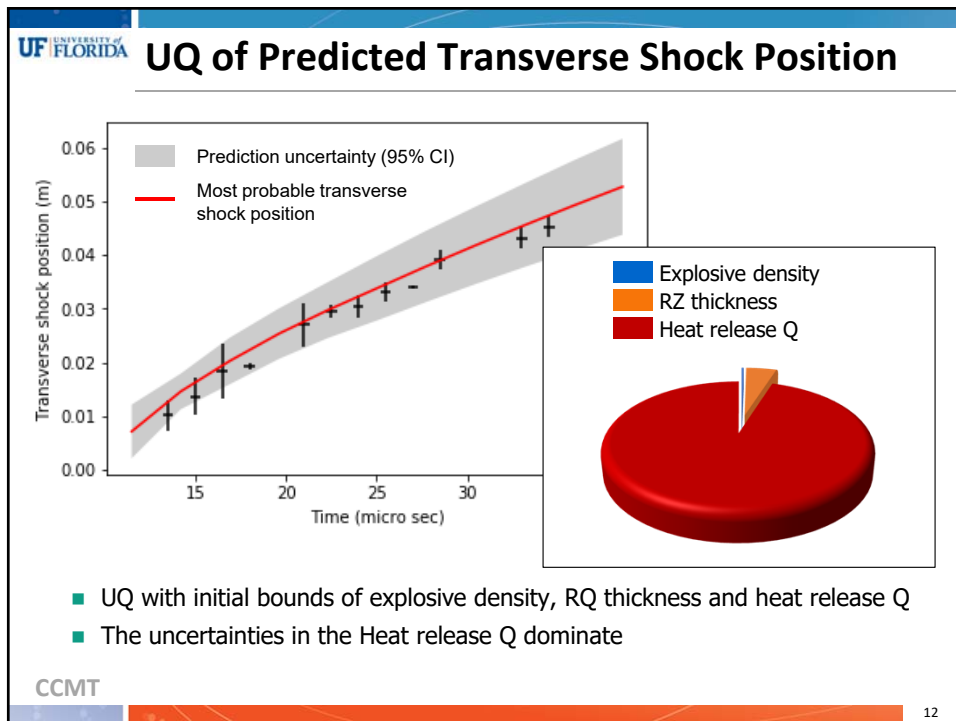
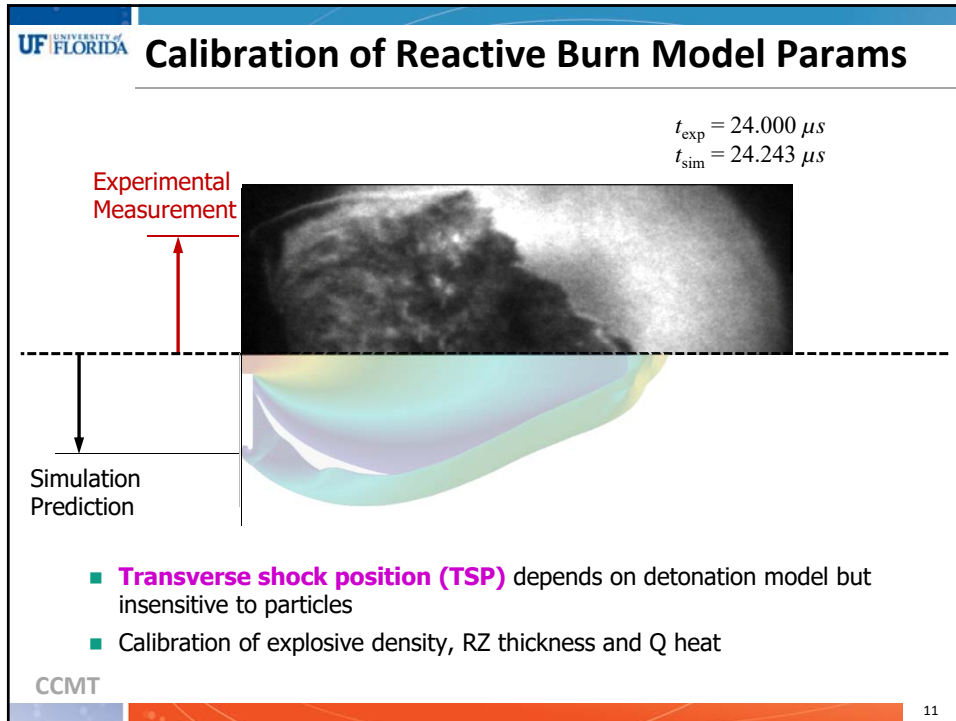
Parameter	Quantity	Method
U1: Explosive density	mean = 1795 kg/m ³ , std = 2.9 kg/m ³ , Bi-modal	Derived
U2: RZ thickness	[0.365, 1.4] mm	Expert opinion
U3: Heat release Q	[11.6, 14.2] MJ/kg	Literature and opinion
U4: Particle diameter	mean = 2.0156 mm, std = 0.0073 mm, Weibull	Direct measurement Micrometer, 52 samples
U5: Particle density	mean = 15540 kg/m ³ , std = 250 kg/m ³ , Normal	Pycnometer Gas Pycnometer, 12 samples
U6: Initial radial position	[0, 0.254] mm	
U7-11: JWL	A, B, w, R ₁ , R ₂	Literature and opinion
...

- Listed uncertainties including uncertainties in simulation parameters (knob)
- **Kyle** actively involved in measurements and uncertainty estimation (forensic UQ)

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UF UNIVERSITY OF FLORIDA **Calibration of Detonation Model Parameters**

- Nonlinear least square method
 - Quantifying uncertainty in calibration based on linearization
$$\arg \min_{\mathbf{u}} \sum_{i=1}^N |y_{sim}(t_i, \mathbf{u}) - y_{exp,i}|^2$$
- Naïve Bayesian calibration
 - Treat data independently
 - Provide uncertainties in parameter estimates from posterior distribution
$$y_{exp,i} = y_{sim}(t_i, \mathbf{u}) + \varepsilon_i \quad \varepsilon_i \sim N(0, \sigma_{n,i}^2)$$
- Bayesian calibration
 - Gaussian process based calibration with discrepancy function
 - Provide uncertainties in parameter estimates from posterior distribution
$$y_{exp,i} = y_{sim}(t_i, \mathbf{u}) + \delta(\mathbf{x}_i) + \varepsilon_i \quad \varepsilon_i \sim N(0, \sigma_n^2)$$

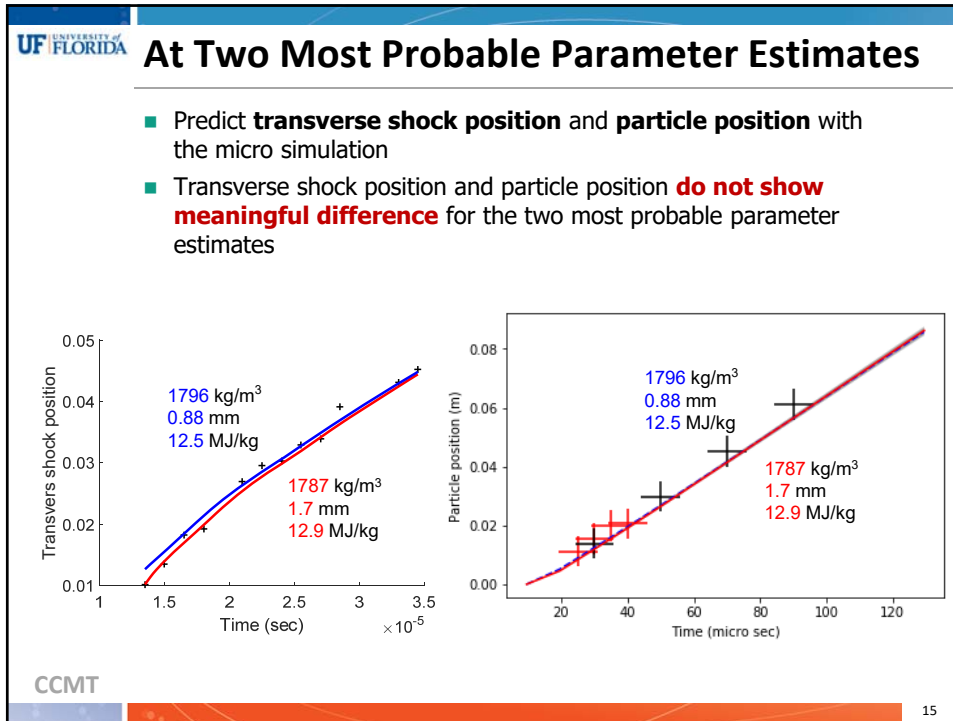
CCMT 13

UF UNIVERSITY OF FLORIDA **Calibrated Parameters**

- Three methods gave similar estimates for explosive density (**U1**) and Q heat (**U3**)
- Large variation in RZ thickness reflects its small effect on transverse shock position

Calibration Methods	Parameters	Most probable estimates
Nonlinear Least Square Method	Explosive density	1787 kg/m ³
	RZ thickness	1.7 mm
	Q heat	12.9 MJ/kg
Naïve Bayesian	Explosive density	1798 kg/m ³
	RZ thickness	1.7 mm
	Q heat	12.9 MJ/kg
Bayesian Calibration	Explosive density	1796 kg/m ³
	RZ thickness	0.88 mm
	Q heat	12.5 MJ/kg

CCMT 14

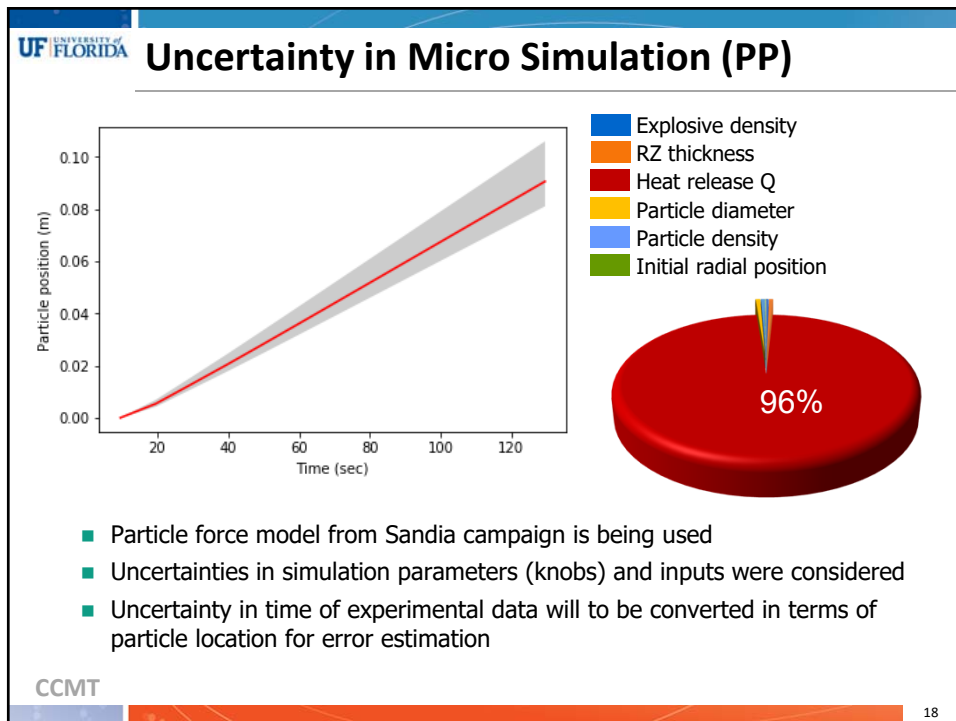
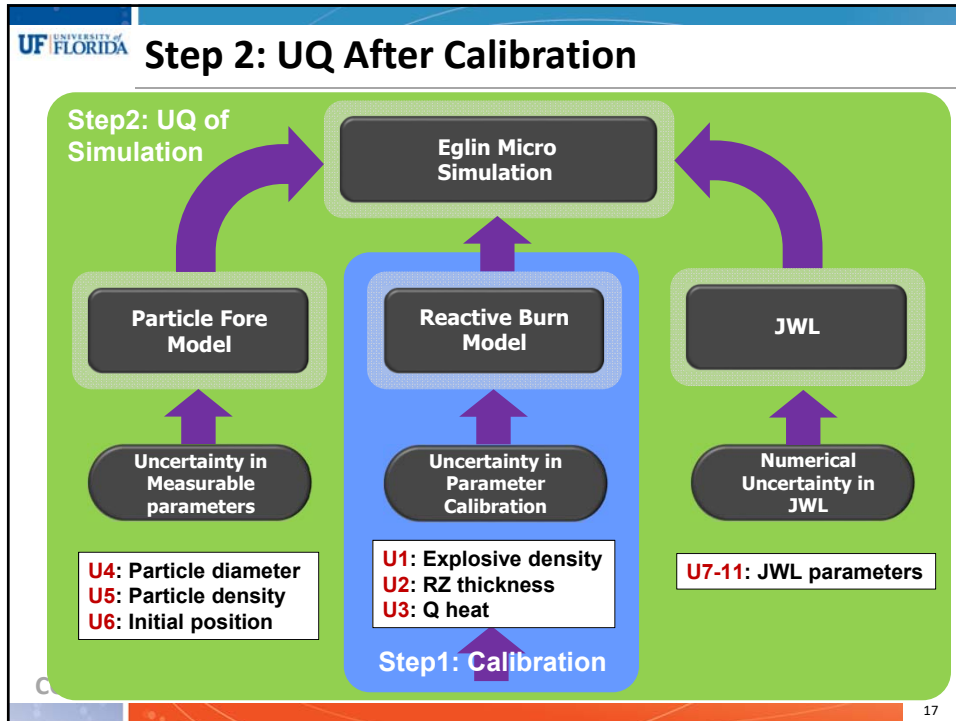


Highest Posterior Density Region

- Highest posterior density regions are different because of the methods interpret the data differently
- The marginal posterior distributions of RZ thickness shows that all the methods indicates that there are two highest posterior density regions for RQ thickness

Calibration Methods	Parameters	Most probable estimates	Highest posterior density region
Naïve Bayesian	Explosive density	1798 kg/m^3	[1787, 1799] kg/m^3
	RZ thickness	1.7 mm	[0.4, 1.1] [1.5, 1.7] mm
	Q heat	12.9 MJ/kg	[11.6, 13.8] MJ/kg
Bayesian Calibration	Explosive density	1796 kg/m^3	[1787, 1802] kg/m^3
	RZ thickness	0.88 mm	[0.4, 1.1] [1.5, 1.7] mm
	Q heat	12.5 MJ/kg	[11.6, 12.9] [13.8, 14.2] MJ/kg

CCMT 16



UF UNIVERSITY OF FLORIDA **Reducing Uncertainty in Casing**

Identified uncertainty in Casing

Redesign the experiment

Pictures of the microscale casing pre- and post- shot

- Reduced explosive and increased casing size
 - Pro: Casing would only deform slightly and can probably be included with some additional approximations
 - Con: Detonator remains a significant amount of the total explosive (multi-species modeling required)

CCMT 19

UF UNIVERSITY OF FLORIDA **Particle Position Comparison with UQ**

- Estimated error between prediction and processed experimental particle position data
- Most probable error (red line) is nearly unbiased
- Dominant uncertainty source changes in time

Sim

- Explosive density
- RZ thickness
- Heat release Q
- Particle diameter
- Particle density
- Initial radial position

Exp

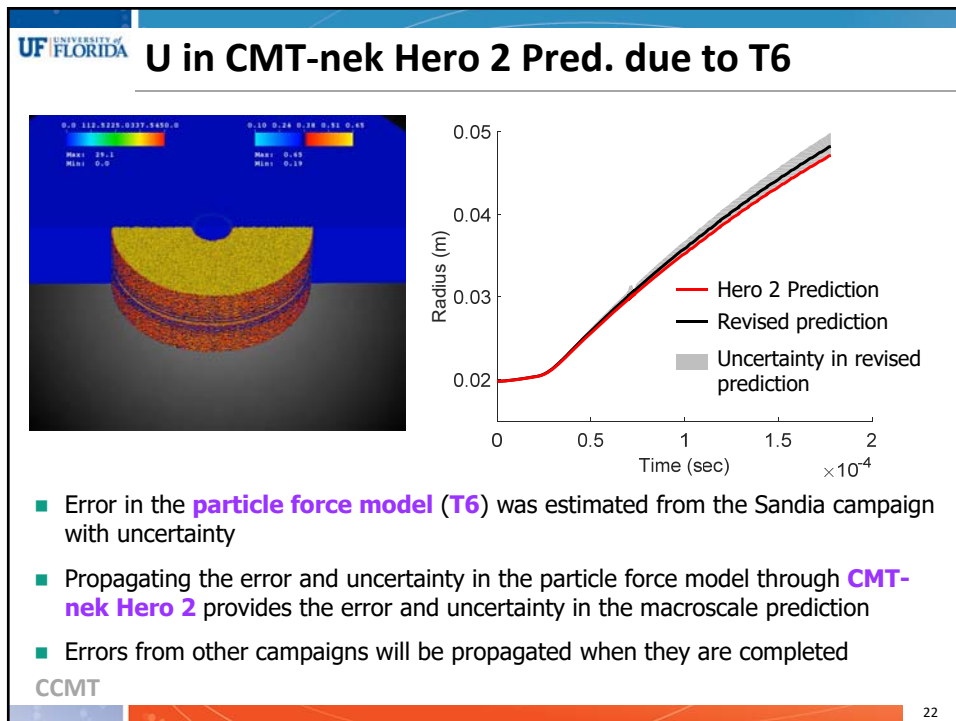
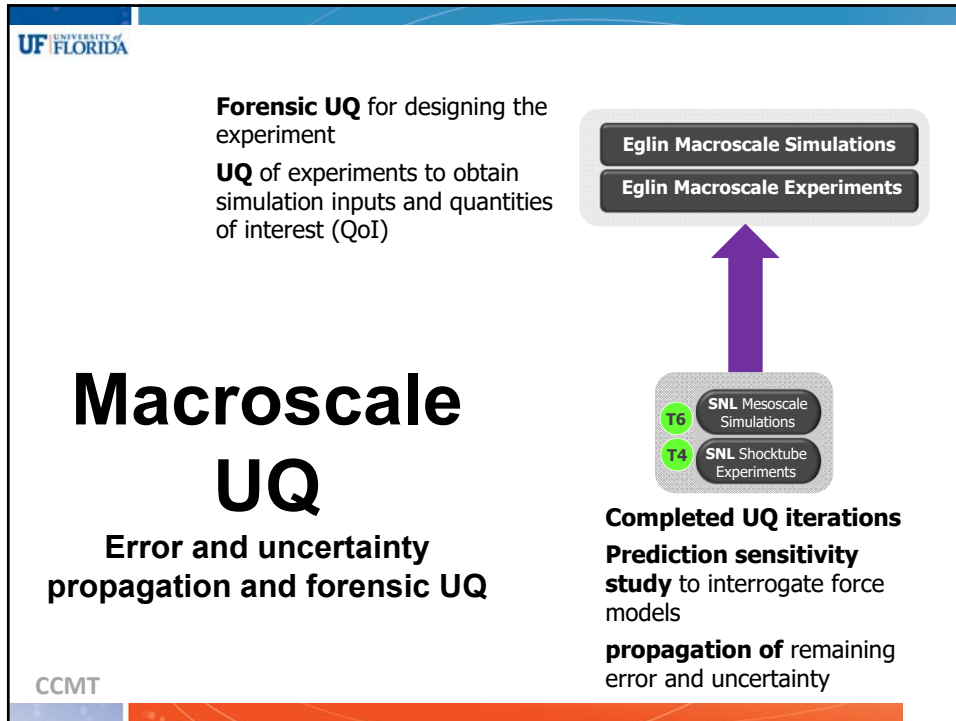
- Regression U
- Data noise
- U in time
- U in position

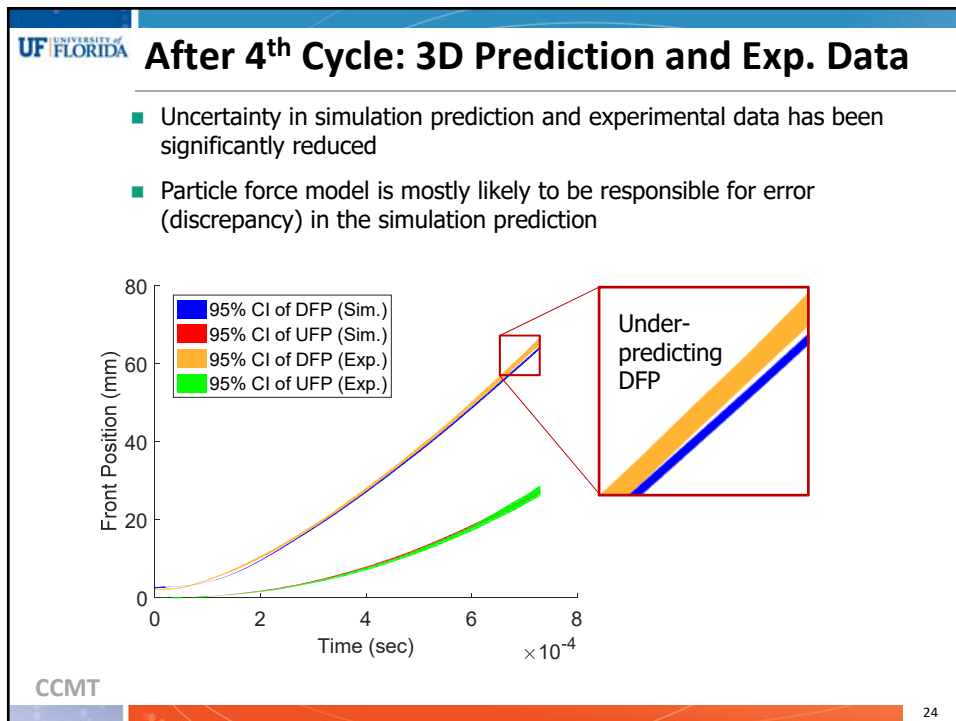
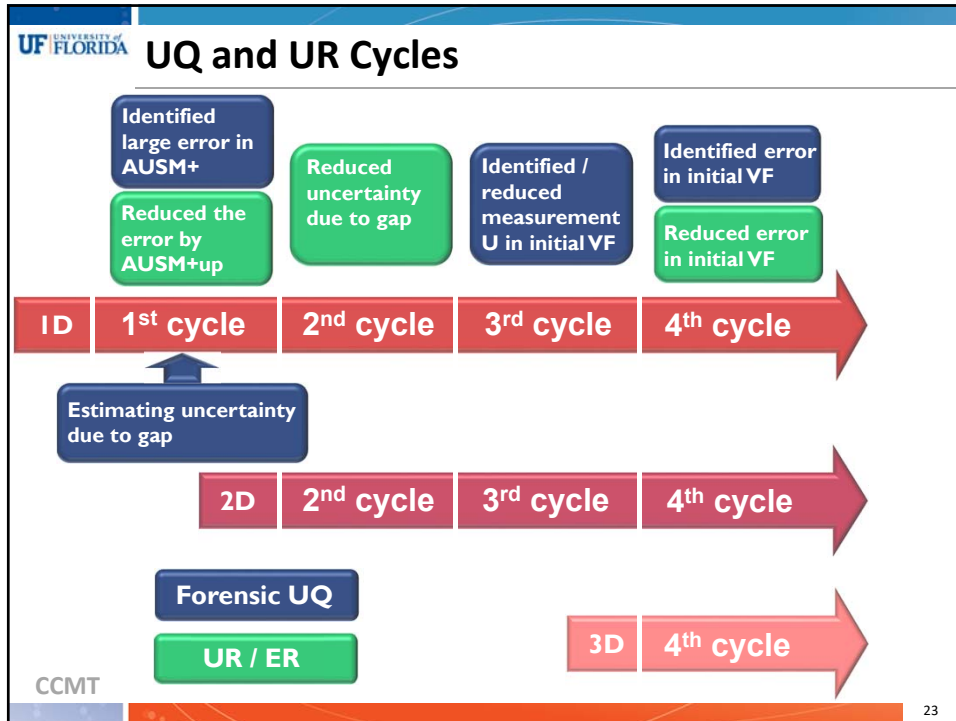
Error estimate (Exp-Sim) (m) vs Time (micro sec)

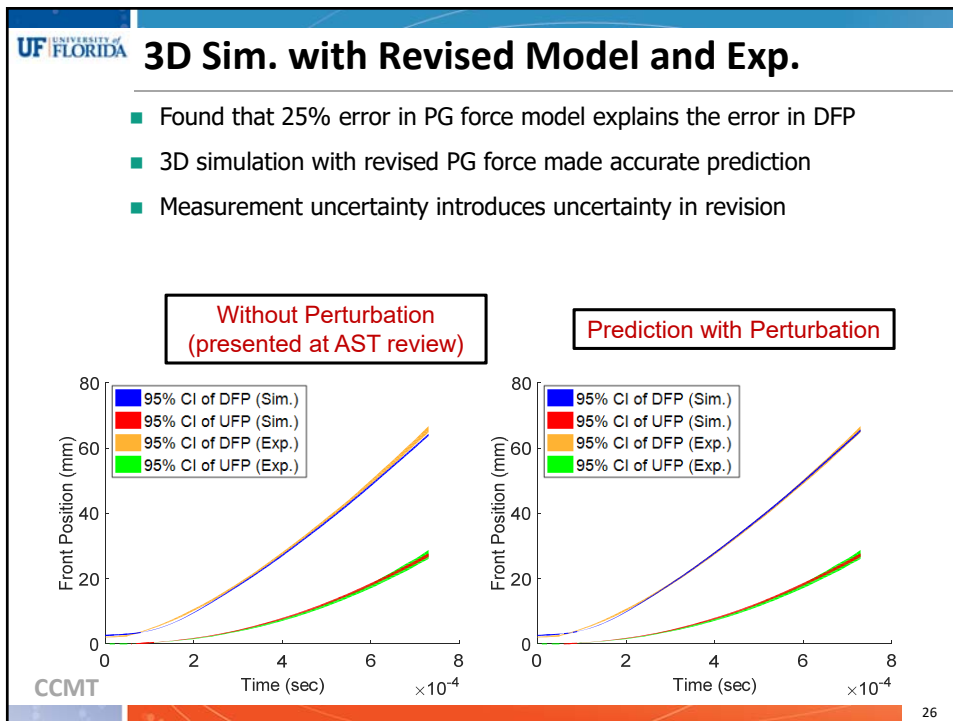
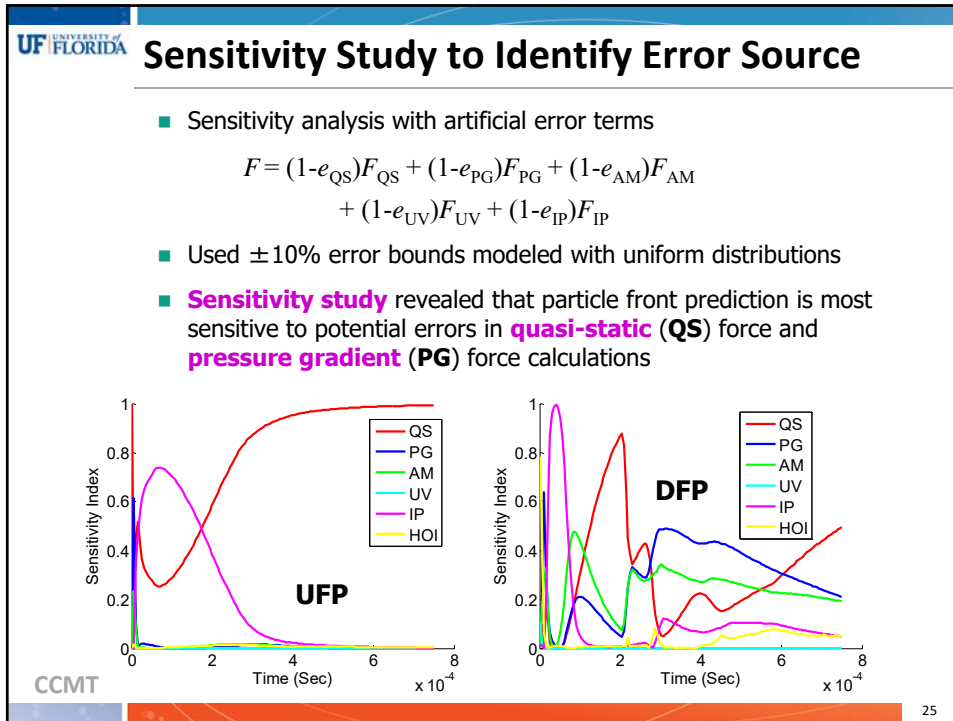
$t=29.5 \mu\text{sec}$

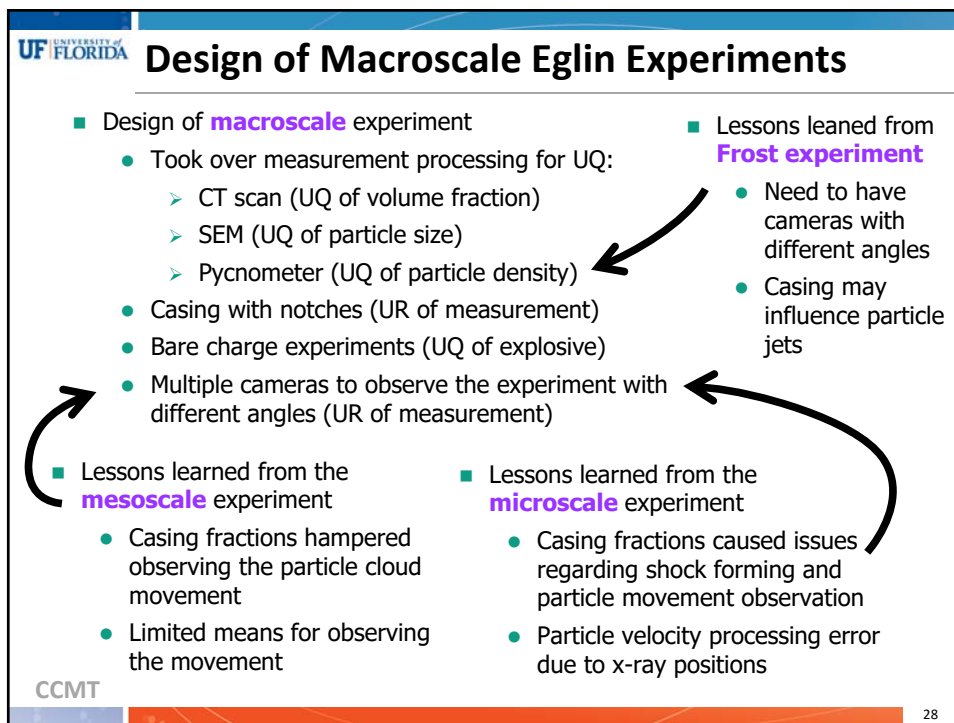
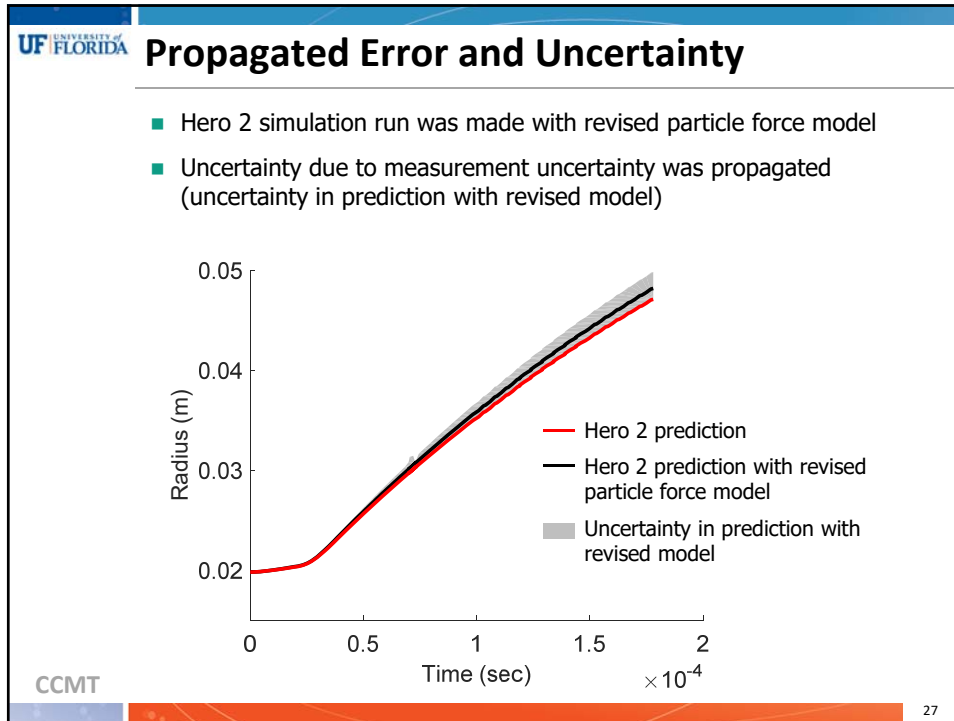
$t=129 \mu\text{sec}$

CCMT 20



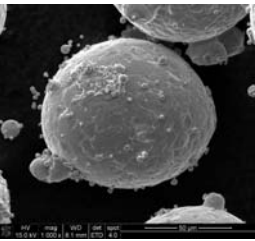




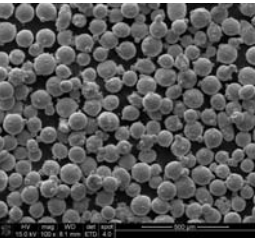


UF UNIVERSITY OF FLORIDA **UQ Driven Experimental Design**

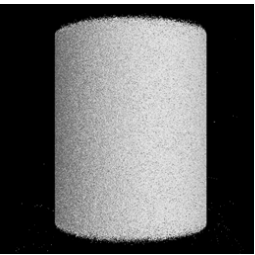
- **Steel particles chosen** for validation experiments (75-125 μm)
- Multiple vendors surveyed. Criteria: Roundness (sphericity) and narrow particle size spread
- SEM shows mostly spherical particles
- Particle bed characterization
 - SEM of particles and particle density via pycnometer
 - Particle bed mock CT scan



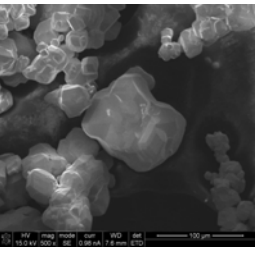
SEM of single steel particle at 1000x zoom



SEM of several steel particles at 100x zoom



Steel mock CT scan

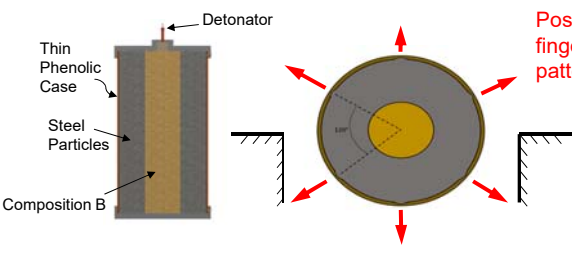


SEM of single tungsten particle, 500x zoom


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UF UNIVERSITY OF FLORIDA **UQ-Driven Casing Design**

- Casing fracture can be a possible cause of jet formation (Zhang et al. 2001, Xu et al. 2013)
- In order to minimize/control the casing effect (**uncertainty reduction**):
 - Thin phenolic tubing (3/16") was chosen for the outer casing with a failure energy estimated to be 0.06% of the energy of the explosive
 - Notches were introduced to the casing to attempt to control the failure



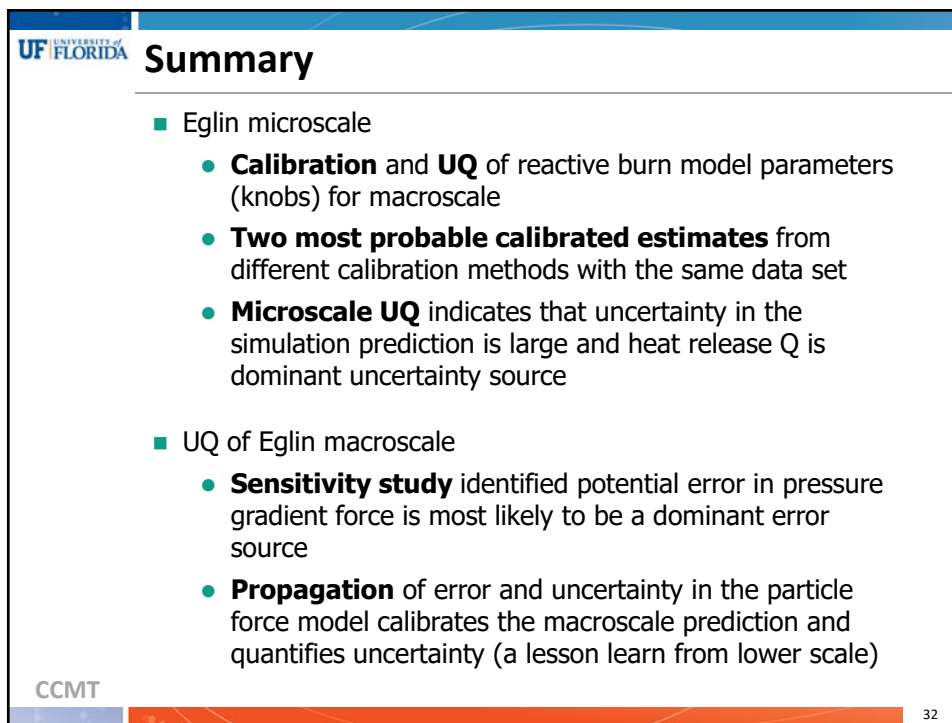
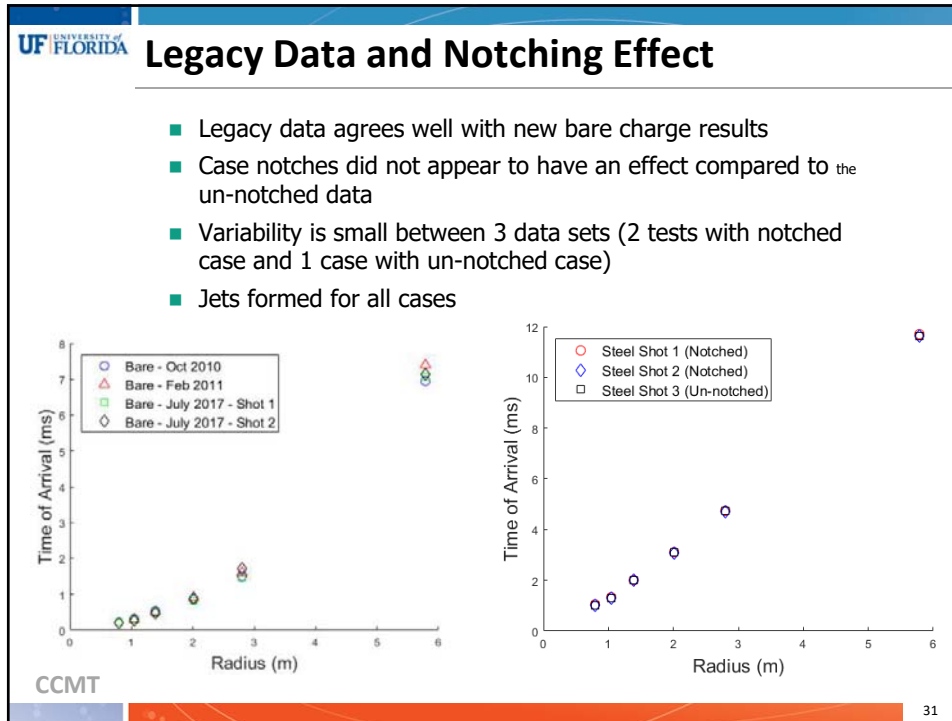
Possible finger pattern



Top view of notched casing (steel liner)

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*Do you have any
questions?*

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NNSA

Backup Slides

Camera 1 vs Camera 3

- Cameras 1 and 4 contain significant perspective errors if used to measure shock position on the centerline
- Camera 3 shows three shock structures due to end-cap effects
- To find the shock time of arrival (TOA) along the 90-degree, camera 3 is used
- Shocks analyzed normal to the ground to examine the shock data for ground effects

Camera 1 at 1.067 ms after detonation

Camera 3 at 1.067 ms after detonation

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Transverse Shock Positions

- Transverse shock positions for RZ thickness 1.7 mm and 0.88 mm
- RZ thickness 0.88 predicts higher shock position but the **initial positions are different**

Transversers shock position

Time (sec) $\times 10^{-5}$

1796 kg/m³
0.88 mm
12.5 MJ/kg

1787 kg/m³
1.7 mm
12.9 MJ/kg

Transversers shock position

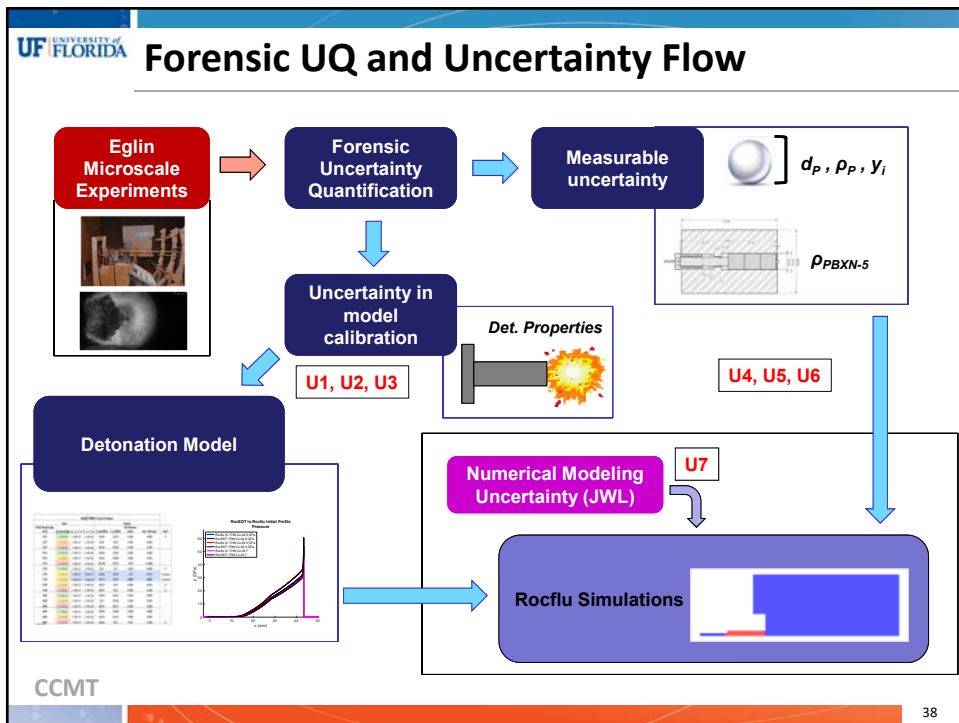
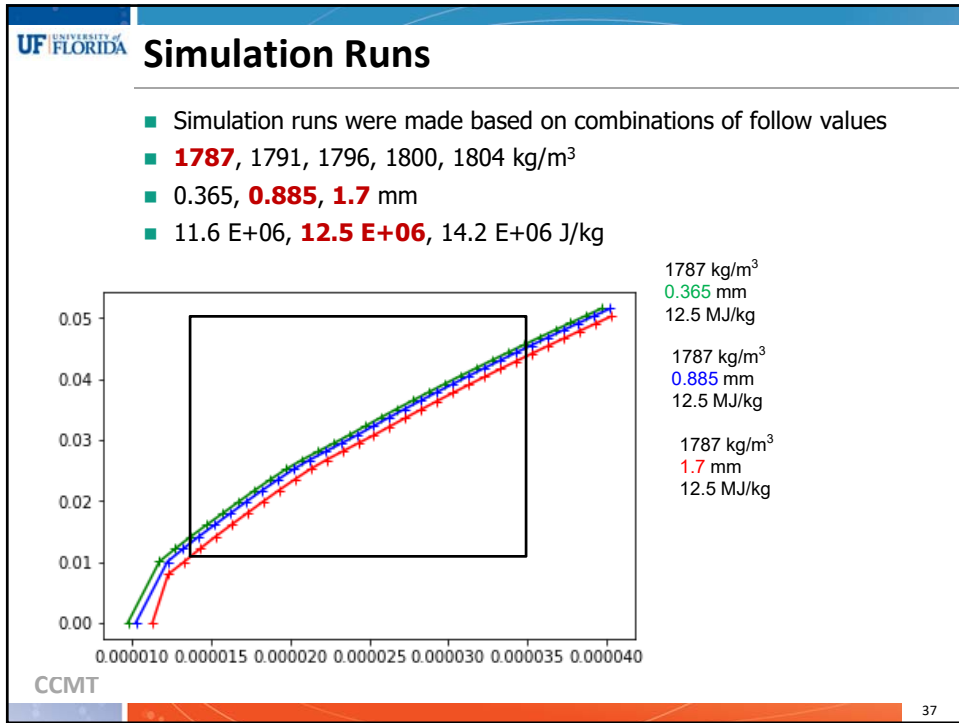
Time (sec) $\times 10^{-5}$

1787 kg/m³
0.88 mm
12.5 MJ/kg

1787 kg/m³
1.7 mm
12.5 MJ/kg

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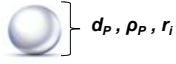


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Uncertainties

Experimental Uncertainties

- Particle diameter: mean = 2.016 mm, st. dev = 0.008 mm
- Particle density: mean = 15.8 g/cc, st. dev = 0.07 g/cc
- The initial radial position of the particle was varied to account for machining tolerance of the explosive assembly (± 0.254 mm)
- Explosive density (PBXN-5): mean = 1796 kg/m³, st. dev = 3 kg/m³



Reactive Burn Model Parameter Uncertainty

- Initial conditions for simulation of the experiment consider the explosive at the moment of complete detonation
- An external code is used to simulate the detonation of the explosive pellets
- Modeling the explosive results in additional uncertainties

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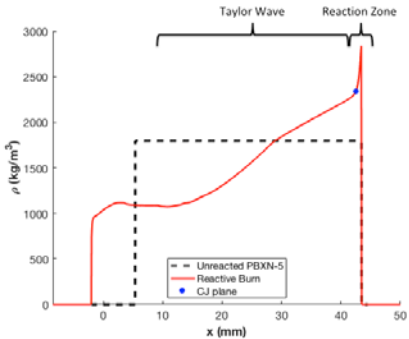
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Explosive Modeling Uncertainty

Inputs for detonation simulations.

Density (PBXN-5)	[1787,1804] kg/m ³
Heat of Reaction	[11.6, 14.2] MJ/kg
Reaction Zone Thickness	[0.365,1.700] mm

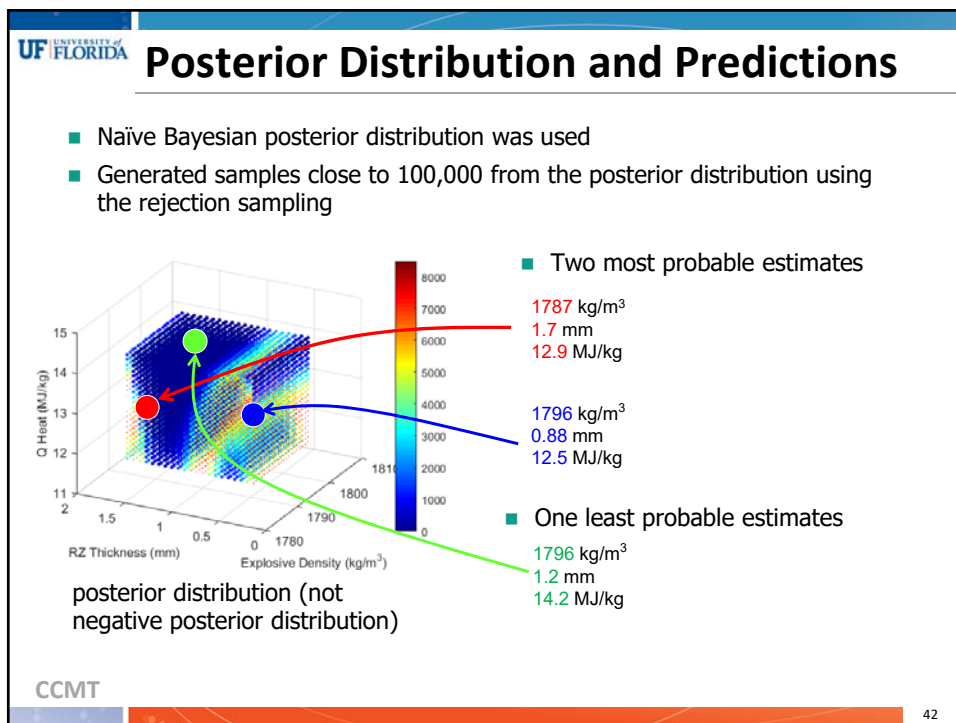
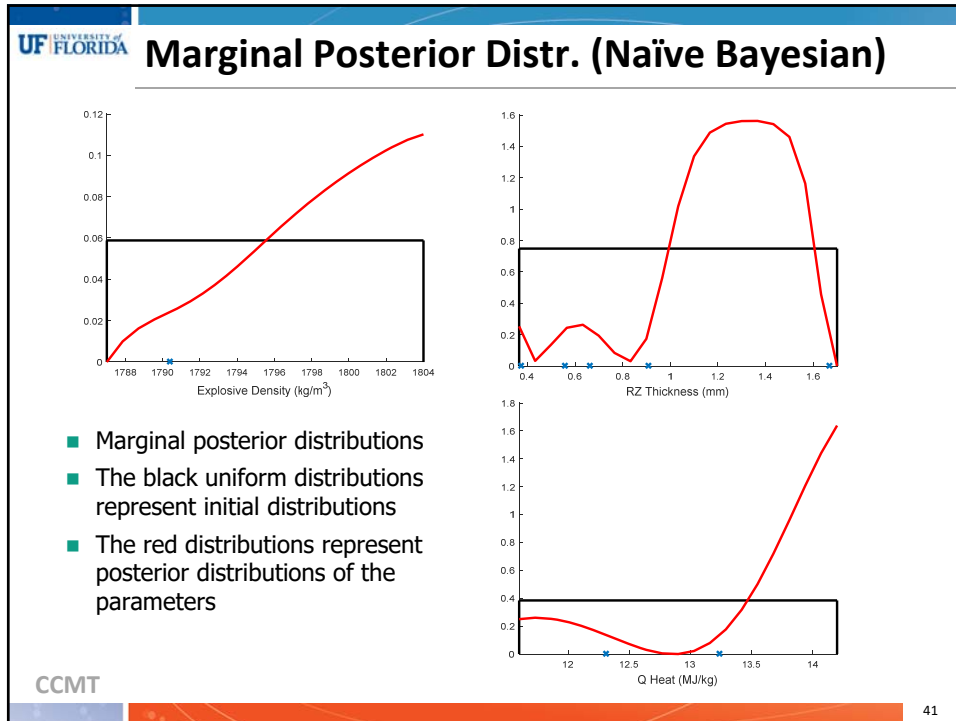


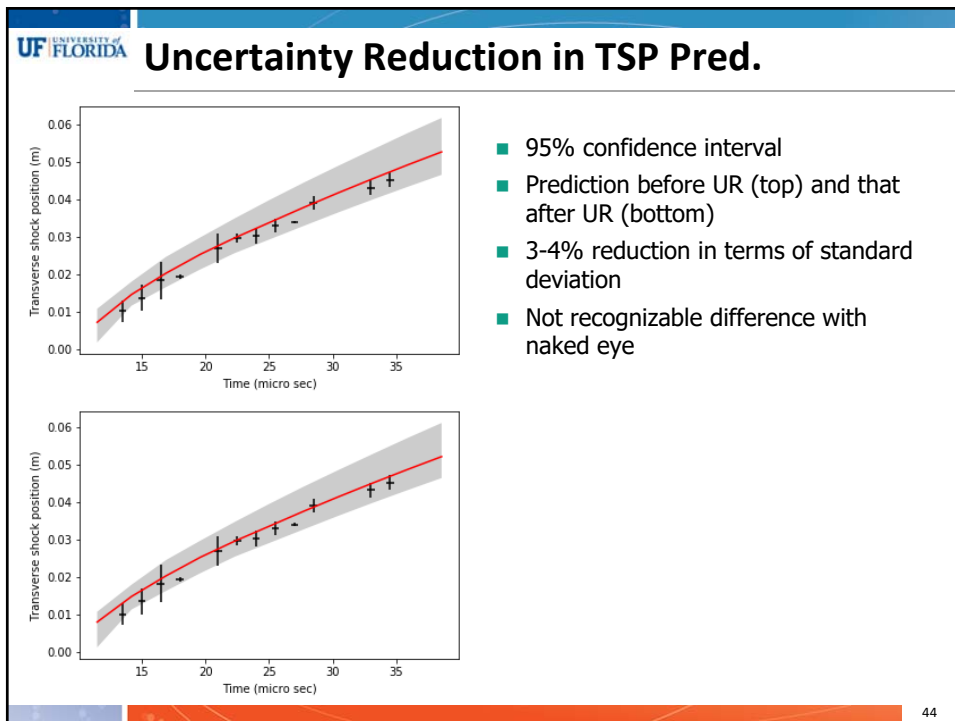
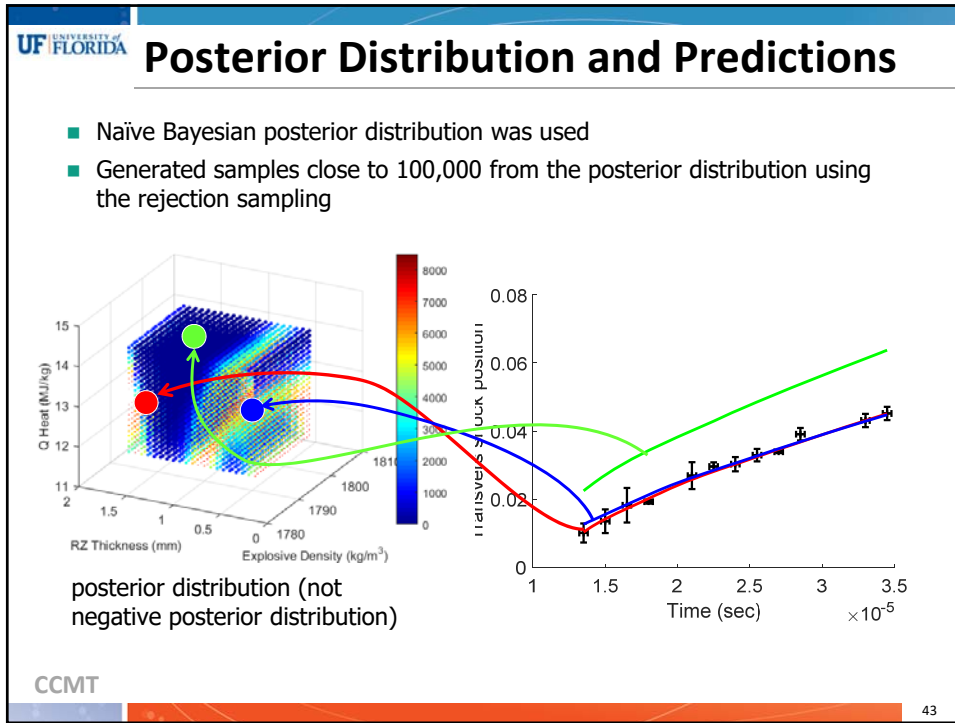
Explosive Modeling Uncertainty

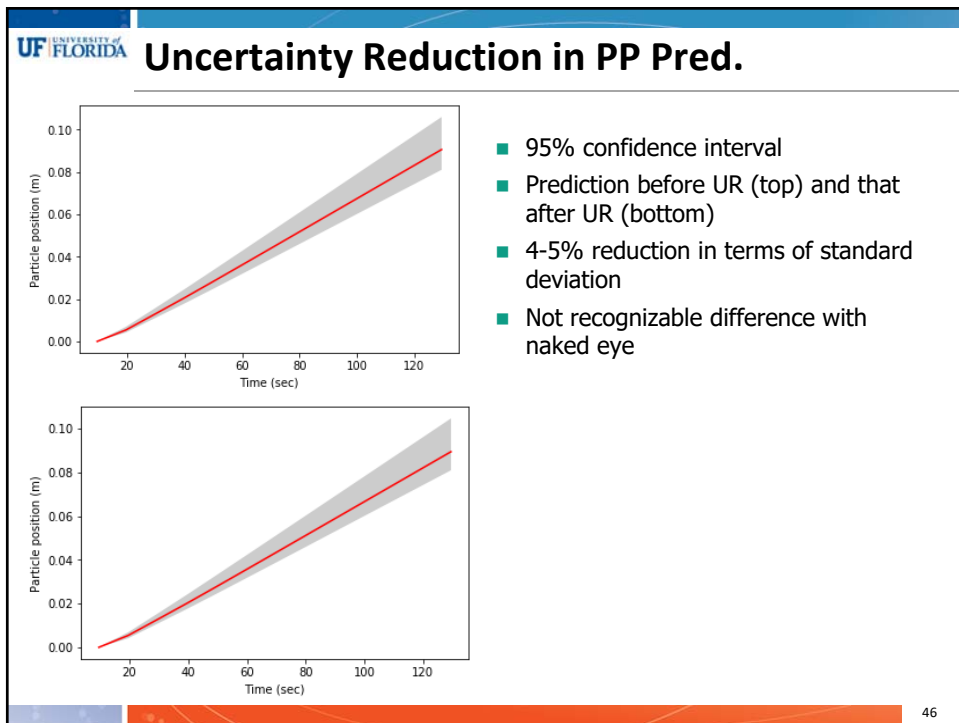
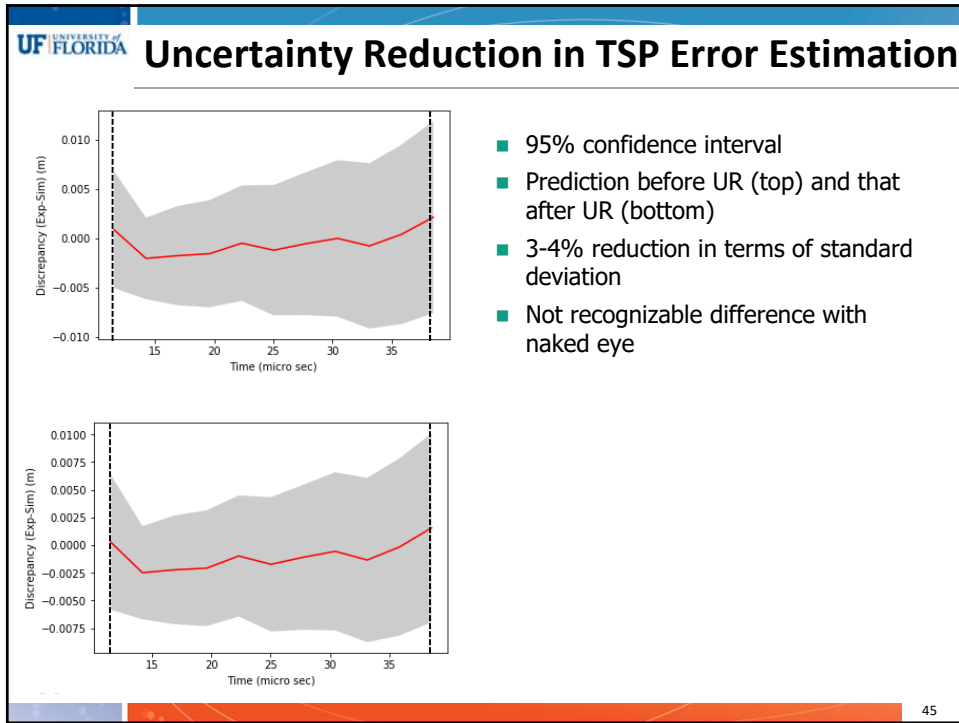
- Detonation simulation uncertain inputs:
 - Explosive density, obtained from laboratory measurements
 - Heat of reaction was calibrated for the pressure solution to reach the CJ pressure for PBXN-5 in the literature (32.7 to 39.3 GPa)
 - Reaction rate parameters govern the reaction zone thickness, the literature gives this thickness as 0.1 to 0.7 mm for PBXN-5

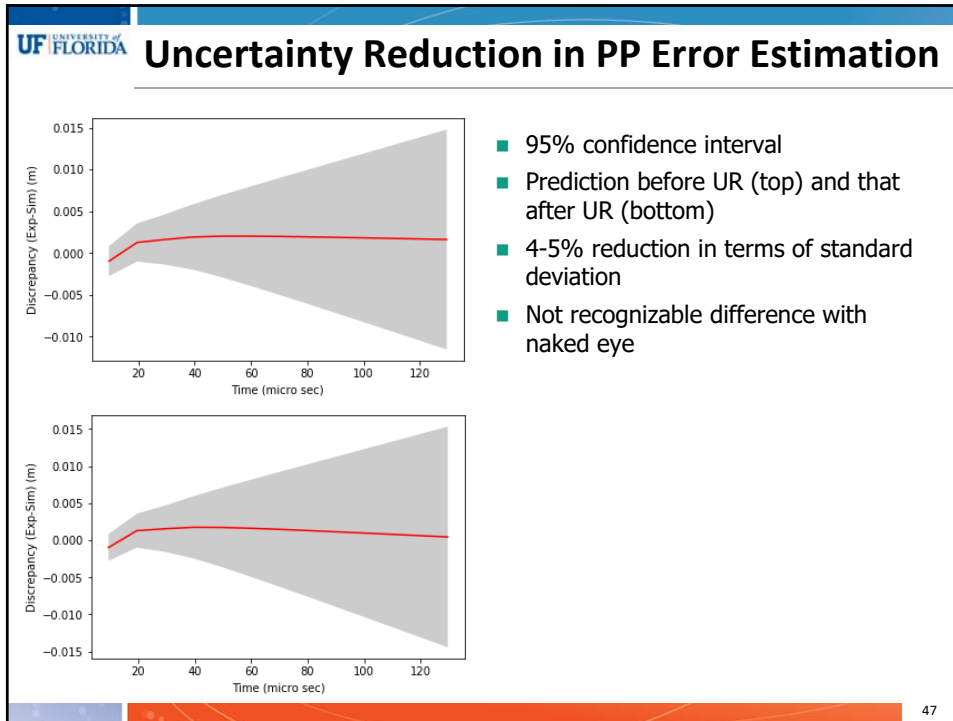
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UF UNIVERSITY OF FLORIDA **Uncertainty in Discrepancy Function (PP)**

- Variance of the discrepancy function

$$V_{disc}(t) = Var(Y_{sim}(t) - Y_{meas}(t))$$

- Sum of variances

$$V_{disc}(t) = \sum_i V_{sim}^{(i)} + \sum_j V_{exp}^{(j)}$$

- Sensitivity indices

$$1 = \sum_i S_{sim}^{(i)} + \sum_j S_{sim}^{(j)} \quad S_{sim}^{(i)} = \frac{V_{sim}^{(i)}(t)}{V_{disc}(t)} \quad S_{sim}^{(j)} = \frac{V_{exp}^{(j)}(t)}{V_{disc}(t)}$$

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Uncertain Parameters

Parameter	Quantity	Method
U1: Explosive density	mean = 1795 kg/m ³ , std = 2.9 kg/m ³ , Bi-modal	Derived, 26 samples
U2: RZ thickness	[0.365, 1.4] mm	Expert opinion
U3: Heat release Q	[11.6, 14.2] MJ/kg	Literature and opinion
U4: Particle diameter	mean = 2.0156 mm, std = 0.0073 mm, Weibull	Direct measurement Micrometer, 52 samples
U5: Particle density	mean = 15540 kg/m ³ , std = 250 kg/m ³ , Normal	Pycnometer Gas Pycnometer, 12 samples
U6: Initial radial position	[0, 0.254] mm	
U7-11: JWL	A, B, w, R ₁ , R ₂	Literature and opinion
...

- Listed uncertainties including uncertainties in simulation parameters (knob)
- Kyle** actively involved in measurements and uncertainty estimation (forensic UQ)

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CMT-nek: CS Update

Tania Banerjee

Computer and Information Science and Engineering
University of Florida
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Members



*Keke
Zhai*



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*Adeesha
Malavi*



*Tania
Banerjee*



CCMT *David
Zwick*



*Jason
Hackl*



*Sai
Chenna*



*Sanjay
Ranka*

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Accomplishments (past year)

- Scalable dynamic load balancing
 - Improved overhead
 - Extensive testing on CMT-nek for particles
- Multilevel Memory Optimization
 - CMT-bone on KNL
 - Bootstrapping techniques for efficient Matrix multiplication
- GPUization and Hybrid Architectures
 - From CMT-bone to CMT-nek

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Overview of Dynamic Load Balancing

Step 1: Domain decomposition
Happens during initialization only

Step 2: Elements/particles to processor mapping
Happens during initialization and on every remapping

- 1) Tightly coupled - Keke
 - a. Compute element load theoretically
 - b. Compute element load using actual computation time
- 2) Loosely coupled - David

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Overview of Dynamic Load Balancing

Step 3: Decide when to trigger a remap

1. Rebalance after every k time steps (user set up)
2. Rebalance automatically after certain time steps (adaptive load balancing)
3. Rebalance when the difference in time between the fastest and the slowest processor exceeds a threshold.

Step 4: Transfer elements and particles and reset other data structures

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Domain Decomposition

P1 – 8 particles

P2 – 11 particles

P3 – 8 particles

P4 – 5 particles

P5 – 8 particles

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Elements to Processor Mapping

P1 – 8 particles P2 – 8 particles P3 – 8 particles

P4 – 8 particles P4 – 8 particles

(b) Elements to processor mapping, after taking into account both elements and particles

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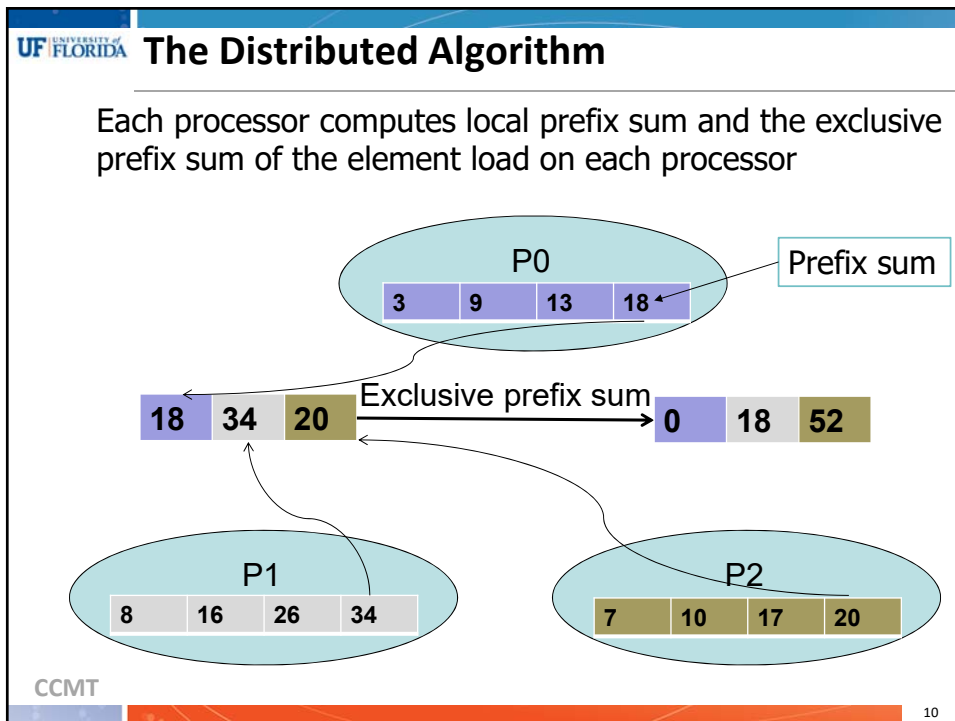
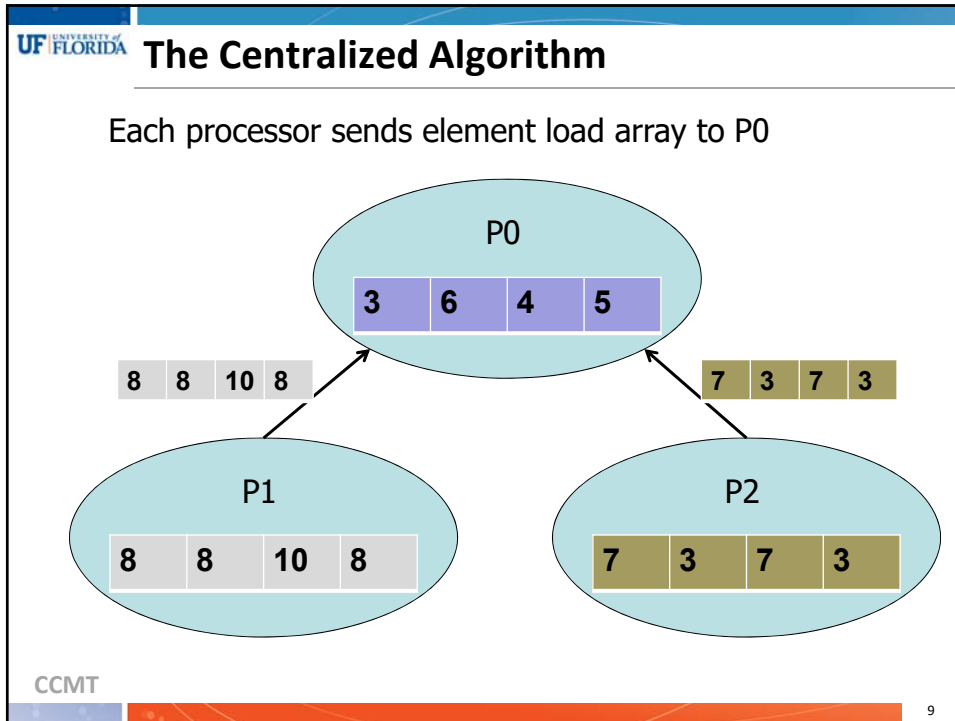
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Overview of Element to Processor Mapping Algorithm

- Centralized
 - Easy to accomplish
 - There is a bottleneck where only processor P0 is working
 - Have more information to achieve better decision
- Distributed
 - There is no bottleneck at all
 - Each processor communicate with each other to get part information
 - Use limited information to make decision
 - MPI_allgather is taking most of the time on Quartz
- Hybrid
 - Combination of centralized and distributed
 - Utilize broadcast in replace of MPI_allgather to reduce communication time

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UF UNIVERSITY OF FLORIDA **The Hybrid Algorithm**

The hybrid algorithm is similar to the [distributed algorithm](#) except for the last step that get the global element->processor mapping. Each processor sends the mapping to P0.

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UF UNIVERSITY OF FLORIDA **Infrastructure for mapping and remapping**

- Recompute partitions
 - Compute element load
 - Partition element array
 - Assign element array sections to processors
 - New element to processor mapping

↓

- Initialize data structures
 - Initialize most variables
 - Setup domain topology
 - Generate geometry
 - Recompute Jacobians
 - Initialize DG machinery

↓

- Transfer arrays

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Infrastructure for array transfer

- Input:
 - A = array to be transferred
 - L = length of the array
 - G = new element to processor mapping


Pack array A

- For each element
 - Get the global element ID
 - Get the current and the future processor ID of the element
 - If the two IDs are different then
 - Copy element data to be transferred to a temporary array



Transfer

- crystal_tuple_transfer



Unpack received array

- Sort the element array according to global element ID
- Copy the received array back to input array according to local element ID

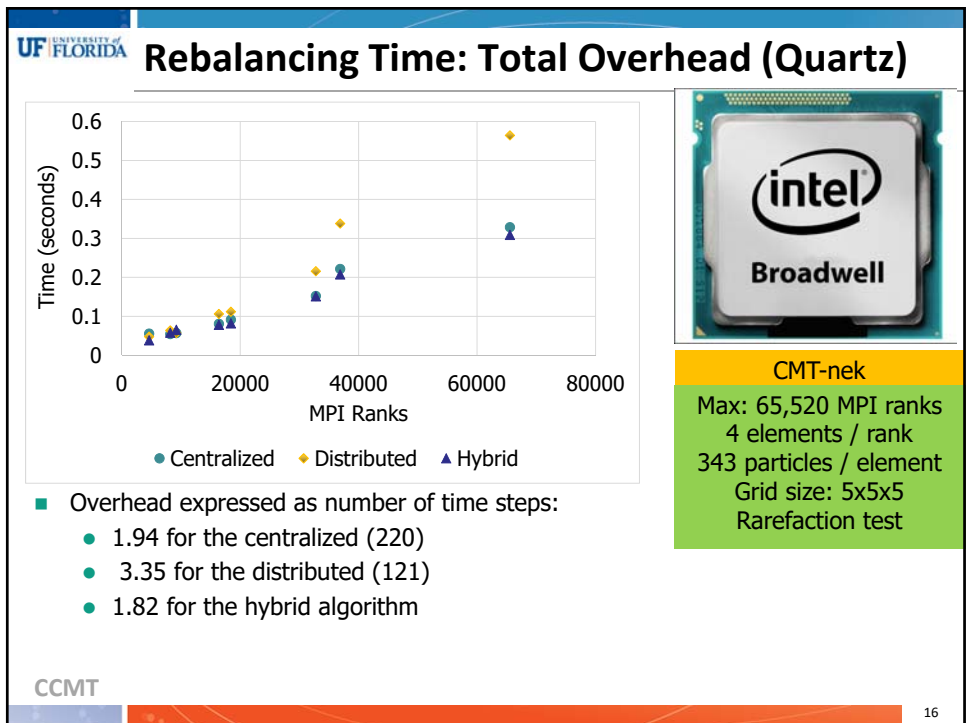
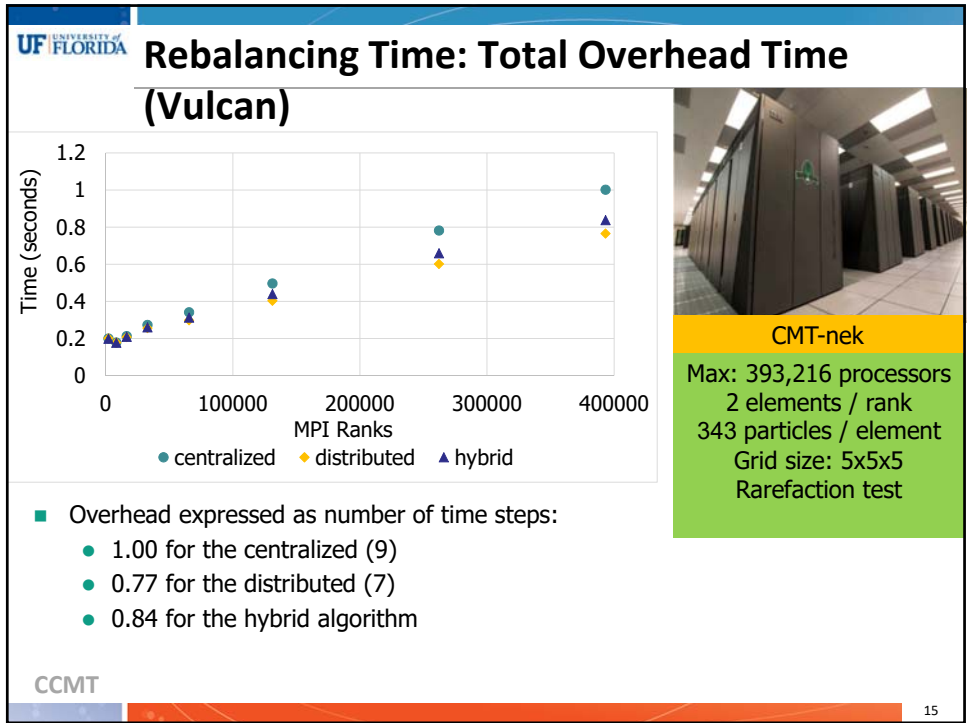
CCMT 13

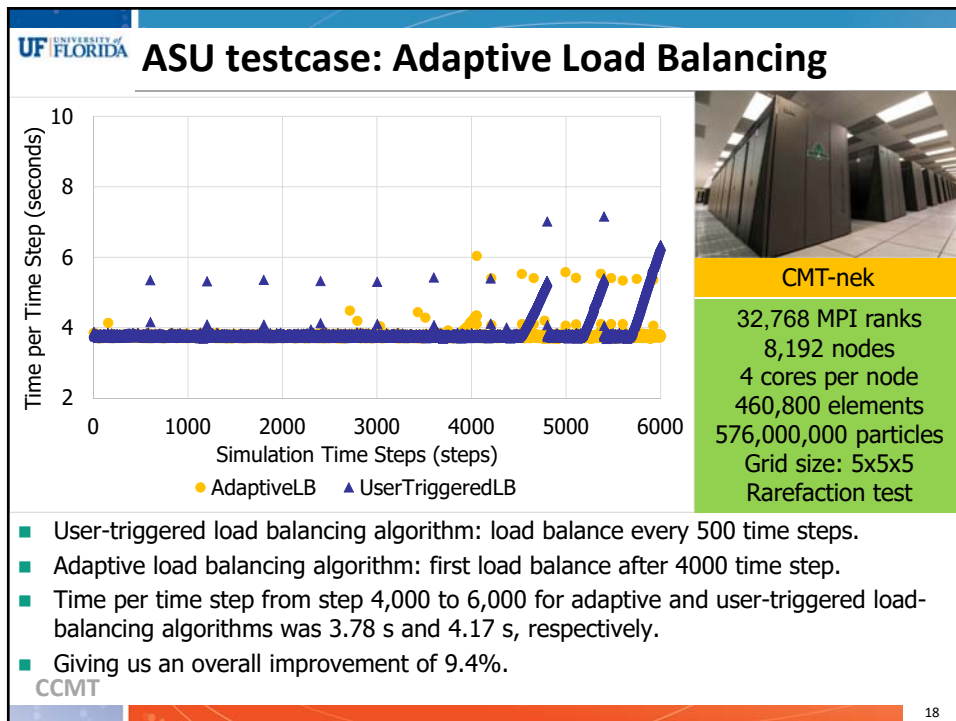
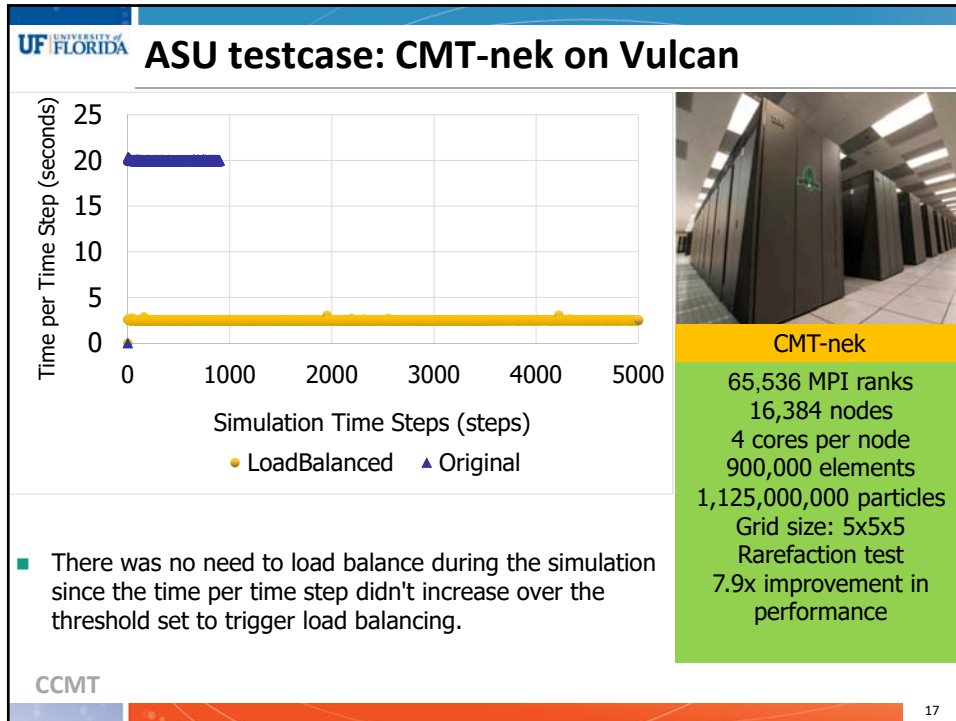
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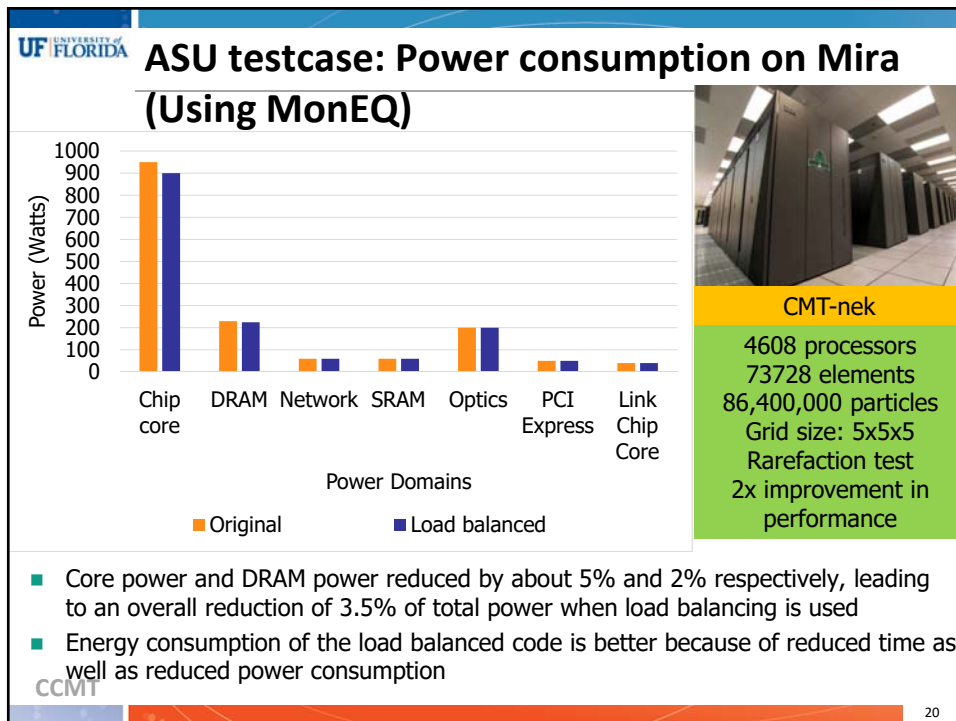
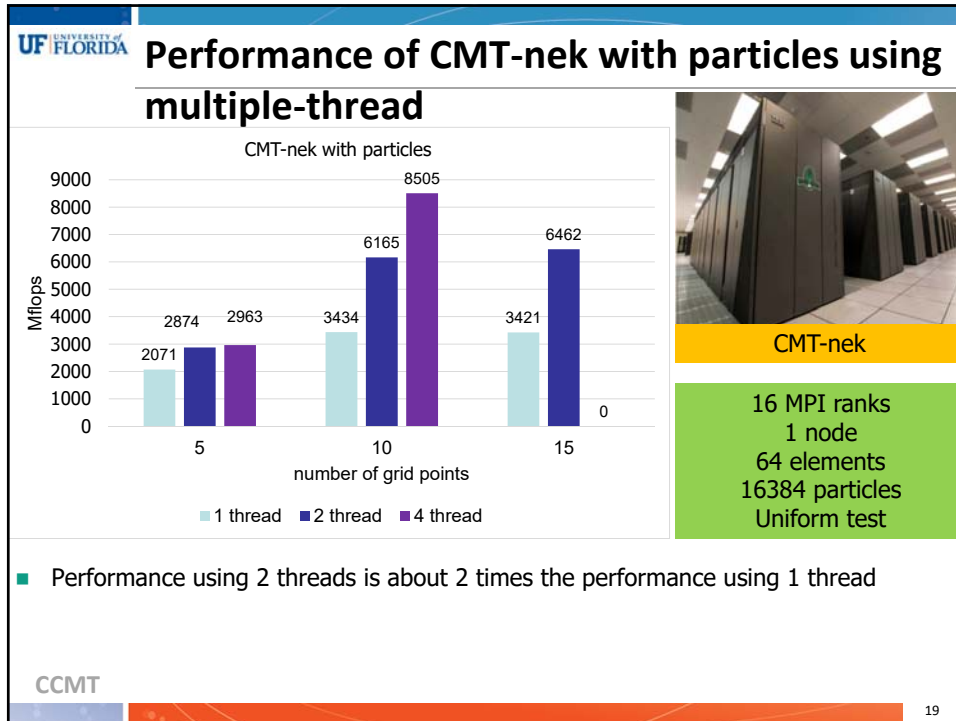
Reduction in Remap Overhead

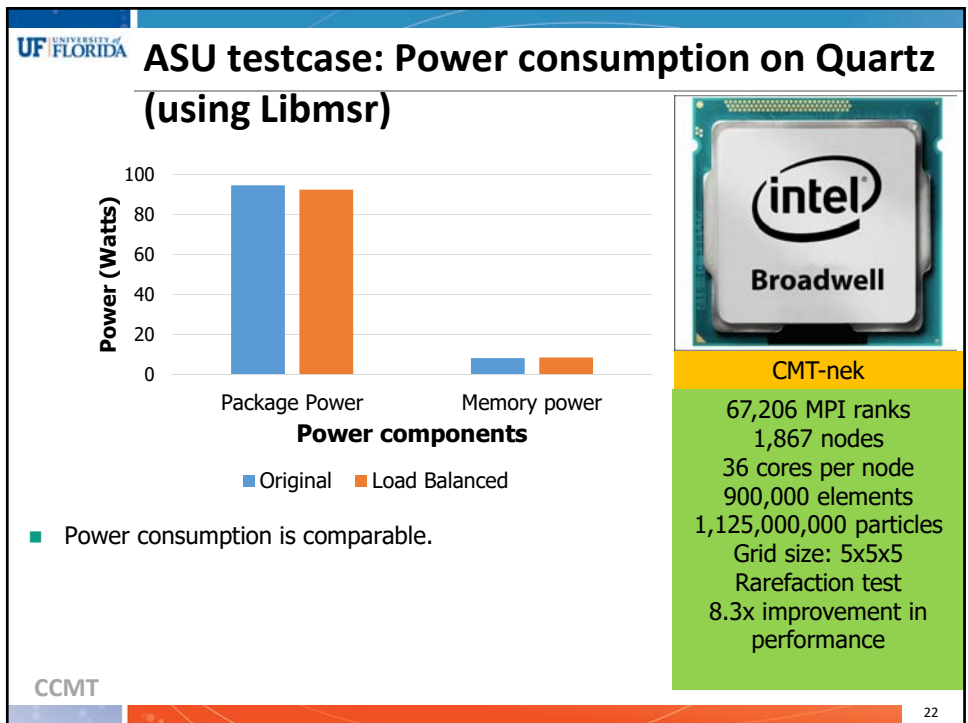
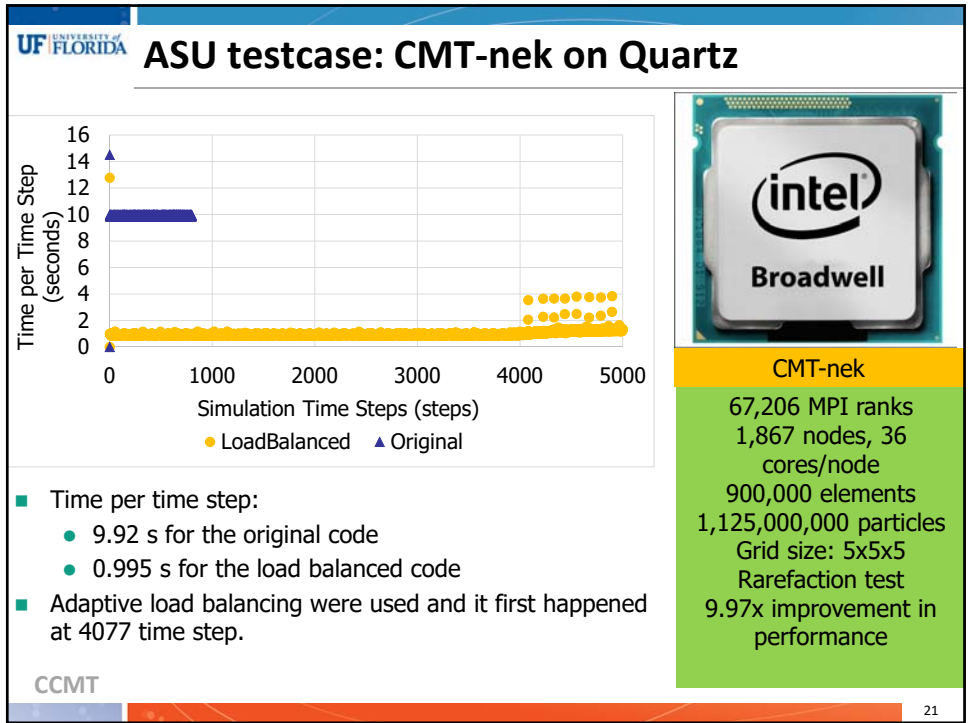
- Previously, each remap step invoked calls for reading ASCII files
 - REA file: parameters pertaining to the simulation, stores the mesh and boundary conditions
 - MAP file: contains information that helps to construct global ID of elements
- We fixed that so that the data in these files gets stored and the files are never read again
- Increases data storage slightly (in the order of few MBs depending on problem size)

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UNIVERSITY OF FLORIDA **Multilevel Memories**

- What is a multi-level memory?
 - Main memory is composed of two or more types of memories
- Long term implications
 - Increase capacity
 - Decrease latency
 - Increase bandwidth
 - Decrease cost
 - Decrease power consumption
 - Challenges: Software and hardware challenges
- Why are they important for exascale
 - High bandwidth, high capacity, lower power consumption are attractive features

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UNIVERSITY OF FLORIDA **Multi-Level Memories**

KNL Overview

TILE		
2 VPU	CHA	2 VPU
Core	1MB L2	Core

Chip: 36 Tiles interconnected by 2D Mesh
Tile: 2 Cores + 2 VPU/core + 1 MB L2

Memory: MCDRAM: 16 GB on-package; High BW
DDR4: 6 channels @ 2400 up to 384 GB

IO: 36 lanes PCIe® Gen3, 4 lanes of DMI for chipset

Node: 1-Socket only

Fabric: Intel® Omni-Path Architecture on-package (not shown)

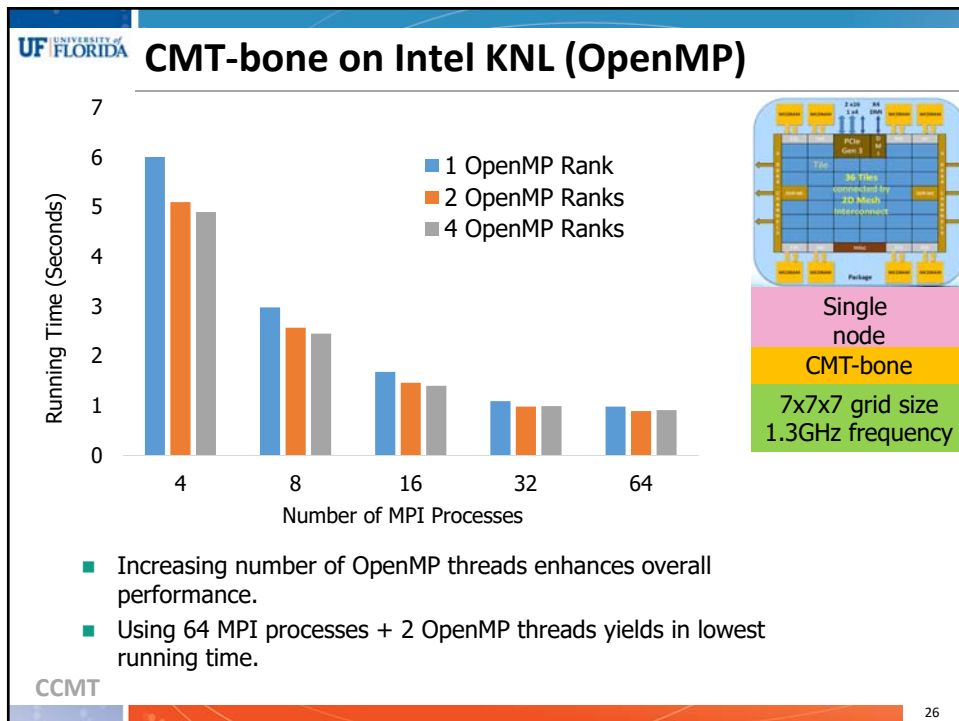
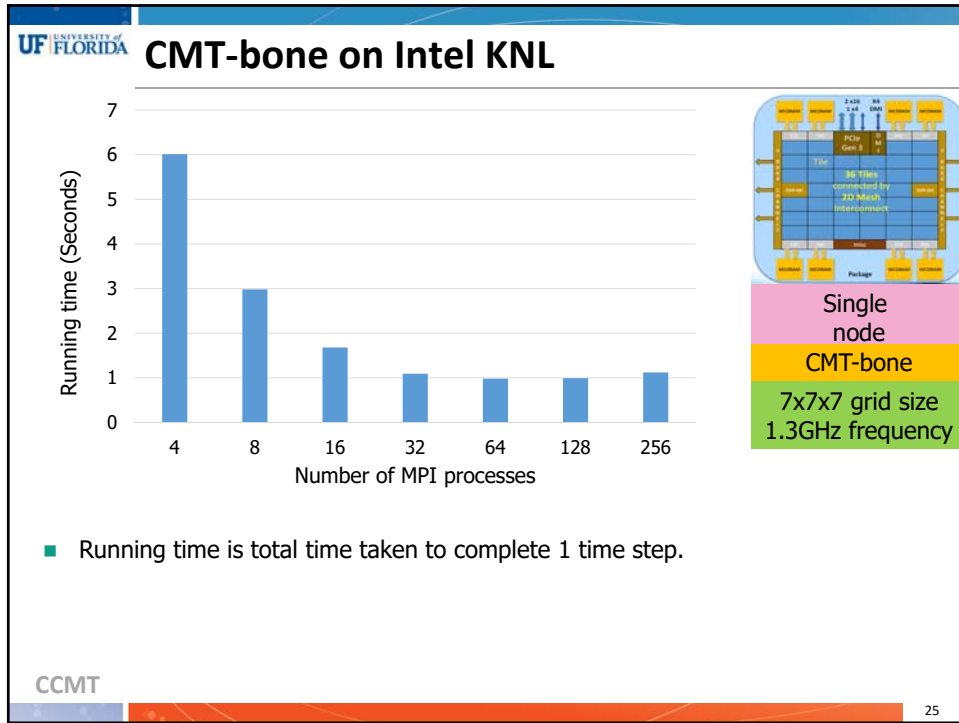
Vector Peak Perf: 3+TF DP and 6+TF SP Flops MCDRAM ~5X Higher BW than DDR

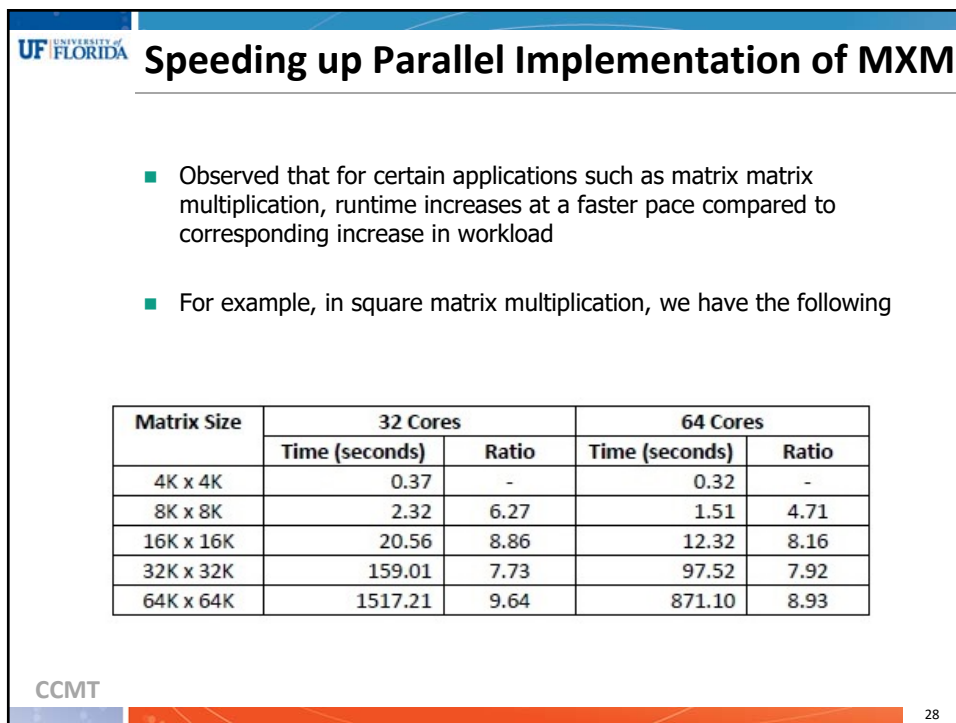
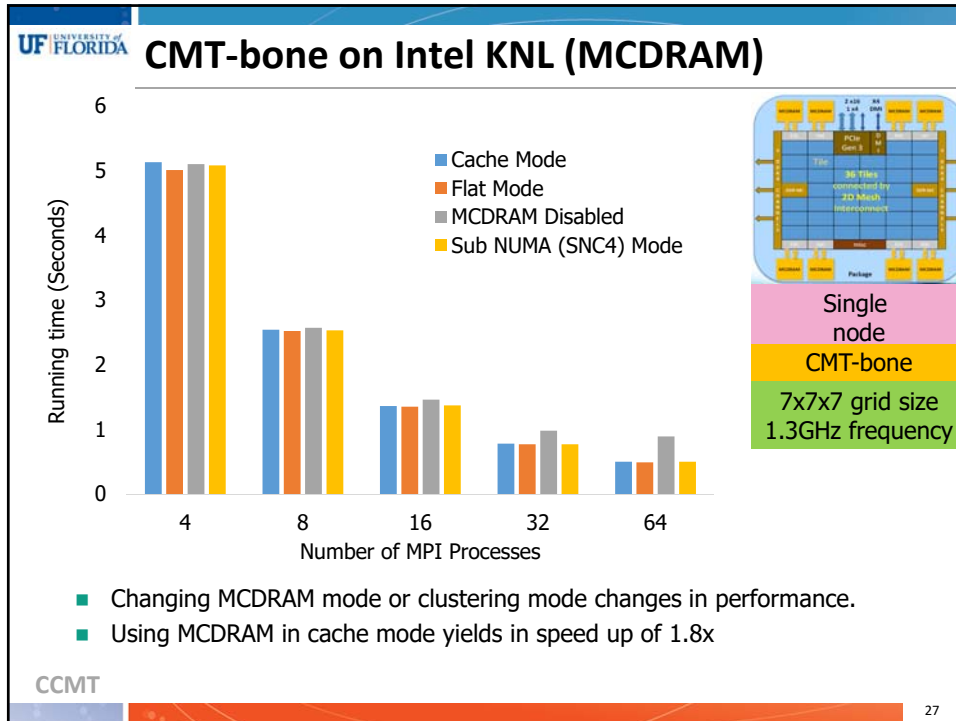
Scalar Perf: ~3x over Knights Corner

Streams Triad (GB/s): MCDRAM : 400+; DDR: 90+

Source: Intel. All products, computer systems, dates and figures specified are preliminary based on current expectations, and are subject to change without notice. KNL data are preliminary based on current expectations and are subject to change without notice. Binary Compatible with Intel Xeon processors using Haswell Instruction Set (except TSX). Bandwidth numbers are based on STREAM-like memory access pattern when MCDRAM used as flat memory. Results have been estimated based on internal Intel analysis and are provided for informational purposes only. Any difference in system hardware or software design or configuration may affect actual performance. *Other names and brands may be claimed as the property of others.

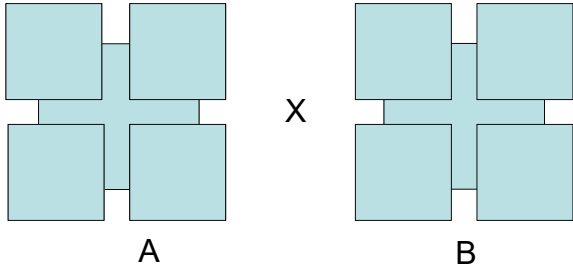
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UF UNIVERSITY OF FLORIDA **Boot Strapping algorithm**

- Condition for applicability:
 - The application should be decomposable
 - Runtime should increase at a faster rate than workload
 - MXM is an ideal candidate
- It is a multi-level process:
 - Two-level algorithm

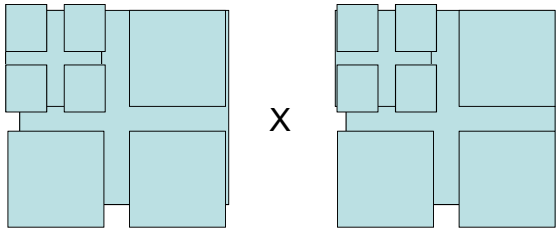


A X B

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UF UNIVERSITY OF FLORIDA **Boot strapping algorithm**

- Three – level algorithm



A X B

- The process may be generalized to more levels

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Optimization to Reduce Cache Misses

- Matrix multiplication at the leaf level is done in an order that reduces cache misses

(a) (b) (c)

(d) (e) (f)

(g) (h) (i)

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Theoretical Performance Analysis

- Using a simple performance model we can predict the expected speedup
- If T_2 is the time taken by the two level algorithm, and T_1 is the time taken by the original algorithm, then we can show that:

$$\frac{T_2}{T_1} = \frac{1+rbk\delta}{rbk\delta}$$

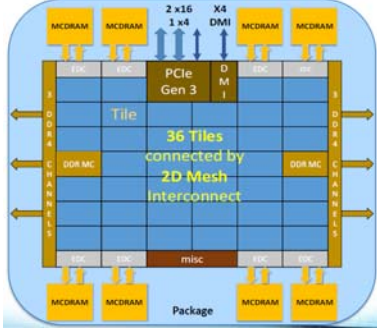
- Where, r = ratio of time taken to solve the original problem using all processors versus the time taken to solve the smallest decomposed matrix all the processors
- b = number of parallel multiplications
- k = ratio of the time taken to solve the original problem using 1/processor versus the time taken to solve the decomposed matrix using the same number of processors
- δ = combination time taken to solve the original problem.

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Experimental Setup

- Target application: Intel MKL cblas-dgemm on KNL platform
- Combinations of memory and clustering modes tested
- OpenMP parallel processing
- 64 cores
- Up to 16Kx16K : average over 10 runs
- 32Kx32K and 64Kx64K : average over 5 runs

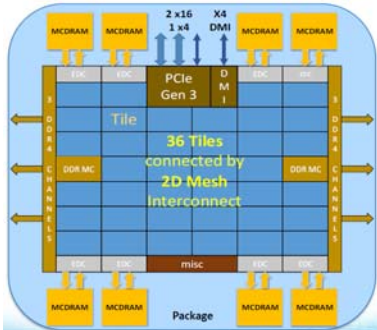


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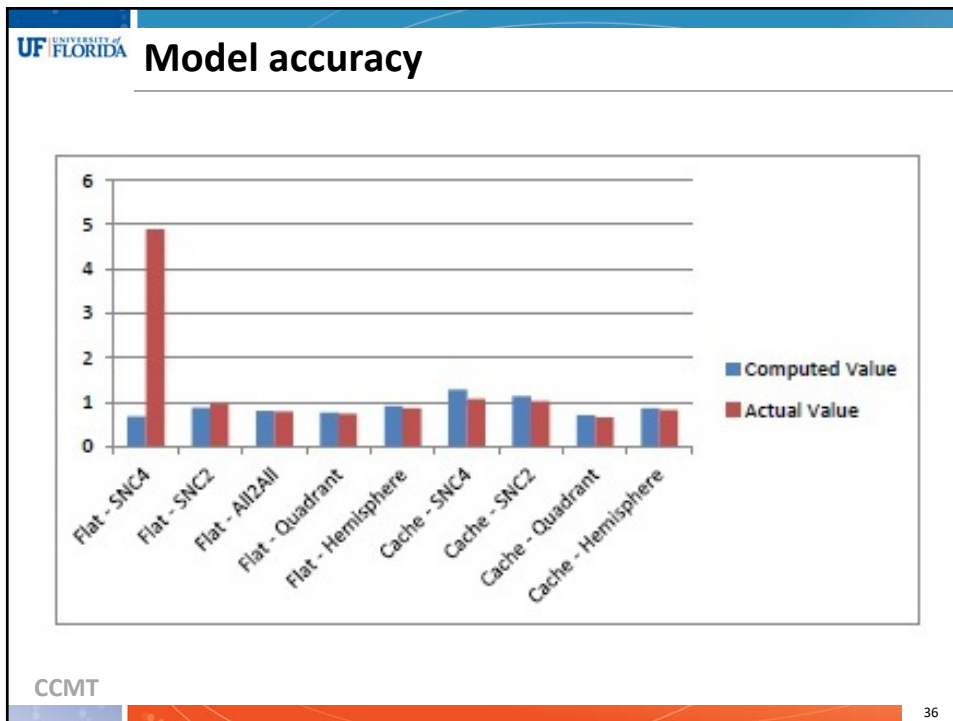
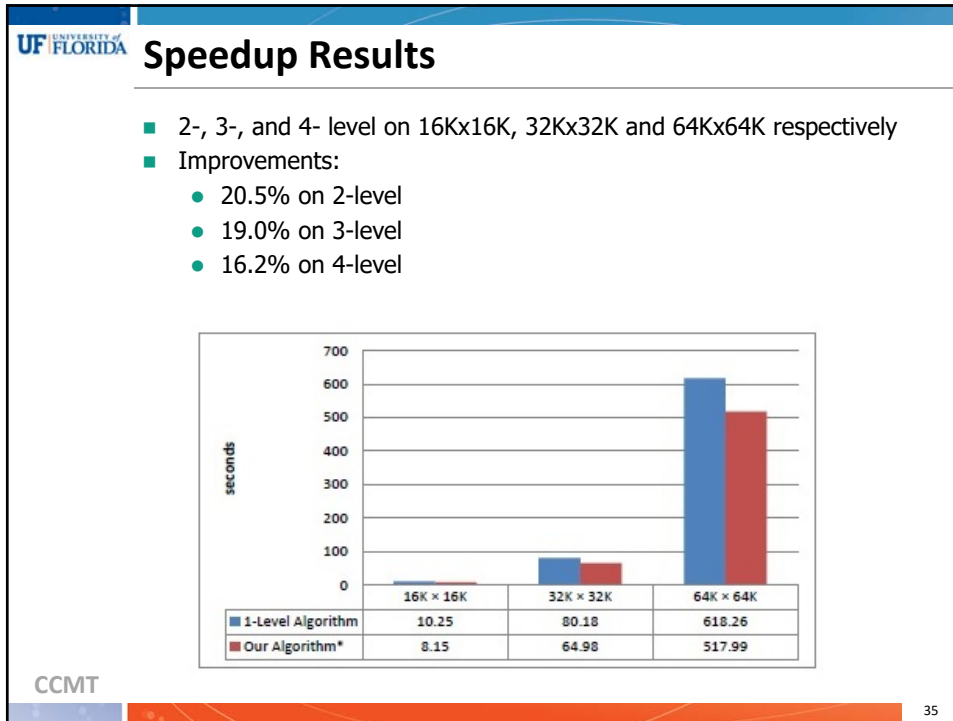
KNL Modes

- Memory modes:
 - Flat mode
 - Cache mode
 - Hybrid mode
- Clustering modes:
 - All-to-all
 - Quadrant
 - Hemisphere
 - SNC2
 - SNC4
- Clustering modes defines the affinity properties of tiles, tag directories and memory controllers



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UNIVERSITY OF FLORIDA **CMT-nek CPU-GPU interaction model**

- GPU performs computational work
- Host CPU core is used for inter process communication
- Idle CPU cores will be used for computation

The diagram illustrates the interaction between a host CPU and a GPU. On the left is an AMD Opteron CPU, and on the right is a Tesla K20x GPU. A large blue arrow labeled 'Computational work' points from the CPU to the GPU. A smaller blue arrow labeled 'Data for communication' points from the GPU back to the CPU. A third blue arrow labeled 'Idle cores utilized for computation' points from the CPU to itself, indicating that the CPU can perform computation when not communicating with the GPU.

Host:
AMD Opteron
16 cores
2.2GHz clock frequency

Tesla K20x:
14 Processors
192 Cores
48k shared memory
64k registers
1310 GFLOP/s Peak
732MHz clock frequency
6GB DDR5

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UNIVERSITY OF FLORIDA **CMT-nek on GPU**

- Based on the implementation of CMT-bone on GPU for fluid, with support for new enhancements (viscous flow)
- Changed matrix multiplication routines to cuBLAS library.
 - cublasDgemmBatched : $(n_1, n_2) \times (n_2, n_3 * n_{elt})$ multiplication
 - cublasDgemmStridedBatched : $(n_1, n_2 * n_{elt}) \times (n_2, n_3)$
- New code for particles
- High level Architecture
 - Assign one element to a SM to minimize array data reading from the global memory
 - Dispatch threads equal to number of grid points ($l_x * l_y * l_z$) when calculations are required for each grid point.
 - Dispatch threads equal to (number of faces * $l_x * l_z$) when calculations are required for each face


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Challenges in porting CMT-nek

- Hardware perspective: Memory constraints
 - 48KB cache per SM :
 - For 512 grid points per element, one array requires 4k memory. Can hold only 12 such arrays. Total array requirement is about 50.
 - Solution: Reduce grid points or split kernels so a kernel requires only few arrays.
 - 64K registers per SM
 - For 512 grid points per element, one thread can have 125 registers. Additional registers will be stored in the global memory
 - Solution: Split kernels, change the order of calculations to reuse registers.
 - 6 GB Main Memory (Tesla K20X)
 - Can be a constraint when using higher polynomial order and large number of elements with particles.

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Challenges in porting CMT-nek

- Coding perspective
 - A number of nek5000 libraries such as routines for spectral methods must be ported to GPU for full GPU support.
 - MPI calls require GPU kernels to be stopped and data to be transferred to CPU
- Validation
 - Floating point rounding off errors make GPU results slightly different (compared to CPU)
 - These small differences propagate and increase with the time steps making it difficult to compare.
 - We used visualization to compare the two simulations as a starting point

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Preliminary results

- CMT-nek total time for 100 time steps(s)
- 125 elements
- For different $lx1$ (Initial size of the grid) and lxd (finer mesh) values

$(lx1, lxd)$	1 core	4 cores	16 cores	1GPU
(6,10)	37.47	14.47	7.15	9.77
(8,12)	86.78	32.68	15.00	14.66
(10,16)	205.94	74.27	30.72	27.57

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Summary of Accomplishments

- Scalable dynamic load balancing
 - Improved overhead
 - Extensive testing on CMT-nek for particles
- Multilevel Memory Optimization
 - CMT-bone on KNL
 - Bootstrapping techniques for efficient Matrix multiplication
- GPUization and Hybrid Architectures
 - From CMT-bone to CMT-nek

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Publications




1. Tania Banerjee and Sanjay Ranka, Genetic Algorithm based Autotuning Approach for Performance and Energy Optimization, Proceedings of International Green and Sustainable Computing, 2015
2. Tania Banerjee, Mohammed Gadou, Sanjay Ranka, A Genetic Algorithm based Approach for Multi-objective Hardware/Software Co-optimization, Journal of Sustainable Computing, Vol 10, June 2016, 36 – 47.
3. Tania Banerjee, Jacob Rabb and Sanjay Ranka, Multi-objective Optimization of Spectral Solvers on Hybrid Multicore Platforms, Journal of Sustainable Computing, Vol 12, Dec 2016, 10 - 20.
4. Tania Banerjee, Jason Hackl, Mrugesh Sringarpure, Tanzima Islam, S. Balachandar, Thomas Jackson, Sanjay Ranka, CMT-bone -- A Proxy Application for Compressible Multiphase Turbulent Flows, Proceedings of HiPC, 2016
5. Mohamed Gadou, Tania Banerjee, Sanjay Ranka, Multi-objective Optimization of CMT-bone on Hybrid Processors, Proceedings of IGSC 2016
6. Tania Banerjee, Jason Hackl, Mrugesh Sringarpure, Tanzima Islam, S. Balachandar, Thomas Jackson, Sanjay Ranka, A New Proxy Application for Compressible Multiphase Turbulent Flows. Sustainable Computing: Informatics and Systems, 2017
7. Mohamed Gadou, Tania Banerjee, Meena Arunachalam, Sanjay Ranka, Multiobjective evaluation and optimization of CMT-bone on multiple CPU/GPU systems, Journal of Sustainable Computing: Informatics and Systems, 2017
8. Keke Zhai, Tania Banerjee, David Zwick, Jason Hackl and Sanjay Ranka, Dynamic Load Balancing for Compressible Multiphase Turbulence, accepted International Conference on Supercomputing, 2018
9. Mohamed Gadou, Sankeerth Mogili, Tania Banerjee, Sanjay Ranka, Multi-objective Optimization on Hybrid Systems using DVFS, in preparation
10. Mohamed Gadou, Meena Arunachalam, Tania Banerjee, Sanjay Ranka, Multiobjective evaluation and optimization of CMT-bone on Intel Knights Landing, accepted in IGSC 2018.



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Do you have any questions?

General n-level algorithm

Algorithm 1 n_level algorithm

```

1: function  $GEN_n(A, C)$   $\triangleright$  Problem  $A$  of size  $s$  is divided
   into  $s/b_s$  blocks of sizes  $b_s$ .  $C$  is the output.
2:   for  $I = 1$  to  $s/b_s$  do
3:     Call  $GEN_{n-1}$  over  $A[I]$ 
4:   end for
5:   do in parallel
6:     Combine the results of  $GEN_{n-1}$ 
7:     to form the final solution  $C$ 
8:   end do
9: end function

```


CPU Experiments

- Constant of proportionality, for total execution time

f \ n	1	2	4	8	12
1200	0.975	0.965	0.965	0.935	0.905
1500	0.985	0.985	0.975	0.955	0.915
1800	0.995	0.995	0.990	0.985	0.975
2100	0.990	0.990	0.990	0.990	0.970

- Constant of proportionality, for communication time

f \ n	1	2	4	8	12
1200	0.82	0.88	0.82	0.72	0.62
1500	0.86	0.86	0.86	0.77	0.70
1800	0.90	0.90	0.90	0.83	0.80
2100	0.95	0.95	0.95	0.92	0.90



Hardware-Software Autotuning

Algorithm: dudr-4loop

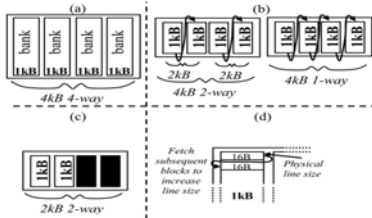
```

do k = 1, Nx
do j = 1, Ny
do i = 1, Nz
do l = 1, Nz
  dudr(l, j, k) = dudr(l, j, k) +
    a(i, l) * u(l, j, k, ie)
enddo
enddo
enddo
enddo
                    
```

Algorithm: dudr-4loop-fused

```

do k = 1, Nx * Ny
do i = 1, Nz
do l = 1, Nz
  dudr(l, k) = dudr(l, k) +
    a(i, l) * u(l, k, ie)
enddo
enddo
                    
```




Total number of variants = 20


Algorithm: dudr-4loop-permuted-and unrolled


```

do k = 1, Nx
do i = 1, Nz
do j = 1, Ny
do l = 1, Nz * 2
  dudr(l, j, k) = dudr(l, j, k) +
    a(i, l) * u(l, j, k, ie)
  dudr(l+1, j, k) = dudr(l+1, j, k) +
    a(i, l+1) * u(l+1, j, k, ie)
enddo
enddo
enddo
enddo
                    
```

Number of 4-loop implementations for dudr
 $N_x = N_y = N_z = 10$
 $= 4! * 4 * 4$
 $= 24 * 256 = 6144$ variants
 Total number of variants = 98,240 ($N=10$)
 Total number of variants = 217,728 ($N=20$)
 Exhaustive search may not be feasible




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Genetic Algorithm

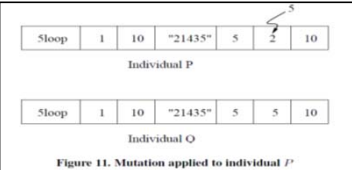


Figure 11. Mutation applied to individual P

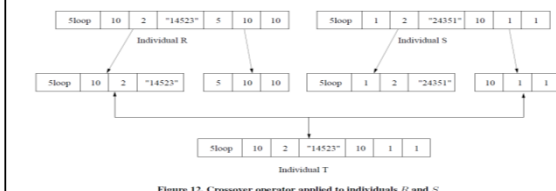




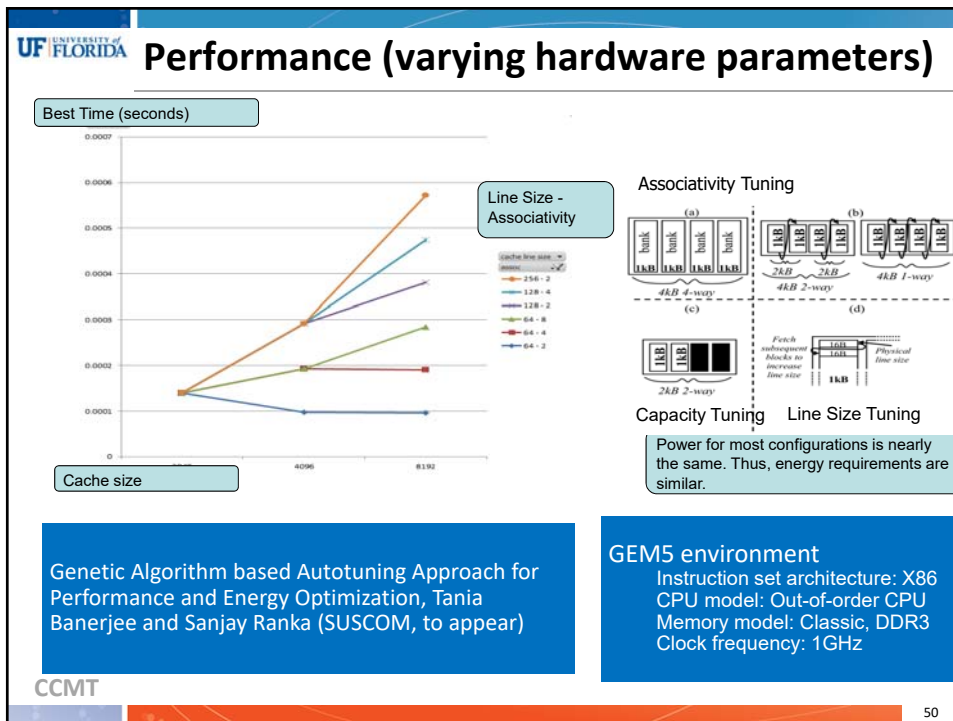
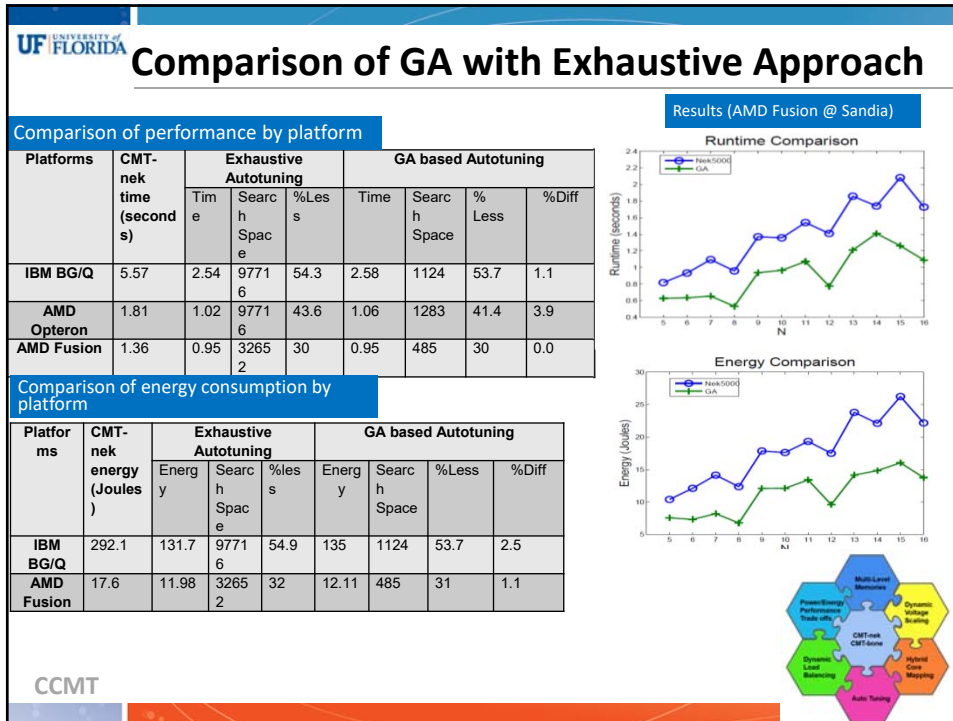
Figure 12. Crossover operator applied to individuals R and S

```

graph TD
    A[Generate Initial Population  
i=1] --> B[Compile and run each individual  
Set fitness value for each individual based on performance and energy]
    B --> C{i < n?}
    C -- No --> D[Population Selection  
Add new individuals using crossover and mutation  
i=i+1]
    D --> B
    C -- Yes --> E[Report the best individual]
                    
```




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Exascale Behavioral Emulation

Principal Investigators:

Herman Lam, Greg Stitt

Center for Compressible Multiphase Turbulence (CCMT)
NSF Center for High-Performance Reconfigurable Computing (CHREC)
ECE Department, University of Florida



UF UNIVERSITY OF FLORIDA

Exascale Behavioral Emulation (BE) Team



Ryan Blanchard
FPGA Acceleration



Sai Chenna
BE Design Space Exp.



Trokon Johnson
BE Tools



Aravind Neelakantan
BE Methods



Carlo Pascoe
FPGA Acceleration



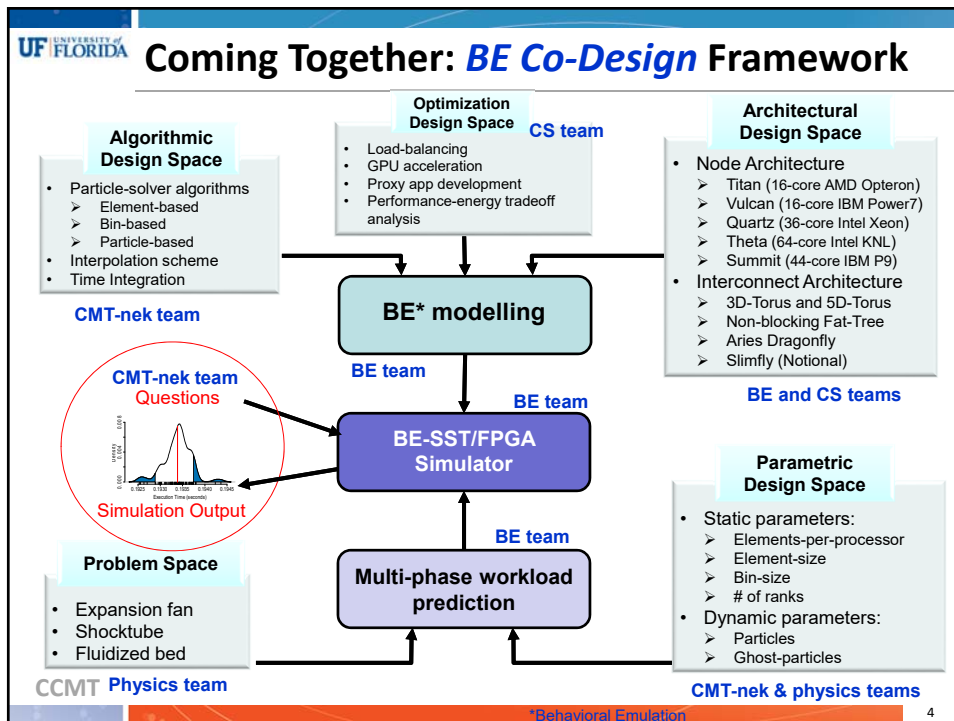
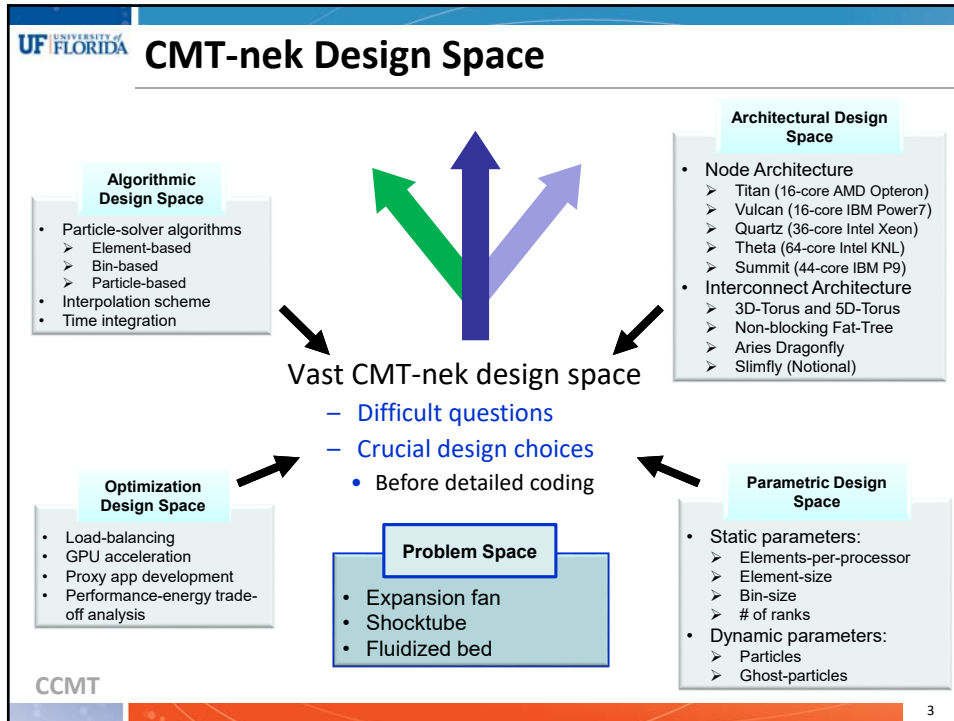
Maneesh Merugu
BE-SST

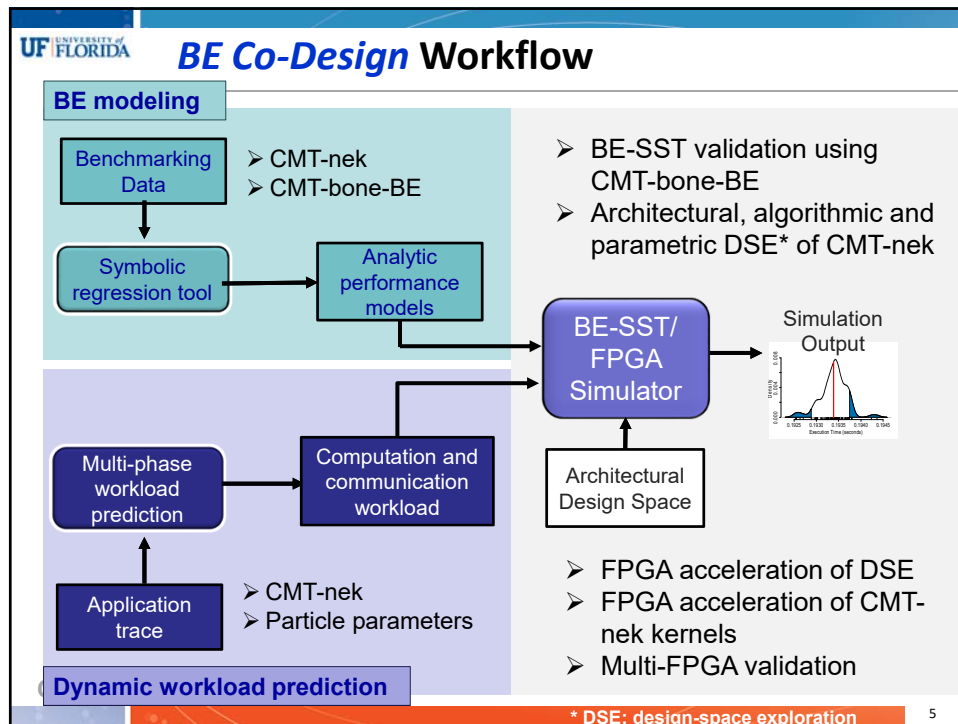


Steven Paek
Thermal Modeling

CCMT

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UF UNIVERSITY OF FLORIDA **Outline**

- Review & overview: Behavioral Emulation (BE)
 - Research thrusts & achievements**
 - Architectural exploration
 - BE method/tool enhancements
 - Multi-phase workload prediction* tool
 - Symbolic regression* for performance modeling
 - Thermal* experiments & modeling
 - Acceleration using FPGAs
 - Single- and multi-FPGA *accelerated DSE*
 - Case study*: FPGA-accelerated DSE
 - End-to-end case study
 - Closing the loop*, CMT-nek DSE to support CMT-nek team
 - Summary, conclusions, & going forward
 - Acceleration of CMT-nek kernels*: approach & preliminary results



UF FLORIDA **Outline**

- Review & overview: Behavioral Emulation (BE)

Research thrusts & achievements

- Architectural exploration
- BE
- Acc
- End

The diagram illustrates the **CMT-nek Design Space** as a central purple cube. It is surrounded by several design spaces:

- Problem Space**: Expansion fan, Shocktube, Fluidized bed.
- Algorithmic Design Space**: Particle-solver algorithms (Tree-based, Bin-based, Particle-based), Interpolation scheme, Time integration.
- Optimization Design Space**: Load-balancing, GPU acceleration, Proxy app development, Performance-energy trade-off analysis.
- Architectural Design Space** (highlighted in red):
 - Node Architecture:
 - Titan (18-core AMD Opteron)
 - Vulcan (16-core IBM Power7)
 - Quartz (36-core Intel Xeon)
 - Theta (64-core Intel Xeon)
 - Summit (44-core IBM PS)
 - Interconnect Architecture:
 - 3D-torus and 2D-torus
 - Non-blocking Fat-Tree
 - Area Disparity
 - Stiffly (Natural)
- Parametric Design Space**:
 - Static parameters:
 - Element size (processor)
 - Element size
 - Block size
 - # of ranks
 - Dynamic parameters:
 - Particles
 - Ghost particles

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UF FLORIDA **Behavioral Emulation Workflow**

The workflow is divided into two main phases:

- BEO design and calibration** (left side, orange background):
 - Application source code (manual / automated) → Instrumented source code → Existing machines OR Fine-grained simulators → benchmarking → Calibration data.
- HW/SW co-design** (right side, blue background):
 - AppBEO (application description) → Simulation Platforms (BE SST, FPGA Acceleration) → Simulation predictions → Alternate algorithms (iterative) → ArchBEO (hardware description) → Notional architectures (iterative) → Simulation Platforms.

Validation loops connect the simulation results back to the instrumented source code and calibration data. A central **BE Simulation** block is also shown.

* BEO – Behavioral Emulation Object

HW/SW co-design

- Algorithmic & architectural design-space exploration (DSE)




Coarse-grained BE simulation

- Balance of simulation speed & accuracy for rapid design-space evaluation

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BE simulations of HPC systems

- Titan @ ORNL**
 - Cray XK7 architecture with Cray Gemini interconnect
 - 16-core AMD opterons; 18k nodes; 300k cores
 - 18k K20X Kepler GPUs
 - 32GB + 6GB memory/node
- Vulcan @ LLNL**
 - IBM BG/Q architecture
 - 16 cores/node, 24k nodes, 390k cores
 - 16GB memory/node
- Quartz @ LLNL**
 - Intel Xeon (Broadwell) with Intel Omni-Path interconnect
 - 36-core Intel Xeon E5-2695; 2,634 nodes; 94k cores
 - 128GB memory/node

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Application Case Study: CMT-bone-BE

- CMT-nek & CMT-bone (mini-app) both are large codes
- To support extensive DSE*, we need:
 - Key compute kernels & comm. patterns that affect performance
 - **Abstract, modular, easy to modify and instrument for algorithmic DSE**

CMT-nek Workflow
(highlighted portions represented in CMT-bone-BE)

- Computation- & communication-intensive portions of CMT-nek workflow in C & MPI
 - Volume-to-surface data extraction
 - Face data exchange (with neighbors)
 - Derivative computation volume points
- Easier to modify for algorithmic DSE
- 1000 vs 10s of thousands lines of code

* design-space exploration

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UNIVERSITY OF FLORIDA BE-SST Simulator*

- BE-SST: Developed by extending Structural Simulation Toolkit (SST)**
 - Framework for parallel simulations
 - Supported by developers & vendors
 - Flexibility in designing custom components

* A. Ramaswamy, N. Kumar, A. Neelakantan, H. Lam, G. Stitt, "Scalable Behavioral Emulation of Extreme-Scale Systems Using Structural Simulation Toolkit", 47th International Conference on Parallel Processing, Eugene, OR, August 13-16, 2018.

** From Sandia National Labs

SST Capabilities

- Parallel simulations
- Discrete event simulation
- Clock and event queues

BE Influences

- Software Definitions
- Probabilistic Simulation
- Abstract Network Definitions
- Abstract Hardware Definitions

BE-SST

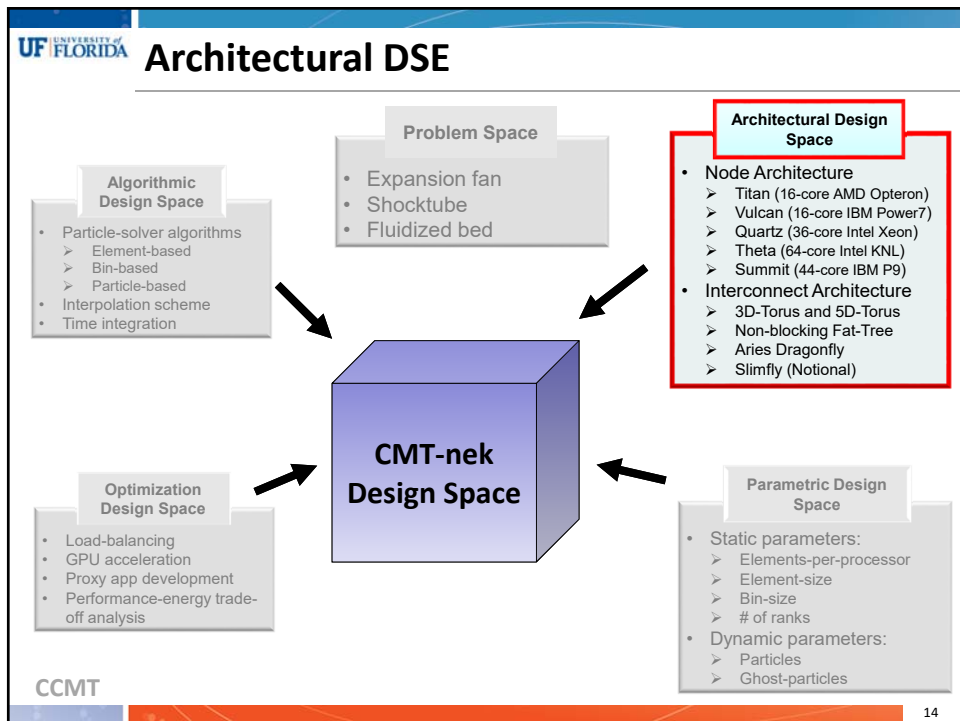
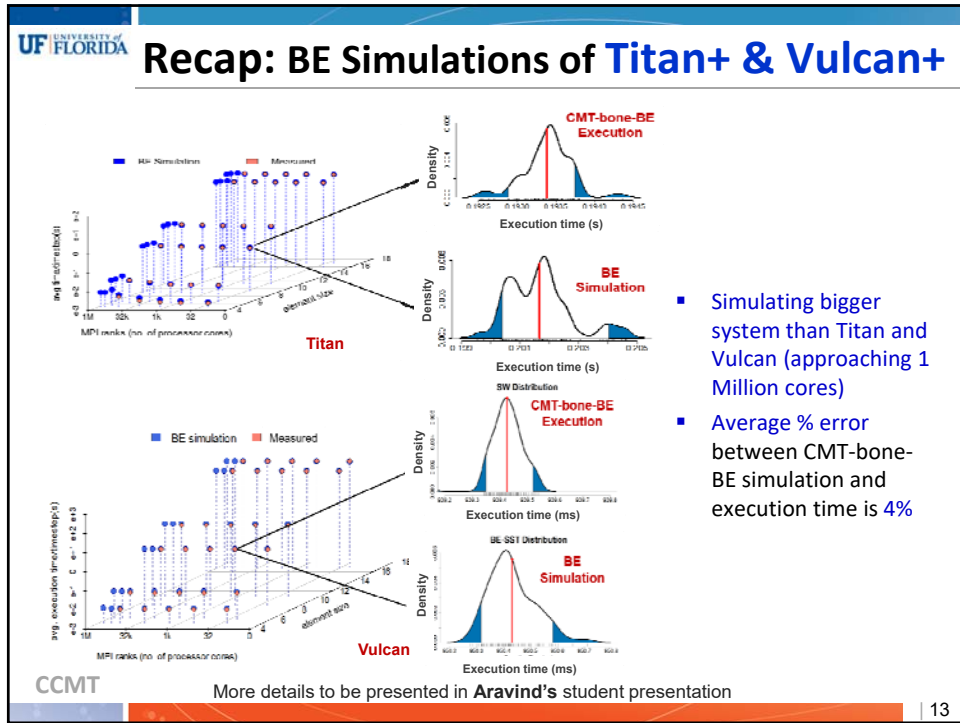
- Parallel discrete event simulation environment
- Distributed component queues
- Software Definitions
- Probabilistic Simulation
- Abstract Network Models
- Abstract Component Models

CCMT More details to be presented in Aravind's student presentation | 11

UNIVERSITY OF FLORIDA Experiment Setup

- Application case study: CMT-bone-BE (gas solver)
- Application parameters
 - Calibration
 - element size: 5,9,13,17,21
 - elements/core: 8,32,64,128,256
 - Validation
 - element size: 5,6,9,11,16,17
 - 64 elements/core
- Machine parameters
 - Validation up to 128k MPI ranks
 - Predictions up to 1M MPI ranks
 - Can perform both predictions and validations for larger systems
- Monte-Carlo Simulations

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UF UNIVERSITY OF FLORIDA **Architectural DSE**

- BE-SST performance validated in the past
- Currently, moving towards notional architecture
- Plug-n-play architectural DSE
 - Mixing node architecture with interconnect architecture

Architectural Design Space

- Node Architecture
 - Titan (16-core AMD Opteron)
 - Vulcan (16-core IBM Power7)
 - Quartz (36-core Intel Xeon)
 - Theta (64-core Intel KNL)
 - Summit (44-core IBM P9)
- Interconnect Architecture
 - 3D-Torus and 5D-Torus
 - Non-blocking Fat-Tree
 - Aries Dragonfly
 - Slimfly (Notional)

Machine	Node architecture	Interconnect
1	36-core Intel Xeon (Quartz architecture)	5D torus (of Vulcan)
2	36-core Intel Xeon (Quartz architecture)	3D torus (of Titan)

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UF UNIVERSITY OF FLORIDA **BE Simulations: Node Architecture**

■ Vulcan ■ Machine 1

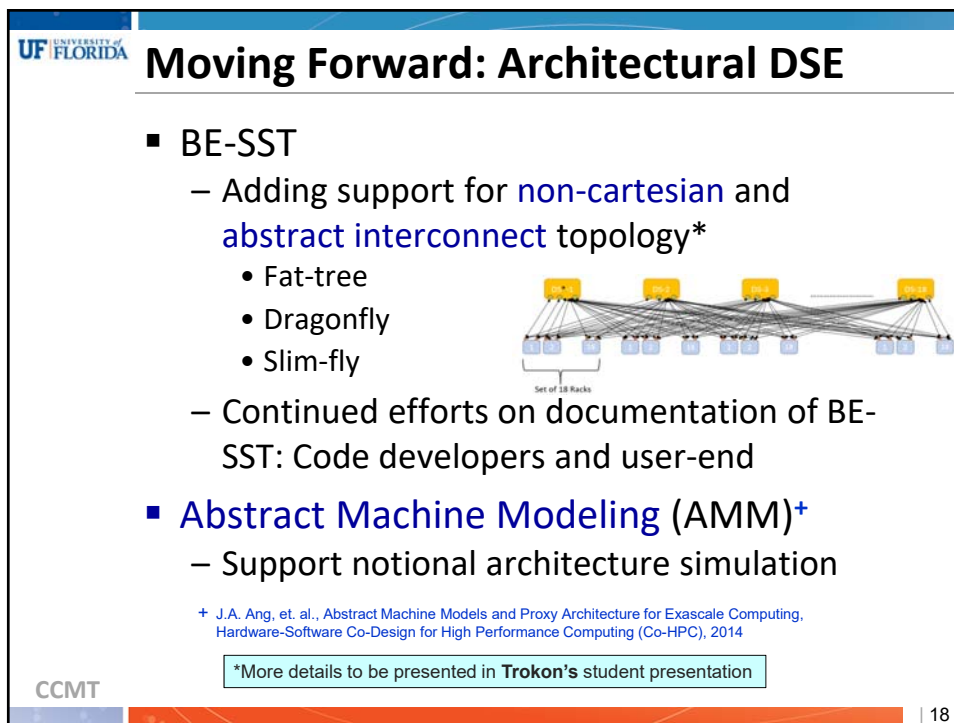
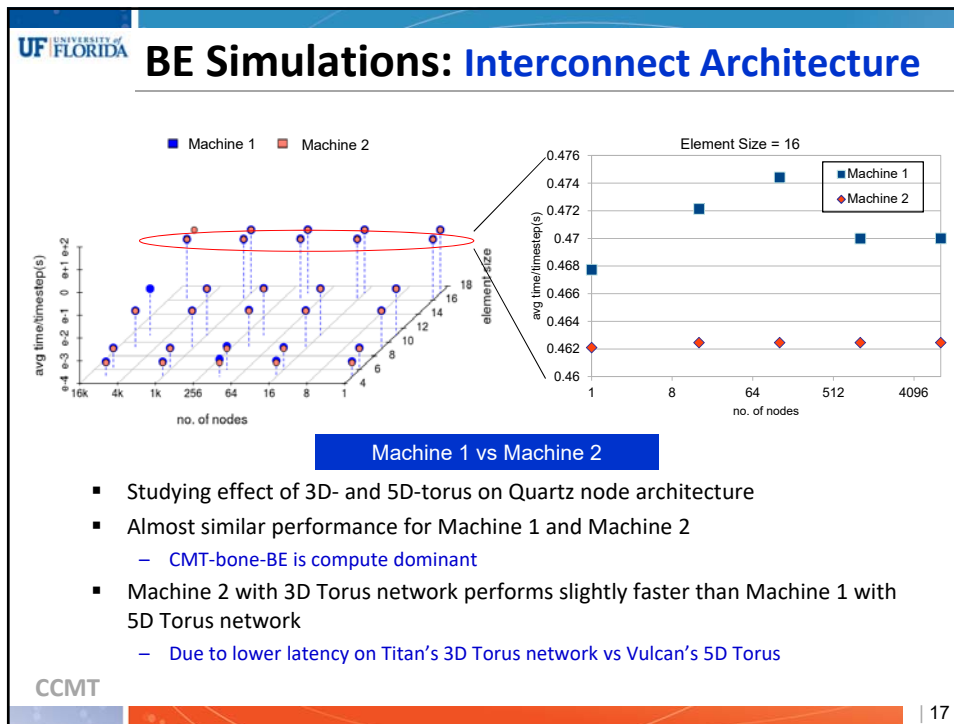
Vulcan vs Quartz with Vulcan interconnect


■ Titan ■ Machine 2

Titan vs Quartz with Titan interconnect

- Comparing compute node performance of Vulcan and Titan with Quartz
- Machine 1 and Machine 2 (with Quartz compute node) is faster than Vulcan compute nodes (by **21x**) and Titan compute node (by **5x**)
 - Expected behavior – Quartz’s Intel Xeon is newer and faster than Vulcan’s Power7 and Titan’s AMD Opteron

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Outline

- Review & overview: Behavioral Emulation (BE)

Research thrusts & achievements

- Architectural exploration
- BE method/tool enhancements
 - **Multi-phase workload prediction** tool

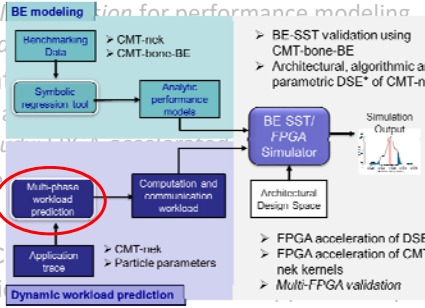


BE modeling

- CMT-nek
- CMT-bone-BE

➢ BE-SST validation using CMT-bone-BE

➢ Architectural, algorithmic and parametric DSE* of CMT-nek





➢ BE-SST/FPGA Simulator

➢ FPGA acceleration of DSE

➢ FPGA acceleration of CMT-nek kernels

➢ Multi-FPGA validation

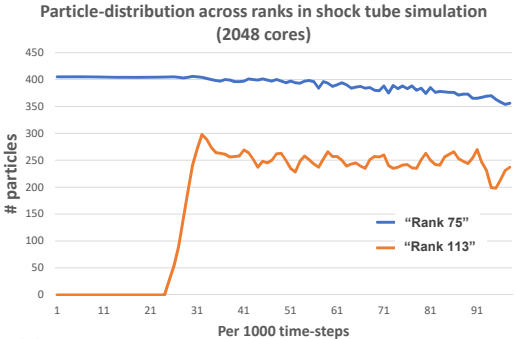

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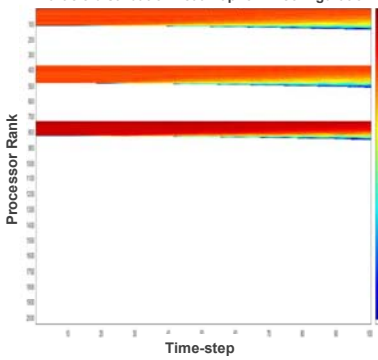
Trace-driven Simulation: Particle Solver


- Why we need trace-driven simulation?
 - Workload per processor depends on # particles
 - # particles/processor is dynamic:
 - Varies among processors – *depends on problem and domain decomposition algorithm*
 - Varies at each timestep – *based on fluid forces and particle collisions*
 - Need a trace to perform simulations

Particle-distribution across ranks in shock tube simulation (2048 cores)



Particle-distribution heat map for 2k configuration




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UF UNIVERSITY OF FLORIDA **Multi-phase Workload Prediction Tool**

Goal: Predict particle-workload **across processors** for **different** machine and application configurations

Input:

- Trace data for a specific problem
- Application & machine (processor-count) parameters

Output:

- # particles residing in each rank @ every timestep
- # ghost particles residing in each rank @ every timestep
- # particles moving across each rank @ every timestep
- # ghost particles moving across each rank @ every timestep

Summary

- **Key assumption:** When increasing problem size, particle movement pattern **does not change**
- **Single trace of a specific problem** is sufficient to predict particle movement
- Generate a synthetic trace for **larger problem sizes** based on original trace from smaller size

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UF UNIVERSITY OF FLORIDA **Case study : ASU Expansion tube**

Experimental Setup:

- Problem: ASU Expansion tube
- Particle count: 133341
- Element count: 15768
- Element size: 4
- Particle trace frequency: 1000 time-steps

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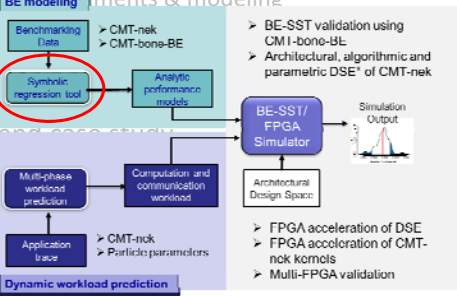
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Outline

- Review & overview: Behavioral Emulation (BE)

Research thrusts & achievements

- Architectural exploration
- BE method/tool enhancements
 - Particle-workload prediction tool
 - Symbolic regression** for performance modeling
 - Thermal models & modeling
- Acceleration
 - Single
 - Case studies
 - End-to-end
 - Closing the loop
- Summary,
 - Acceleration

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Symbolic Regression: Motivation

- Motivation:
 - Multi-parameter performance modeling is challenging
 - Requires accuracy, computational efficiency, simple implementation, flexibility
- Our previous approaches:
 - N-dimensional interpolation:
 - Computationally expensive
 - Difficult to implement, needs customization for each example
 - Insufficient accuracy
 - Multiple linear regression:
 - Faster than interpolation, relatively simple to implement
 - Need thorough understanding of code to generate performance models
 - May not capture machine-specific behaviors that deviate from basic model

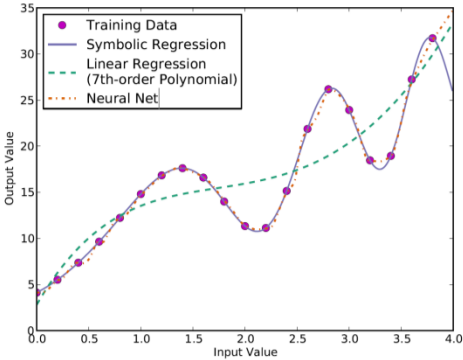
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Symbolic Regression Overview

- Definition: *discovers* equation for model from training samples
 - e.g. $5.802(\sin(x^2) + x) + 4.12$
- Advantages:
 - Captures machine-specific performance behaviors
 - Enables tradeoffs between accuracy and computational complexity
 - Requires no prior knowledge of the kernel



	% Testing Error	# of Model Operations
Symbolic Regression (20 training samples)	0.05%	5
Linear Regression (7th-order Polynomial)	57.9% (1157x)	44 (8.8x)
NN (16 neurons/layer, 1 hidden layer)	16.6% (332x)	608 (122x)
NN (32 neurons/layer, 4 hidden layers)	3.0% (60x)	8,480 (1,696x)
NN (64 neurons/layer, 8 hidden layers)	2.9% (57x)	66,368 (13,273x)
NN (w/ 5000 training samples)	0.4% (7.2x)	8,480 (1,696x)
Average Improvement	323x	3,359x



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Symbolic Regression: Results

CMT-nek Particle Solver Kernel	Linear Regression	Symbolic Regression	Error Improvement
	MAPE	MAPE	$\text{MAPE}_{\text{Linear}}/\text{MAPE}_{\text{Symbolic}}$
Compute1	0.87%	0.76%	1.4x
update_particle_location	0.42%	0.37%	1.1x
interp_props_part_location	1.82%	2.02%	0.9x
upd_vel_and_pos_rk3_stage1	46.31%	3.82%	12.1x
upd_vel_and_pos_rk3_stage3	10.10%	3.38%	2.9x
upd_vel_and_pos_rk3_allstage	16.32%	2.46%	6.6x
upd_vel_and_pos_bdf	20.70%	1.94%	10.7x
red_interp	5.49%	1.91%	2.9x
usr_particles_forces	1.09%	1.14%	0.96x
tri_interp	3.47%	2.81%	1.2x
Average	10.66%	2.06%	4.1x

MAPE: Maximum Absolute Percentage Error


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Outline

Review & Research

- Archit
- BE me
- Part
- Sym
- Ther

- Acceleration using FPGAs
 - Single- and multi-FPGA *accelerated DSE*
 - *Case study*: FPGA-accelerated DSE
- End-to-end case study
 - Closing the loop, CMT-nek DSE to support CMT-nek team
- Summary, conclusions, & going forward
 - Acceleration of CMT-nek kernels: approach & preliminary results

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FPGA-Accelerated Design-Space Exploration

Example design space:

- 20 values for left, lx1, lpart
- 10 different numbers of cores
- 10 different core types
- 10 different memory configurations
- 4 different network topologies

32 million options to explore

↓

BE Simulator (SW)

↓

Optimized Design

~Minutes per simulation

- **BE Advantage:** significantly faster than existing simulators
- **BE Limitation:** still not fast enough for DSE of exascale systems
- **Approach:** Use **FPGA acceleration** to improve exploration
 - Sacrifice analysis capabilities to prune design space
 - Use BE-SST to analyze remaining candidates

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FPGA-Acceleration Summary

- Previously introduced two approaches for accelerating BE simulations:
 - Fully expanded pipeline: 7-9 orders of magnitude faster than BE-SST, but limited multi-FPGA scalability
 - Collapsed pipeline: 6 orders of magnitude faster than BE-SST, and scales well across multiple FPGAs
- Demonstrated simulations up to 1M ranks

Ranks	TS	Num. of Events	% LU	Latency (cycles)	Hardware MSPS ¹	Hardware GEPS ²	BE-SST KEPS ²	Hardware Speedup
32	32	43,008	44	7,532	10.5	450	10.14	44x10 ⁶
64	32	92,160	46	10,252	5.23	482	24.91	19x10 ⁶
128	32	190,464	47	10,316	2.62	498	24.42	20x10 ⁶
256	32	393,216	67	13,516	1.31	515	23.92	21x10 ⁶
512	32	811,008	84	20,684	0.65	531	21.47	25x10 ⁶
1K	32	1,646,592	84	21,196	0.33	539	18.21	29x10 ⁶
32K	32	55,443,456	84	241,868	1.02x10 ⁻²	567	12.96	44x10 ⁶
128K	16	111,673,344	46	333,932	2.56x10 ⁻³	285	8.91	32x10 ⁶
1M	2	112,459,776	5	1,148,056	3.19x10 ⁻⁴	36	X	X

As number of ranks increase:

- additional logic per rank approaches zero
- simulation throughput reduced by a factor of “ranks”
- event throughput remains proportional to instantiated event hardware

*Collapsed performance of CMT-Bone-BE with varied MPI ranks and simulation timesteps on a single Stratix V SSGSMD8K1F40C2 @ 335MHz

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Case Study: Single-FPGA-accelerated DSE

- Goal: Use FPGA to explore enormous design space
- Application: CMT-nek particle solver kernel
- Initial Design Space:

Parameter	Range
<i>Element-size</i>	5-25 (21)
Element-count	4-512 (8)
<i>particles/gridpoint</i> (α)	0.1-10 (19)
# of ranks	16-1M (17)

Algorithmic options:

Time Integration	Rk3,bdf
Interpolation	Barycentric, Reduced Barycentric, Trilinear interpolation

Design space = 273,904 candidates

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Case Study: Single-FPGA-accelerated DSE

Ranks	No of events	FPGA Avg. Simulation time (microseconds)	BE-SST* Avg. Simulation time (seconds)	FPGA speedup
1024	24,576	3.53	8.234	2.33x10 ⁶
2048	49,152	7.06	6.5	9.21x10 ⁵
4096	98,304	14.12	17.3	1.22x10 ⁶
8192	196,608	28.24	31	1.10x10 ⁶
16k	393,216	56.48	120	2.12x10 ⁶
32k	786,432	112.96	145	1.28x10 ⁶
64k	1,572,864	225.92	405	1.79x10 ⁶
128K	3,145,728	451.84	531	1.17x10 ⁶
256K	6,291,456	903.68	1569	1.74x10 ⁶
512K	12,582,912	1807.36	7416	4.10x10 ⁶
1M	25,165,824	3614.72	X	X

- Each configuration (row) has **16,112** design candidates
- BE-SST takes **5.5 years** to explore the design space – **FPGA takes 61 seconds**
 - 550 years** including 100 Monte Carlo simulations for each design candidate vs **100 minutes**
 - FPGA sacrifices analysis capabilities of BE-SST to achieve this speedup

CCMT *ran on HiPerGator - 64 cores

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Case Study: Single-FPGA-accelerated DSE

- Application: CMT-nek particle solver
- Architecture: Vulcan (512k cores)
- 1 design candidate** – BE-SST* - **12000 minutes**
- Entire space (270K candidates) – BE-FPGA - **100 minutes**

BE-FPGA Simulation

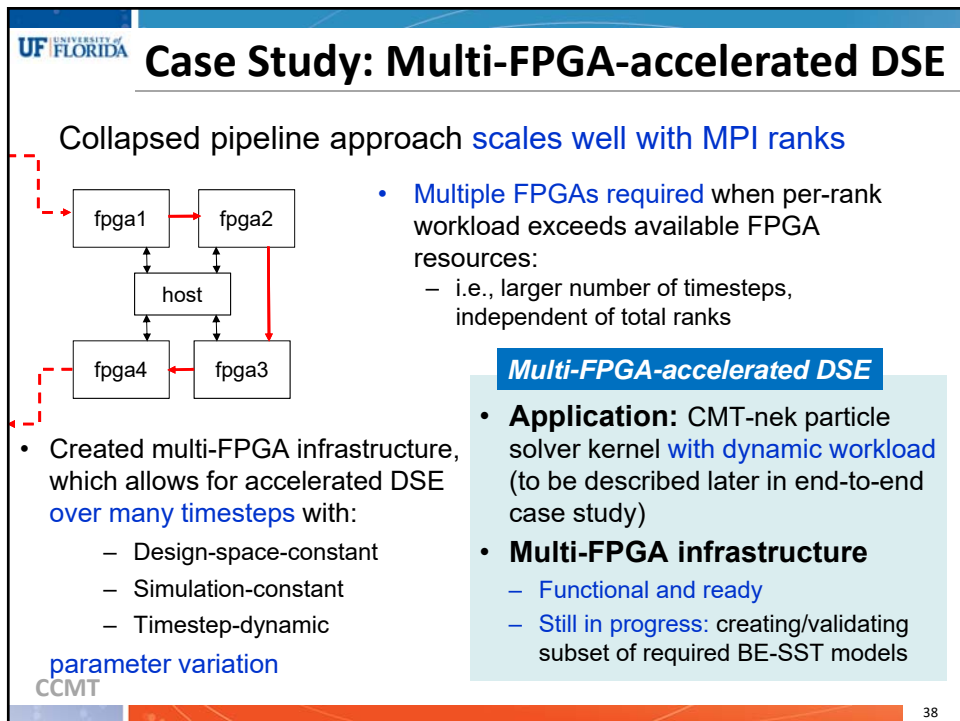
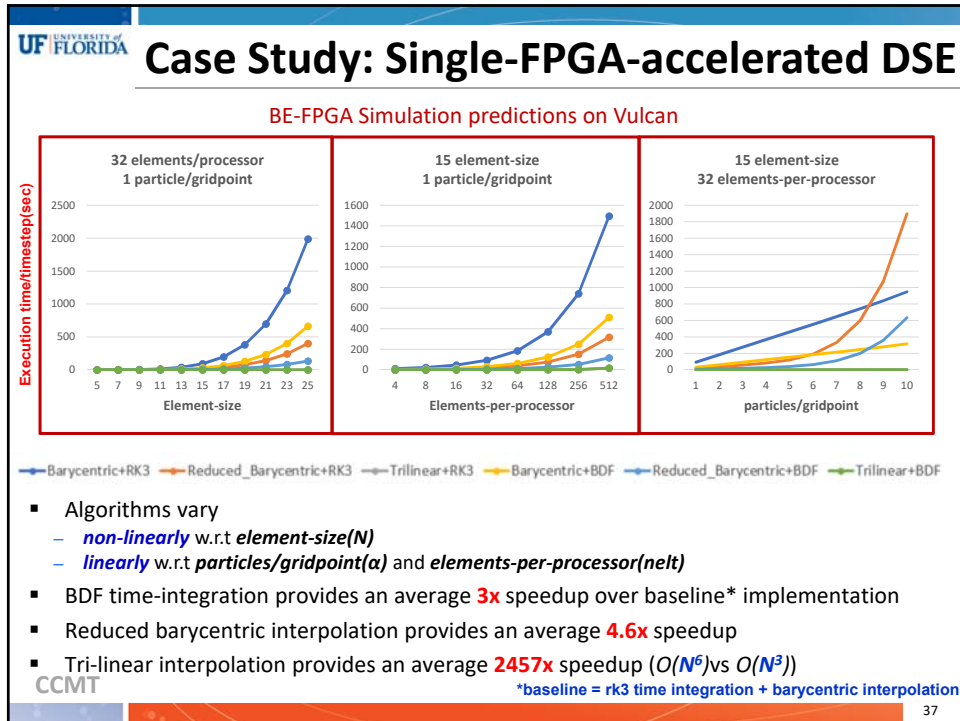
BE-SST Simulation (32k cores)

100 runs & 100 simulations

CMT-nek Particle-Solver Execution (4k cores)

CCMT *ran on HiPerGator - 64 cores **100 Monte-Carlo simulations/candidate

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Outline

- Review & overview: Behavioral Emulation (BE)

Co-Design Framework and Teams

The diagram illustrates a co-design framework with the following components and teams:

- Problem Space:** Includes Expansion fan, Shocktube, and Filohead bed. Associated with the **CMT-nek physics team**.
- Particle-workload prediction:** Receives input from the Problem Space and feeds into the BE-SST/FPGA Simulator. Associated with the **CMT-nek physics team**.
- BE-SST/FPGA Simulator:** Receives input from Particle-workload prediction and feeds into BE* modelling. Associated with the **BE team**.
- BE* modelling:** Receives input from the BE-SST/FPGA Simulator and feeds into BE team. Associated with the **BE team**.
- Simulation Output:** Receives input from BE* modelling and feeds into the BE team. Associated with the **BE team**.
- Algorithmic Design Space:** Includes Particle solver algorithms, Element-based, Run-based, Particle-based, Interpolation scheme, and Time integration. Associated with the **CMT-nek team**.
- Optimization Design Space:** Includes Load balancing, GPU acceleration, Proxy app development, and Performance-energy tradeoff analysis. Associated with the **CMT team**.
- Architectural Design Space:** Includes Node Architecture (Titan (16-core AMD Opteron), Vulcan (16-core IBM Power7), Quartz (36-core Intel Xeon), Theta (60-core Intel Xeon), Summit (44-core IBM P8)) and Interconnect Architecture (3D-Torus and 5D-Torus, Non-blocking Fat-Tree, Area Dragonfly, Slingshot (National)). Associated with the **BE and CS team**.
- Parametric Design Space:** Includes Static parameters (Elements-per-processor, Element-size, Bin-size, # of ranks) and Dynamic parameters (Particles, Ghost-particles). Associated with the **CMT-nek & physics team**.


- End-to-end case study
 - Closing the loop, CMT-nek DSE to support CMT-nek team
- Summary, conclusions, & going forward
 - Acceleration of CMT-nek kernels: approach & preliminary results

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Closing the Loop

- Goal:
 - Use BE to provide valuable feedback to CMT-nek code development team (closing the loop)
 - Performance prediction is difficult in case of dynamic workload
 - Solution: trace-driven simulation
 - Performing trace-driven simulations to answer the following questions:
 - How the workload varies across different processor configurations?
 - Used to find pareto-optimal point between speedup and resource utilization
 - How would the problem scale for more number of elements?
 - Useful to predict the application performance for large scale problem sizes
 - What is the effect of projection filter on application performance?

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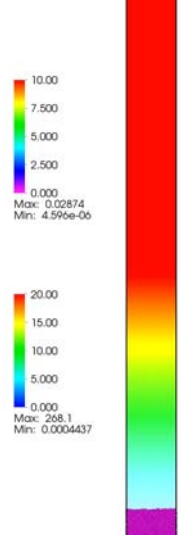


Case Study : Shocktube Simulation


- Experimental Setup:
 - Problem: ASU Shocktube
 - Particle count: 4194625(4.2 million)
 - Element count: 74304
 - Element size: 5
 - Particle trace frequency: 10 time-steps

Experiment Design Space

Processor configuration	2048,4096,8192,16384,32768,65536
Element configuration	16x774x6 (74304) 32x774x12 (297216) 64x774x24 (1188854)



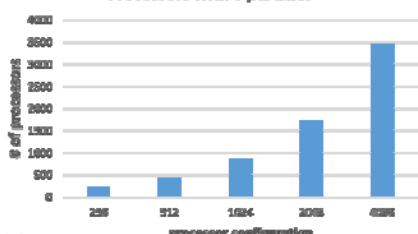
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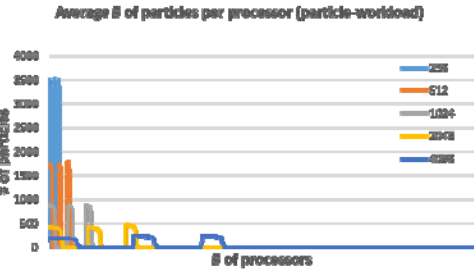
Closing the Loop: Question 1

- Question 1: What is the optimal configuration (# of processors) to run for a given problem? (or) How the resource utilization varies across different processor configurations?
 - Strong scaling decreases element workload and average particle workload
 - However strong scaling results in poor resource utilization (# of processors with no particle workload increases)
 - Useful in detecting load-imbalance associated with running the problem with different processor counts.

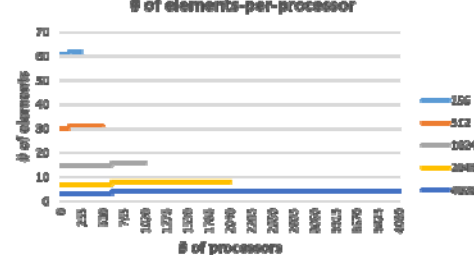
Processors with 0 particles



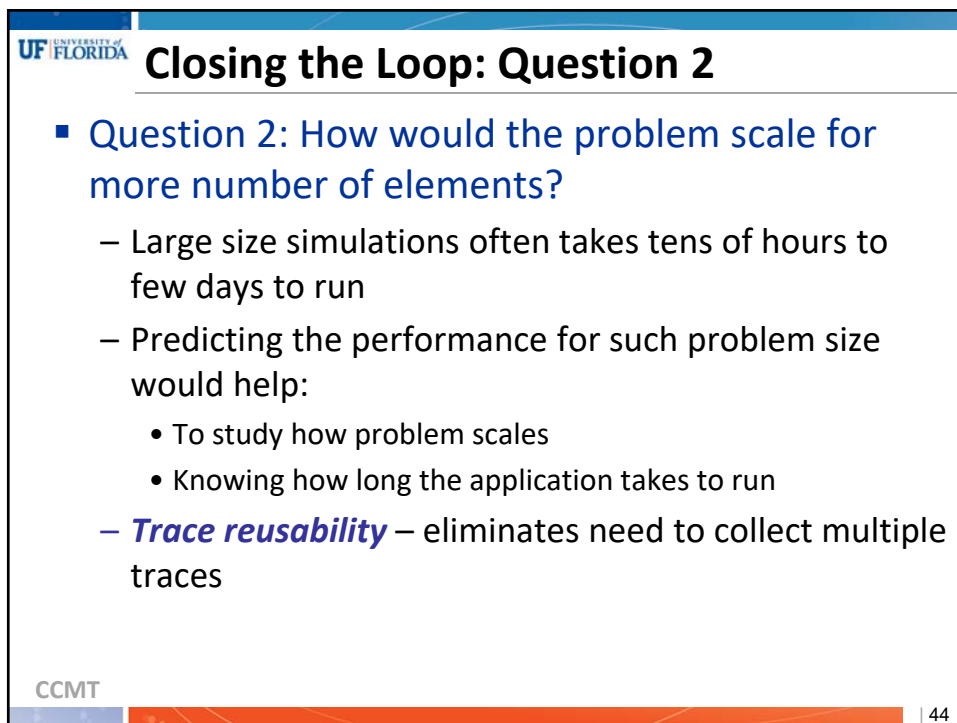
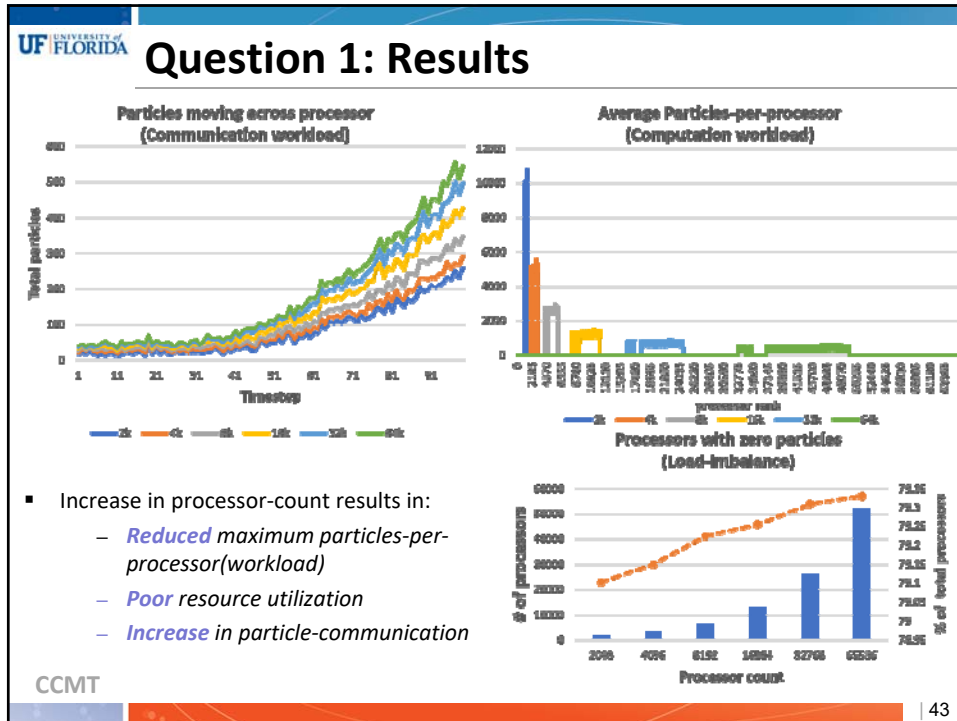
Average # of particles per processor (particle-workload)

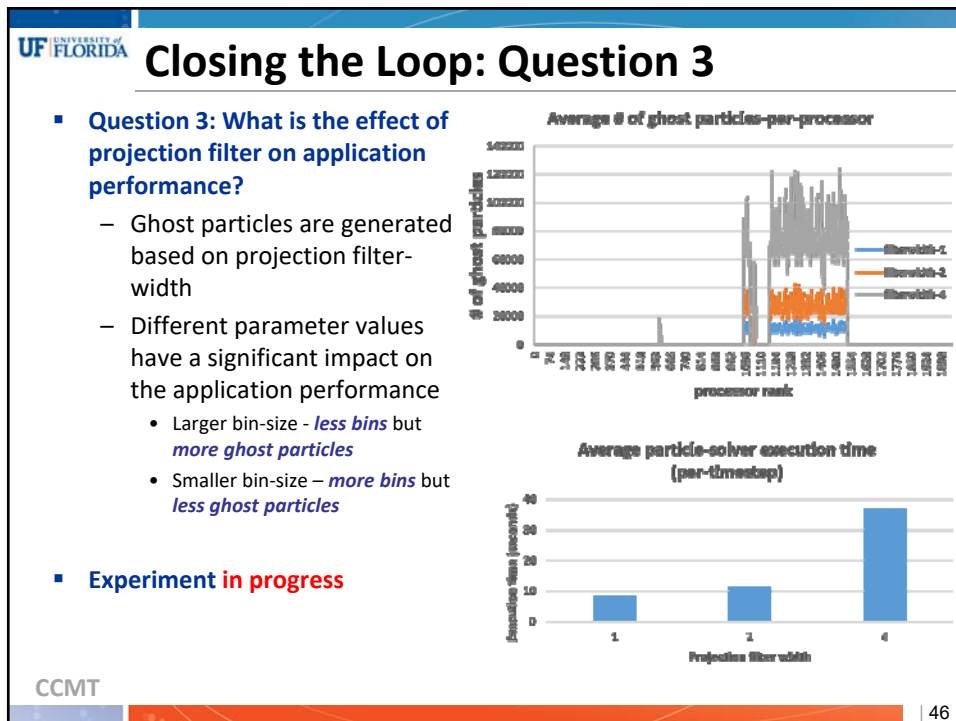
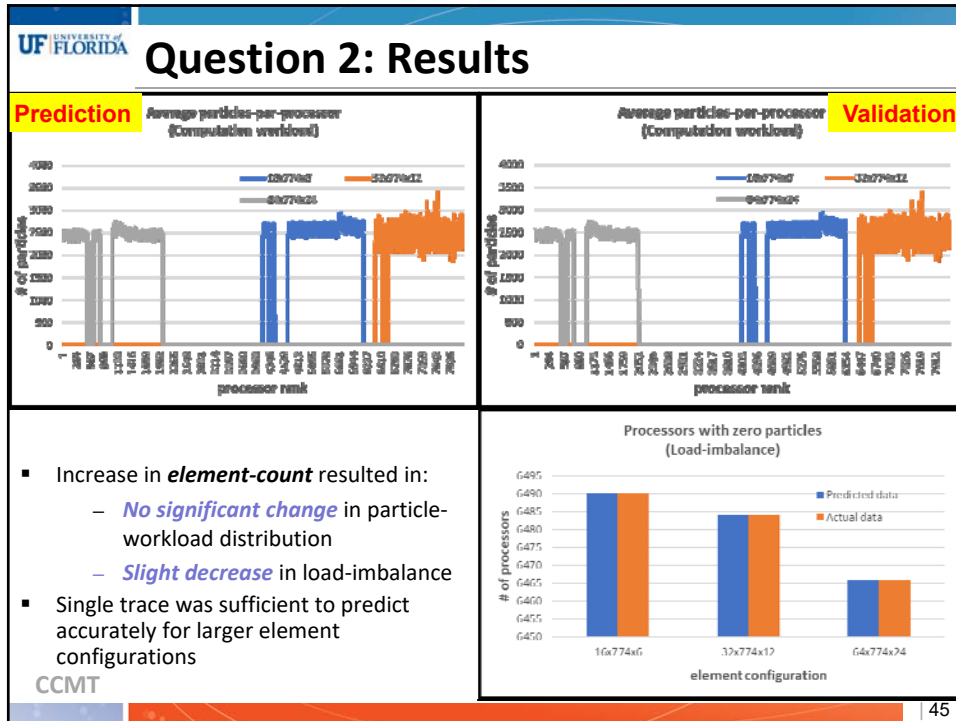


of elements-per-processor



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Outline

- Review & overview: Behavioral Emulation (BE)
 - **Research thrusts & achievements**
 - Architectural exploration
 - BE method/tool enhancements
 - *Particle-workload* prediction tool
 - *Symbolic regression* for performance modeling
 - *Thermal* experiments & modeling
 - Acceleration using FPGAs
 - Single- and multi-FPGA *accelerated DSE*
 - *Case study*: FPGA-accelerated DSE
 - End-to-end case study
 - Closing the loop, CMT-nek DSE to support CMT-nek team
 - Summary, conclusions, & going forward
 - [Acceleration of CMT-nek kernels](#): approach & preliminary results



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Summary & Conclusions

- Architectural exploration
 - [Benchmarking & validation](#) on Titan, Vulcan, & Quartz
 - [Predictive](#) simulations approaching a [million](#) MPI ranks
 - Architectural DSE – [Node](#) architecture; [interconnect](#) architecture
- BE method/tool enhancements
 - [Multi-phase workload prediction](#) tool; [symbolic regression](#) modeling; [thermal](#) modeling
- FPGA acceleration of BE simulation
 - [Validation](#) of dataflow, pipelined approaches
 - Case study demonstration of [FPGA-accelerated DSE](#)
- End-to-end case study: CMT-nek DSE
 - Exploring [non-obvious questions](#) from CMT-nek team
 - Highlighting [multi-phase workload prediction tool](#)

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FPGA Acceleration Trends

- FPGAs growing trend in data centers
 - Microsoft Catapult, Amazon F1, Intel Broadwell+FPGA
 - National lab interest
- Recently performed case study on Intel Broadwell+Arria 10 (BDW+A10) for 2D convolution
 - 96x less energy than Broadwell
 - 15.7x less energy than P6000 GPU
- Future research
 - FPGA acceleration of CMT-nek functions
 - Intel Xeon+FPGA system
 - Faculty part of NSF CHREC/SHREC
 - NSF centers for high-performance and low-energy FPGA research




		Energy Comparison of BDW+A10 vs. Broadwell					
		Image Size					
Precision	Kernel Size	256x256	512x512	1024x1024	2048x2048	Avg	
16-bit Fixed	3x3	35x	47x	67x	68x	54x	
	5x5	68x	116x	171x	179x	133x	
	7x7	58x	135x	153x	164x	128x	
	9x9	44x	98x	135x	148x	106x	
	Avg	51x	99x	131x	140x		
32-bit Float	3x3	33x	46x	65x	66x	53x	
	5x5	64x	111x	162x	169x	127x	
	7x7	55x	113x	131x	139x	109x	
	9x9	32x	58x	71x	75x	59x	
	Avg	46x	82x	108x	112x		

		Energy Comparison of BDW+A10 vs. P6000 GPU					
		Image Size					
Precision	Kernel Size	256x256	512x512	1024x1024	2048x2048	Avg	
16-bit Fixed	3x3	15.5x	19.8x	19.4x	19.0x	18.4x	
	5x5	12.9x	18.6x	19.5x	19.8x	17.7x	
	7x7	10.5x	18.2x	19.7x	20.5x	17.2x	
	9x9	8.1x	15.3x	16.9x	18.6x	14.7x	
	Avg	11.8x	18.0x	18.9x	19.5x		
32-bit Float	3x3	14.6x	19.4x	19.0x	18.6x	17.9x	
	5x5	12.2x	17.8x	18.6x	18.7x	16.8x	
	7x7	9.9x	15.2x	16.9x	17.3x	14.8x	
	9x9	5.8x	9.0x	8.9x	9.4x	8.3x	
	Avg	10.6x	15.4x	15.8x	16.0x		

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Do you have any questions?

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LANL Internship Overview

Frederick Ouellet

Mentor: Cristoph Junghans

Group: CCS-7

ISTI Co-Design Summer School 2018





LA-UR-18-27165



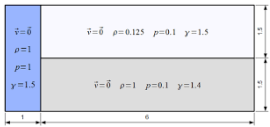


Project Overview

Goal: Optimize performance of the FleCSALE hydrocode and its linking to EOSPAC

Details:

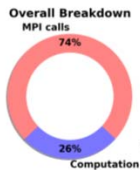
- FleCSALE is a nominally ALE code built on the FleCSI framework
- Supports multiple parallel backends (MPI, Legion, ...)
- The EOSPAC library provides the interface from FleCSALE to the SESAME database for equation of state calculations and interpolations
- The test problem run for optimization was the 2D triple point problem



Strategies:

- Optimizing the EOSPAC Interface
- Machine Learning for EoS interpolation
- GPU Porting
- Sparse data optimization (MPI)

Overall Breakdown



Computation Breakdown

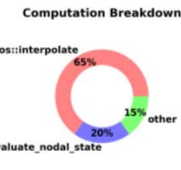


Figure 1: Initial communication and computation breakdown of FleCSALE

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Results

- Interface was optimized through processor grouping and array sorting

Figure 2: Timing and performance data for FleCSALE runs with various integration techniques. Inversion/grouping speeds up the code by a factor of 1.5, adding sorting increases it to a factor of 1.6-1.7

Figure 3: Timing data for 10,000 interpolation calls to EOSPAC

Figure 4: Relative difference between the serial and CUDA implementations

- Kernel Ridge Regression and Random Forest models were used to replace interpolation

Figure 5: Absolute error of random forest model compared to EOSPAC

Figure 6: Integration of KRR ML with FleCSALE

- The MPI handling of ghost cells was modified to remove excessive copies of data which were created in runtime

Figure 10: Strong scaling graph for FleCSALE with relative speedup

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Outside the Lab

Santa Fe – Meow Wolf (left), International Food and Art Show (right)

Bathtub Row (with the CDSS group) Bandelier National Monument and the Valles Caldera

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Do you have any questions?



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